

Bldg C.-FAX 561-681-9392 - West Palm Beach, Florida 33406

Mail Remittance To:
Gulfstream Aerospace Corporation
P.O. Box 730349
Dallas, Texas 75373-0349

Overnight Remittance To:
Gulfstream Aerospace Corporation (Box # 730349)
c/o JP Morgan Chase Bank, NA
ATTN: Dallas National Wholesale Lockbox TX1-0029
14800 Frye Road
Ft. Worth, TX 76155

Wire Remittance To:
Gulfstream Aerospace Corporation
JP Morgan Chase Bank, NA
Chicago, IL 60670
Account No. 10-15825
Fedwire Routing No. (ABA) 021 000 021
ACH Credit Routing No. (ABA)071 000 013

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Bill To: **HYPERION AIR, INC.**
3800 SOUTHERN BLVD
SUITE 204
WEST PALM BEACH, Florida 33406
United States

Invoice Number: **7112000954**
Invoice Date: **3/15/2012**
Customer ID: **CH200**
P.O. Number:
Repair Station: **FAA CRS PB8R624N**
In Date: **02/06/2012**
Out Date: **03/23/2012**
Payment Type: **Net 30-Open terms**
Payment Terms:

Title: **HYPERION AIR, INC.**

INVOICE SUMMARY FOR WORK ORDER NUMBER: SC10420

	<u>Time & Materials</u>	<u>Flat Rate</u>	<u>Total</u>
Labor USD:	\$27,009.12	\$161,722.63	\$188,731.75
Parts USD:	\$91,680.83	\$6,500.00	\$98,180.83
Services USD:	\$33,067.13	\$0.00	\$33,067.13
Detail Subtotal USD:	\$151,757.08	\$168,222.63	\$319,979.71
Subtotal USD:			\$319,979.71
Funds on Deposit USD:			\$0.00
Total USD:			\$319,979.71

For detail description of the work complied with see the Work Order Detail report.

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Item: **I** **Airframe Gulfstream G1B Scheduled N909JE**Team Manager: **Sean Brown**Customer Coordinator: **Ray Kraber**Part/Model #: **G1B**AC Serial No./Tail No.: **151 / N909JE**

TSN/TSO:

CSN/CSO: **5663**Aircraft Time: **10804.3**Discrepancy: **1.1** **Return to Service FAR 91.409 (f) (3) Inspection**Resolution: **ACTT: 10,804.3****ACTL: 5,663****DATE: March XX, 2012**

I certify that this aircraft has been inspected in accordance with a **Due List Inspection** under the approved manufacturer's recommended inspection program, FAR 91.409 (f)(3) and is approved for return to service with regards to the maintenance accomplished. Further details of this inspection and related maintenance are on file at this repair station under the above work order.

Thomas J. Choynowski, Sr. Airworthiness Inspector

Gulfstream-PBI
1500-C Perimeter Road
West Palm Beach, FL 33406
FAA CRS PB8R624N

Step: 1.1.1 **Waiting** **A/C Times and Cycles**
approval (L)

Step: 1.1.2 **Waiting** **A/C Warranty Status / Contracts**
approval (L)

Discrepancy: **1.2** **Inbound/Outbound Inspection**Resolution: **Complied with inbound inspection as required.**Labor USD: **No Charge****N/C**

Step: 1.2.1 **Completed** **Inbound/Preliminary Inspection**

Step: 1.2.2 **Waiting** **Outbound Inspection**
approval (L)

Discrepancy: **1.3** **Preflight Inspection**

Resolution:

Discrepancy: **1.4** **Comply with Debrief Checklist**

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Discrepancy: **1.10 Remove Valuables & Secure.**
Resolution: **N/A**

Discrepancy: **1.11 Comply with Service Team Manager Checklist**
Resolution:

Discrepancy: **1.12 Fluids Used (Oils/Skydrol) - (Materials Only)**
Resolution: **Billed as required.**

Discrepancy: **1.13 STM / Quality review of work order and logbook entry prior to departure**
Resolution:

Discrepancy: **1.14 Towing (Does not include Hangar Moves)**
Resolution: **Towed as required.**

Labor USD: **No Charge**

N/ C

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Discrepancy: **1.15** **HazMat - Identify & provide a manifest, and a Shipper's Declaration for Dangerous Goods for all hazardous materials that are loaded aboard any aircraft.**

FAR Part 121 or 135 Operators will provide a hazardous material notification letter required by FAR 135.505 (e) prior to maintenance at this Repair Station iaw FAR 145.206 (a) (b).

If there is to be loading of hazardous materials it shall be complied with by Hazmat trained authorized GDAS-PBI personnel only.

Resolution: **N/A**

Discrepancy: **1.16** **Aircraft Cleaning**

Resolution: **N/A**

Discrepancy: **1.17** **Billable Consumables & Shop Supplies**

Resolution: **as required.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Parts USD:	Flat Rate			\$4,000.00
Total USD:				\$4,000.00

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Discrepancy: **1.18** **Major Alterations regulatory compliance requirements to be completed by installer as indicated on FAA form 8110-3.**

Copy and paste from attached statements as needed and edit as required.

COMPLY WITH REQUIREMENTS OF FAR 25.1301 (d), 25.1353(a) and 25.1431(c) (d).

COMPLIED WITH THE REQUIREMENTS OF FAR 25.1301 (d), 25.1353(a) and 25.1431(c) (d), via GROUND TEST REPORT-REPORT No. CEXXXXXXXXXX REV. X, (IN SUPPORT OF STC XXXXXXXXXXXX) and GR4R216M REPAIR STATION INTERFERENCE TESTING REQUIREMENTS POST MAJOR ALTERATIONS COMPLIANCE FORM.

COMPLY WITH REQUIREMENTS OF FAR 25.1529, INSTRUCTIONS FOR CONTINUED AIRWORTHINESS.

REF XXXXX MANUAL, P/N XXXXXXXX REV. X, FOR INSTRUCTIONS FOR CONTINUED AIRWORTHINESS AND BLOCK 8 OF FAA FORM 337, DATED XX/XX/XXXX.

COMPLY WITH THE WIRE ROUTING REQUIREMENTS OF FAA POLICY ANM-01-04 AND FAR 25.1353 (B)

COMPLIED WITH THE WIRE ROUTING REQUIREMENTS OF FAA POLICY ANM-01-04 AND FAR 25.1353 (B) BY ENSURING ROUTING WAS IN ACCORDANCE AC43.13-1B, CHANGE 1 CHAPTER 11, SECTION 8.

Resolution: **N/A**

Discrepancy: **1.19** **Update Weight & Balance**

Resolution: **N/A**

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Discrepancy: 1.20 Exterior Paint and Weight & Balance Requirements.

Review scope of aircraft exterior paint touch-up/repairs, and skin replacements resulting in paint reapplication, and determine affect on weight and balance.

For surfaces where paint is stripped and reapplied or added on top of existing paint on an area greater than 36 square feet the thickness values will be documented before and after painting . If average thickness values remain the same, weight and balance revision is not required.

For surfaces painted with an area greater than 36 square feet and before and after paint thickness average varies, QC review for possible weight and balance revision will be required.

* Thickness will be plotted on an aircraft drawing derived from AMM chapter 6 or other suitable sketch, attached to the work order and measurements taken on a two square foot grid.

* Reference GAMPS 4000, paragraph 3.5.1 for paint restrictions to flight control surfaces and RVSM certified aircraft.

* Reference GAMPS 4000, total average thickness shall not exceed 6 mils.

* Per GAC Mass Properties Engineering, paint thickness of 5 mils equals .055 pounds per square foot.

* If after QC review, weight change is greater than 2 pounds or .05% of MAC, the weight and balance report will require recomputation.

Resolution: N/A

Discrepancy: 1.21 "Advise customer of preservation requirements per ATA Chapter 10 of the relevant maintenance manuals for Airframe, Engines, and APU; review requirements in relation to scheduled Date Promised and later potential Date Completed; and perform maintenance or preservation actions as directed by the customer and documented as a Work Order Squawk.

Resolution: N/A

Item: 6 Avionics

Team Manager: Sean Brown

Customer Coordinator: Ray Kraber

Part/Model #:

AC Serial No./Tail No.:

TSN/TSO:

CSN/CSO:

Aircraft Time:

Discrepancy: 6.1 MED emergency light does not illuminate when EMER LT switch is turned on or in the AUTOMATIC mode. Works when switch is placed in MANUAL

Resolution: Performed operational check. Found MED emergency light works in manual but not automatic. Gained access to MED emergency lights. Found G-switch tripped. Reset G-switch. Performed operational check. Checks good. All work performed IAW G-II AMM chapter 33-02-00.

For detail description of the work complied with see the Work Order Detail report.

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	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	ST	2.83	@	\$115.00	\$325.45
Total USD:						\$325.45

Discrepancy: **6.2** **AFT entryway light inop.**

Resolution:

Item: **8** **Mechanical**

Team Manager: **Sean Brown**

Customer Coordinator: **Ray Kraber**

Part/Model #:

AC Serial No./Tail No.:

TSN/TSO:

CSN/CSO:

Aircraft Time:

Discrepancy: **8.1** **Extra time for removing BUL-1 & BUL-2 so that BUL-3 can be removed for elevator removal.**
Note: these panel not on removal sheet for elevators.

Resolution: **Complied with info only.**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate					\$920.00
Total USD:						\$920.00

Discrepancy: **8.2** **While complying with CMP# 272015, Right aileron removal/installation, found most screws corroded/rusted.**
Quantity : (100) (P/N: NAS 1203-3) screws, (40) (P/N: NAS 1203-5) screws.
NOTE: Quantity included's rusted/corroded Left aileron R&R panel screws.

Resolution: **Installed new screws as reuired. P/N NAS 1203-3 and NAS 1203-5. IAW GIIB AMM 27-00-00 and 20-00-00.**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate					\$230.00
Parts USD:	Time and Material					\$300.32
Total USD:						\$530.32

For detail description of the work complied with see the Work Order Detail report.

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Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.2 #1	NAS1203-3	BOLT	100	\$0.26	\$26.00
8.2 #2	NAS1203-5	BOLT	50	\$0.26	\$13.00
8.2 #3	GAS550E3S9	SCREW	10	\$4.12	\$41.20
8.2 #4	GAS550E3S10	SCREW	4	\$1.81	\$7.24
8.2 #5	NAS583-11	BOLT	4	\$24.66	\$98.64
8.2 #6	NAS583-13	BOLT	2	\$15.68	\$31.36
8.2 #7	NAS583-14	BOLT	2	\$23.77	\$47.54
8.2 #8	NAS583-16	BOLT	2	\$17.67	\$35.34

Discrepancy: **8.3** While comply with CMP 272810 Rudder removal. Found upper bonding wire broken at vert stab

Resolution: **Removed broken upper rudder bonding wire at vert stab P/N MS25083-2BB13 and replaced with new bonding wire P/N MS25083-2BB13 IAW GIIB AMM 20-00-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$10.30
Total USD:				\$240.30

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.3 #1	MS25083-2BB13	JUMPER-ELEC BONDING,13 IN	1	\$10.30	\$10.30

Discrepancy: **8.4** While complying with CMP#'s 575030/575025 , Wing to fuselage interior Left/Right inspection, found ALL screws rusted/corroded. Most needed screwbusters for removal which amounted to considerable time over CMP aloted time.

Screws needed as follows:
 P/N: MS24694s50 : Qty: 330
 P/N: 1159B21241-71 Qty: 1
 P/N: 1159B21241-78 Qty: 1
 P/N: 1159B21241-33 Qty: 1
 P/N: 1159B21241-73 Qty: 1
 P/N: 1159B21241-81 Qty: 1
 P/N: 1159B21241-35 Qty: 1

Resolution: **Removed and discarded corroded screws and installed new screws.P/N: MS24694S50 : Qty: 330**

P/N: 1159B21241-71 Qty: 1
 P/N: 1159B21241-78 Qty: 1
 P/N: 1159B21241-33 Qty: 1
 P/N: 1159B21241-73 Qty: 1
 P/N: 1159B21241-81 Qty: 1
 P/N: 1159B21241-35 Qty: 1

All work done i/a/w G-II B SRM ch 51-41-00. No further defects noted.

For detail description of the work complied with see the Work Order Detail report.

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	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$78.00
Total USD:				\$538.00

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.4 #1	MS24694-S50	SCREW-MACHINE,FLAT CSK HEAD	300	\$0.26	\$78.00

Discrepancy: **8.5** **Change float valve mid point on right wing per customer.**

Resolution: **Remove vent float valve P/N 159SC-P-006 S/N1010. Install a New float valve P/N 159SCP006-1 S/N 4771. Install new O rings under mount screw heads. All work performed IAW GII B AMM 28-02-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST 15.88	@ \$115.00	\$1,826.20
Labor USD:	Time and Material	OT 0.13	@ \$173.00	\$22.49
Parts USD:	Time and Material			\$2,557.27
Total USD:				\$4,405.96

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.5 #1	1159SCP006-1	PUMP	1	\$1,925.35	\$1,925.35
8.5 #3	HT183-3-2	SCREW	3	\$197.76	\$593.28
8.5 #4	GAS13AF6	O-RING	3	\$12.88	\$38.64

- Step: 8.5.1** **Completed** Remove float valve mid point on right wing P/N159SC-P-006 S/N 1010. Work IAW GII B MM 28-02-00
- Step: 8.5.2** **Completed** Positioned new float valve in place, P/N 1159SCP006-1, S/N 4771, and tightened mounting screws. Float valve will have to be removed in order to install 3 o-rings that are to be placed under mounting screws. Refer to Gulfstream IIB IPC 28-10-00 figure 6, sheet 1, item 3. O-rings are on order.
- Step: 8.5.3** **Completed** Install 3 o rings under float valve mounting screw heads IAW GII B AMM 28-01-01.

Discrepancy: **8.6** **Nine rivnuts need replaced on R/H wing wedge panel. Required drilling of screws due to rivnuts spinning.**

Resolution: **Removed and replaced nine rivnuts on the R/H wing wedge panel. Installed new rivnuts P/N GAN512B3-166 wet with sealant. Had to gain access to fuel tank area to remove drilled out rivnuts All work done IAW G-IIB SRM 51-41-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST 13.44	@ \$115.00	\$1,545.60
Labor USD:	Time and Material	OT 4.35	@ \$173.00	\$752.55
Parts USD:	Time and Material			\$273.47
Total USD:				\$2,571.62

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.6 #1	GAN512B3-166	BLIND NUT	10	\$1.52	\$15.20

For detail description of the work complied with see the Work Order Detail report.

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Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.6 #2	GAN512B3-216	NUT	10	\$6.10	\$61.00
8.6 #3	GAN512B3-266	RIV-NUT	10	\$15.88	\$158.80
8.6 #4	PR1422-B-2	SEALANT, PRC, CARTRIDGE	1	\$38.47	\$38.47

- Step: 8.6.1 Completed Disconnected the 5 hydraulic lines to gain access to the sponson access cover 22-SR-2 all lines are capped and or plugged. RLP
- Step: 8.6.2 Completed Removed access cover 22-SR-2 on outboard side of sponson .
- Step: 8.6.3 Completed Removed all spinning and bad rivnuts on the R/H wing wedge panel IAW G-IIB AMM 51-40-00.

Discrepancy: 8.7 Left wing flap tracks A,B,C, and D have corrosion on roller bearing surfaces.

Resolution: Removed corrosion from flap track bearing surfaces on tracks A,B,C and D IAW G IIB CPCM 20-50-01.Measured tracks A,B,C and D for thickness as required IAW Gulfstream drawing SE05811901. See item 8.88 for details on track C. All others check good. Performed brush cadmium plating repair to all exposed surfaces of flap tracks A,B,C and D IAW Gulfstream drawing SE05811901.Cleaned and protected tracks and surrounding area, masked-off track bearing surface, applied two coat epoxy primer 2012 and surface primer, applied top coat, IAW. GAMPS 4000.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$6,900.00
Parts USD:	Time and Material			\$153.12
Total USD:				\$7,053.12

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.7 #1	PR1440-B-1/2	SEALANT, PRC,CARTRIDGE	4	\$38.28	\$153.12

- Step: 8.7.1 Completed Removed corrosion from flap track bearing surfaces on tracks A,B,C and D IAW G IIB CPCM 20-50-01.
- Step: 8.7.2 Completed Measure tracks A,B,C and Dfor remaining thickness.
- Step: 8.7.3 Completed Performed brush cadmium plating repair to all exposed surfaces of flap tracks A,B,C and D IAW Gulfstream drawing SE05811901.
- Step: 8.7.4 Completed Cleaned and protected tracks and surrounding area, masked-off track bearing surface, applied two coat epoxy primer 2012 and surface primer, Prep. tracks to top coat. IAW. GAMPS 4000.

Discrepancy: 8.8 Right wing flap tracks A,B,C, and D have corrosion on roller bearing surfaces.

Resolution: Removed corrosion from flap track bearing surfaces on tracks A,B,C and D IAW G IIB CPCM 20-50-01.Measured tracks A,B,C and D for thickness as required IAW Gulfstream drawing SE05811901, with no defects noted. Performed brush cadmium plating repair to all exposed surfaces of flap tracks A,B,C and D IAW Gulfstream drawing SE05811901.Masked off flaps tracks A B C and D, washed with solvent and primed with 2012 primer, and applied topcoat IAW GAC GAMPS 4000.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$6,900.00
Total USD:				\$6,900.00

For detail description of the work complied with see the Work Order Detail report.

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- Step: 8.8.1** **Completed** Removed corrosion from flap track bearing surfaces on tracks A,B,C and D IAW GHB CPCM 20-50-01.
- Step: 8.8.2** **Completed** Measure tracks A,B,C and D for remaining thickness.
- Step: 8.8.3** **Completed** Performed brush cadmium plating repair to all exposed surfaces of flap tracks A,B,C and D IAW Gulfstream drawing SE05811901.
- Step: 8.8.4** **Completed** Cleaned and protected tracks and surrounding area, masked-off track bearing surface, applied two coat epoxy primer 2012 and surface primer, Prep. tracks to top coat. IAW. GAMPS 4000. (Written in ERROR this step belongs under item 8.7) Refer to squawk 8.7.4.
- Step: 8.8.5** **Completed** Masked off flaps tracks A B C and D, washed with solvent and primed with 2012 primer ready to spray paint.

Discrepancy: **8.9** L/H wing wedge has 18 drilled off screw heads because rivnuts were spinning.

Resolution: **Removed and replaced eighteen rivnuts on the L/H wing wedge panel. Installed new rivnuts P/N GAN512B3-166 wet with sealant. Had to gain access to fuel tank area to remove drilled out rivnuts All work done IAW G-IIB SRM 51-41-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$2,300.00
Parts USD:	Time and Material			\$117.00
Total USD:				\$2,417.00

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.9 #1	GAB511GA3-7	HI-BOLT	1	\$2.06	\$2.06
8.9 #2	PR1422-B-2	SEALANT, PRC, CARTRIDGE	1	\$38.47	\$38.47
8.9 #3	GAN512B3-166	BLIND NUT	25	\$1.52	\$38.00
8.9 #4	PR1422-B-2	SEALANT, PRC, CARTRIDGE	1	\$38.47	\$38.47

- Step: 8.9.1** **Completed** Removed all spinning and bad rivnuts on L/H wing wedge IAW G-IIB AMM 51-40-00.

Discrepancy: **8.10** Aileron trim tab has cracked skin in the area of the inner and outer hinges on the bottom side of the tab.

Resolution: **Trimmed aileron tab hinge skin areas as required IAW Gulfstream drawing SE35813107.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Total USD:				\$460.00

Discrepancy: **8.11** R/H elevator, top, inbd, hinge cover attach screw hole, heli-coil is bad.

Resolution: **Replaced bad R/H elevator, top, inbd, hinge cover attach screw hole, heli-coil with a new one IAW GAMPS 2212. Checks good.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00

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Total USD:

\$460.00

Discrepancy: **8.12 L/H elevator No.3 hinge, attach bolt bore is corroded.**

Resolution: **Removed corrosion as required and remeasured L/H elevator No.3 hinge, attach bolt bore. Hole measurement is within limits. Work performed IAW GIB MM 27-01-00. and SRM 51-13-10.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Total USD:				\$460.00

Discrepancy: **8.13 R/H elevator, oubd top skin, has a spot of corrosion.**

Resolution: **Removed light corrosion spot on R/H elevator outbd top skin, Sanded treated and primed top coat as required IAW GIV gamps 4000**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$345.00
Total USD:				\$345.00

Step: 8.13.1 Completed Removed and blend out light surface corrosion from the R/H elevator, upper surface
Treated bare metal, and spot primer as required. IAW. SRM.51-13-20.

Discrepancy: **8.14 R/H wing upper sponson plate is corroded.**

Resolution: **Mechanically removed corrosion from edges of the R/H upper sponson plate, Strip heavy paint for a better look IAW. GAC. SE45802903. Removed all old sealant from around sponson plate. Treated and primed all bare metal. Resealed sponson plate. All work done IAW Standard Repair/Corrosion of Sponson Plates DWG No. SE45802903 REV.D. Performed cadmium plating repair on R/H upper sponson plate. All work done IAW Standard Repair/Corrosion of Sponson Plates DWG No. SE45802903 REV.D**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST 17.65	@ \$115.00	\$2,029.75
Labor USD:	Time and Material	OT 4.33	@ \$173.00	\$749.09
Services USD:	Time and Material			\$400.00
Total USD:				\$3,178.84

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
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Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
8.14 #1	SERVICE	1.00	\$400.00	\$400.00

- Step: 8.14.1** **Completed** Mechanically removed corrosion from edges of the R/H upper sponson plate, Strip heavy paint for a better look IAW. GAC. SE45802903.
- Step: 8.14.2** **Completed** Removed all old sealant from around sponson plate. Treated and primed all bare metal. Resealed sponson plate. All work done IAW Standard Repair/Corrosion of Sponson Plates DWG No. SE45802903 REV.D
- Step: 8.14.3** **Completed** Performed cadmium plating repair on R/H upper sponson plate. All work done IAW Standard Repair/Corrosion of Sponson Plates DWG No. SE45802903 REV.D

Discrepancy: **8.15** R/H wing flap has 2 spots of corrosion. (Marked)

Resolution: **Removed light surface corrosion from R/H flap upper surface [outboard flap track & mid track]. Cleaned treated bare metal, primer and top coat as required. All work performed IAW. GAMPS 4000.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$1,150.00
Total USD:				\$1,150.00

- Step: 8.15.1** **Completed** Removed and blend out light surface corrosion from the R/H Flap, upper surface around #4 track, Treated bare metal, and spot primer as required. IAW. SRM.51-13-20.

Discrepancy: **8.16** R/H wing flap outbd Yoke is cracked in radius.

Resolution: **Removed damaged flap "C" track yoke assembly (PN - 1159CM20621-3) and discarded. Installed a new replacement yoke assembly (PN - 1159CM 20621-3) by reference of the Gulfstream IIB M/M chapter 27-06-00 entitled, Flap Actuator Yoke - Inspection. During this install, no defects were noted at this time.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$988.80
Total USD:				\$1,448.80

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.16 #2	1159CM20621-3	YOKE ASSY, FLT CONT - FLAP DRIVE	1	\$494.40	\$494.40
8.16 #3	1159CM20621-3	YOKE ASSY, FLT CONT - FLAP DRIVE	1	\$494.40	\$494.40

- Step: 8.16.1** **Completed** Removed damaged flap "C" track yoke assembly (PN - 1159CM20621-3) and discarded. Installed a new replacement yoke assembly (PN - 1159CM 20621-3) by reference of the Gulfstream IIB M/M chapter 27-06-00 entitled, Flap Actuator Yoke - Inspection. During this install, no defects were noted at this time.

Discrepancy: **8.17** R/H flap, outbd L/E flapper doors are binding on each other.

Resolution: **Ajusted flap door as required IAW GII AMM 27-00-00. Ops checks good.No further action required.**

Billing Method	Hours	Rate	Total
For detail description of the work complied with see the Work Order Detail report.			

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Bill To: **HYPERION AIR, INC.**

WO #: **SC10420**

Invoice Number: **7112000954**

Labor USD: **Flat Rate**

\$460.00

Total USD:

\$460.00

Discrepancy: **8.18 R/H flap, outbd yoke upper bushing phenolic is debonded from the bushing. Bushing needs to be replaced.**

Resolution: **Removed and replaced the flap "C" track Support Assembly PN - 1159CSM20836-1) by reference of the Gulfstream IIB M/M chapter 27-06-00 entitled, Flap Actuator Yoke - Inspection. During this install, no defects were noted at this time.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$3,450.00
Parts USD:	Time and Material			\$840.93
Total USD:				\$4,290.93

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.18 #4	1159CSM20836-1	SUPPORT ASSY,TRACK C-ACT YOKE	1	\$628.30	\$628.30
8.18 #6	ADLF8-4005	BUSHING	2	\$62.28	\$124.56
8.18 #7	1159C20615-5	WASHER	1	\$46.87	\$46.87
8.18 #8	1159C20615-11	WASHER	1	\$41.20	\$41.20

- Step: 8.18.1 Completed** Drilled out existing solids and huck rivets for 20 inches to gain access. Peeled back skin and inspected bolt heads and found both bolt heads mushroomed over from install. Removed fwd bolt and bagged as required. All work done I.A.W GIB M.M.27-06-00.
- Step: 8.18.2 Completed** Removed damaged flap "C" yoke support assembly (PN -1159CSM20836-1) from flap assembly and installed a new replacement "C" yoke support assembly (PN - 1159CSM20836-1) by reference of the Gulfstream IIB M/M chapter 27-06-00 entitled, wing flap yoke fittings - removal / installation.

Discrepancy: **8.19 L/H elevator trim tab has cracked skin in the area of the inner and outer hinges on the top side of the tab.**

Resolution: **Trimmed L/H elevator outbd trim tab hinge skin area as required IAW Gulfstream drawing SE05516001.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Total USD:				\$460.00

Discrepancy: **8.20 L/H aileron trim tab is missing its data plate.**

Resolution: **Completed Traveler RS-QCT-0069 Rev.05. No defects noted on trim tab.Installed new data plate IAW traveler RS-QCT-0069 Rev.05. No further action required.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

	Billing Method		Hours		Rate	Total
Labor USD:	Time and Material	ST	0.25	@	\$115.00	\$28.75
Labor USD:	Time and Material	OT	0.27	@	\$173.00	\$46.71
Parts USD:	Time and Material					\$5.97
Services USD:	Time and Material					\$1,135.00
Total USD:						\$1,216.43

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.20 #1	1159F30041-11	NAMEPLATE	1	\$5.97	\$5.97

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
8.20 #1	SERVICE	1.00	\$1,135.00	\$1,135.00

Discrepancy: **8.21** The left side, outbd wing attachfitting located at FS 345.875 has exposed bare metal and needs to be painted to prevent corrosioin.

Resolution: **Painted the left side, outbd wing attachfitting located at FS 345.875 with epoxy primer IAW GAMPS 4000.**

	Billing Method		Hours		Rate	Total
Labor USD:	Flat Rate					\$460.00
Total USD:						\$460.00

Discrepancy: **8.22** The left side wing to fuselage rib., located at FS 365.500, is cracked at the bottom of the rib, through the nut plate.

Resolution: **Fabricated and installed doubler repair as required. All work accomplished per Gulfstream IIB, Structural Repair Manual, 51-70-00.**

	Billing Method		Hours		Rate	Total
Labor USD:	Time and Material	ST	7.67	@	\$115.00	\$882.05
Labor USD:	Time and Material	OT	0.27	@	\$173.00	\$46.71
Total USD:						\$928.76

Discrepancy: **8.23** LH fuselage to wing leading edge anti-ice duct has broken bonding wire. P/N: MS25083-1BB4.

Resolution: **Removed broken bonding wire P/N: MS25083-1BB4 from LH fuselage to wing leading edge anti-ice duct and installed new bonding wire P/N: MS25083-1BB4 IAW GAMPS 2605. Resistance check good.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Bill To: **HYPERION AIR, INC.**

WO #: **SC10420**

Invoice Number: **7112000954**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$172.50
Parts USD:	Time and Material			\$56.70
Total USD:				\$229.20

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
8.23 #2	PS870-B-1/2	SEALANT 6 OZ CARTRIDGE 654 SEMKIT, 3.5 FL. OZ	1	\$56.70	\$56.70

Discrepancy: **8.24 Strap on bottom of fuselage behind nose gear rusted**

Resolution:

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$1,725.00
Total USD:				\$1,725.00

Step: 8.24.1 Completed Removed light surface corrosion on strap at bottom of fuselage behind nose gear. All work performed IAW GIB MM 51-10-00.

Step: 8.24.2 Completed Performed brush cadmium plating of belly strap aft of nose gear door IAW GIB CPCMM 20-50-05 6.D.

Discrepancy: **8.25 R/H aileron boost bell crank bottom wing plate has groove carved into it from bell crank bushing. Ref Cmp 12.36**

Resolution: **After further inspection found not to be grooved but to be a shim. No further action required on this item.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$287.50
Total USD:				\$287.50

Discrepancy: **8.26 R/H aileron boost bell crank has signs of chafing on top wing plate and bell crank. Ref CMP 12.36**

Resolution: **Cleaned area of wing plate and treated, Installed bell crank and measured for proper clearance. All work done IAW GIB AMM 27--00-00 and 20-00-00. No further action required.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$287.50
Total USD:				\$287.50

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **8.27 R/H aileron boost bell crank has four bearings that are stiff and grinding. Ref CMP 12.36**

Resolution: **Replaced the right aileron boost bellcrank assy. with a "Repaired" bellcrank assembly P/N 1159CM20216-6 NSN and P/N 1159CM20217-2 NSN. Refer to item 12.36 for removal and installation of the R/H aileron boost bellcrank assy.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	No Charge			N/ C
Services USD:	Time and Material			\$3,004.49
Total USD:				\$3,004.49

Services:

<u>Ref. No.</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell Price USD</u>	<u>Extended USD</u>
8.27 #2	Vendor Supplied Labor	1.00	\$1,753.56	\$1,753.56
	Part Ref: 1159CM20216-2 aileron boost bellcrank			
8.27 #3	REPAIR	1.00	\$1,250.93	\$1,250.93
	Part Ref: 1159CM20217-2 BELLCRANK ASSEMBLY			

Discrepancy: **8.28 Upper and lower bearing on L/H aileron boost bellcrank is chipped.**

Resolution: **Reinspected L/H aileron boost bellcrank bearings and determined that the retaining plates on the bearings are not chipped but is actually a design feature in manufacturing. No further action required pertaining to this item.**

Labor USD:	No Charge			N/ C
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Discrepancy: **8.29 CMP GIIB 803013 Engine Starter Assembly (Left) - Removal / Installation**

Resolution: **COMPLIED WITH CMP GIIB 803013 Engine Starter Assembly (Left) - Removal / Installation. Removed Left Engine Starter Assy P/N 383700-1-1 S/N P-16758 for spline lubrication and inspection and re-installed StarterP/N 383700-1-1 S/N P-16758 IAW G-2B AMM 80-01-00 with no defects noted.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$690.00
Total USD:				\$690.00

Discrepancy: **8.30 CMP GIIB 803012 Engine Starter Splines (Left) - Lubrication / Inspection**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **COMPLIED WITH CMP GIIB 803012 Engine Starter Splines (Left) - Lubrication / Inspection with satisfactory results IAW G2-B AMM 80-01-00.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$690.00
Total USD:				\$690.00

Discrepancy: **8.31 Nylon plugs (anti torsion bumper) on L/H aileron actuator worn and need replacing. (4 ea)**

Resolution: **Removed 4 worn plugs and installed 4 new plugs P/N 87048 on L/H aileron actuator IAW GIIB AMM 27-00-00.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$90.64
Total USD:				\$320.64

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
8.31 #1	87048	PLUG	4	\$22.66	\$90.64

Discrepancy: **8.32 R/H aileron actuator anti torsion bumpers(4) need replaced. P/N 87048 Ref CMP 12.37**

Resolution: **Removed 4 worn plugs and installed 4 new plugs P/N 87048 in the R/H aileron actuator IAW GIIB AMM 27-00-00.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$90.64
Total USD:				\$320.64

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
8.32 #1	87048	PLUG	4	\$22.66	\$90.64

Discrepancy: **8.33 Left outbd flight spoiler bellcrank assy to be sent to Savannah for insp.**

Resolution: **Replaced the L/H outboard flight spoiler bellcrank assy. with repaired bellcrank P/N 1159CM20209-5. Refer to item 12.153 for Removal and installation of bellcrank assy.**

<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
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For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Bill To: HYPERION AIR, INC.

WO #: SC10420

Invoice Number: 7112000954

Services USD: **Time and Material**

\$1,322.88

Total USD:

\$1,322.88

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
8.33 #1	Vendor Supplied Labor	1.00	\$1,322.88	\$1,322.88
Part Ref: 1159CM20209-5 CRANK ASSEMBLY				

Discrepancy: **8.34** Right outbd flight spoiler bellcrank assy to be sent to Savannah for insp.

Resolution: Replaced the R/H outboard flight spoiler bellcrank P/N 1159CM20209-6 and installed "repaired" bellcrank P/N 1159CM20209-6. Refer to item 12.149 for Removal and installation of bellcrank.

Services USD: **No Charge**

N/C

Discrepancy: **8.35** Right inbd flight spoiler bellcrank assy to be sent to Savannah for insp.

Resolution: Replaced the R/H inboard flight spoiler bellcrank with a "repaired" bellcrank P/N 1159CM20208-2. Refer to item 12.152 for Removal and installation of bellcrank.

Billing Method	Hours	Rate	Total
Services USD: Time and Material			<u>\$1,995.50</u>
Total USD:			<u>\$1,995.50</u>

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
8.35 #2	Vendor Supplied Labor	1.00	\$1,995.50	\$1,995.50
Part Ref: 1159CM20208-2 CRANK ASSEMBLY				

Discrepancy: **8.36** L/H elevator hinge #2, hinge to airframe through hole (2-3) is burred on one end. Needs to be de-burred.

Resolution: Removed burr as required and remeasured L/H elevator hinge #2, hinge to airframe through hole (2-3) Hole measurement is within limits. Work performed IAW GIIB MM 27-01-00. and SRM 51-13-10.

Billing Method	Hours	Rate	Total
Labor USD: Time and Material ST	1.58	@ \$115.00	<u>\$181.70</u>
Total USD:			<u>\$181.70</u>

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: 8.37 R/H elevator hinge #4, outbd bushing (4-2) is No-Go. P/N 1159CSM20118-19.

Resolution: Removed bad hat bushing and cleaned inner surface. Prepped and installed new bushing P/N 1159CSM20118-19 IAW Drawing SE055000901 rev.T. No further action required.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$170.28
Total USD:				\$630.28

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.37 #1	1159CSM20118-19	PAD ASSY,ELEV HINGE-FLT CONT	1	\$170.28	\$170.28

Discrepancy: 8.38 R/H elevator trim tab hinge #1 bearing is frozen.

Resolution: Removed corroded R/H elevator trim tab hinge #1 bearing P/N MS14101-4 and replaced with new bearing P/N MS14101-4 IAW GAC drawing GAB14E. Checks good.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$18.85
Total USD:				\$478.85

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.38 #1	MS14101-4	BEARING,PLAIN,SELF LUB & ALIGN	1	\$18.85	\$18.85

Discrepancy: 8.39 R/H elevator trim tab hinge #2 bearing is frozen.

Resolution: Removed R/H elevator trim tab hinge #2 bearing P/N MS14101-4 and replaced with new bearing P/N MS14101-4 IAW GAC drawing GAB14E. Checks good.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$18.85
Total USD:				\$478.85

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.39 #1	MS14101-4	BEARING,PLAIN,SELF LUB & ALIGN	1	\$18.85	\$18.85

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: 8.40 R/H elevator trim tab, outbd push pull rod, aft rod end is frozen. Bearing P/N REP4M6. Nut P/N AN316-6R.

Resolution: Removed the R/H elevator trim tab outboard push pull rod's aft corroded rod end P/N UNK S/N NSN and installed a new rod end / bearing P/N MS21151-8C S/N NSN and new nut P/N AN316-6R S/N NSN. Refer to Gulfstream Drawing 1159C20455 Rev. D for work performed.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$76.48
Total USD:				\$536.48

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.40 #1	MS21151-8C	ROD END WITH KEY-WAY	1	\$76.22	\$76.22
8.40 #2	AN316-6R	NUT	1	\$0.26	\$0.26

Discrepancy: 8.41 R/H elevator trim tab #3 hinge bearing is corroded.

Resolution: Removed corroded R/H elevator trim tab hinge #3 bearing P/N MS14101-4 and replaced with new bearing P/N MS14101-4 IAW GAC drawing GAB14E. Checks good.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$18.85
Total USD:				\$478.85

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.41 #1	MS14101-4	BEARING,PLAIN,SELF LUB & ALIGN	1	\$18.85	\$18.85

Discrepancy: 8.42 R/H elevator trim tab hinge #1 bushings are corroded.

Resolution: Removed corroded R/H elevator trim tab hinge #1 lower hat bushings P/N's 1159CSM20155-1 (2 each) and replaced with new hat bushings P/N's 1159CSM20155-1 (2 each) IAW GAC Drawing # SE05000901-T. Checks good.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$122.52
Total USD:				\$582.52

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.42 #1	1159CSM20155-1	BUSHING	2	\$43.26	\$86.52
8.42 #2	GAB10F4-4	BUSHING	1	\$36.00	\$36.00

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Step: 8.42.1 **Completed** **Removed corroded R/H elevator trim tab hinge #1 lower hat bushings P/N's 1159CSM20155-1 (2 each) and replaced with new lower hat bushings P/N's 1159CSM20155-1 (2 each) IAW GAC Drawing # SE05000901-T. Checks good.**

Discrepancy: **8.43** **L/H wing, outbd flap drive rod is scored.**

Resolution: **Removed L/H wing outbd flap drive shaft . P/N 1159SCC213-17, S/N unreadable. Installed new flap shaft P/N 1159SCC213-17 , S/N 3232 IAW GIIB AMM 27-06-00. See item 8.123 for rigging of flaps.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$4,557.00
Total USD:				\$4,787.00

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
8.43 #1	1159SCC213-17	SHAFT, WING FLAP & STABILIZER	1	\$4,557.00	\$4,557.00

Discrepancy: **8.44** **L/H wing, second from outbd flap drive rod is scored.**

Resolution: **Removed L/H wing second from outbd flap drive shaft .P/N 1159SCC213-13, S/N unreadable. Installed new flap shaft P/N 1159SCC213-13, S/N 2034 IAW GIIB AMM 27-06-00. See item 8.123 for rigging of flaps.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$4,556.67
Total USD:				\$4,786.67

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
8.44 #1	1159SCC213-13	SHAFT	1	\$4,556.67	\$4,556.67

Discrepancy: **8.45** **L/H elevator trim tab hinge #1 bearing is corroded.**

Resolution: **Removed corroded L/H elevator trim tab hinge #1 bearing P/N MS14101-4 and replaced with new bearing P/N MS14101-4 IAW GAC drawing GAB14E. Checks good.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$18.85
Total USD:				\$478.85

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
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For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.45 #1	MS14101-4	BEARING,PLAIN,SELF LUB & ALIGN	1	\$18.85	\$18.85

Discrepancy: 8.46 L/H elevator trim tab hinge #2 bearing is corroded / frozen. P/N MS21232-4.

Resolution: Removed corroded L/H elevator trim tab hinge #2 bearing P/N MS14101-4 and replaced with new bearing P/N MS14101-4 IAW GAC drawing GAB14E. Checks good.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$18.85
Total USD:				\$478.85

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.46 #1	MS14101-4	BEARING,PLAIN,SELF LUB & ALIGN	1	\$18.85	\$18.85

Discrepancy: 8.47 L/H elevator trim tab hinge #3 bushings are corroded. P/N 1159CSM20155-1 and P/N 1159CSM20155-5.

Resolution: Removed corroded L/H elevator trim tab hinge #3 lower hat bushings P/N 1159CSM20155-1 and P/N 1159CSM20155-5 and replaced with new hat bushings P/N 1159CSM20155-1 and P/N 1159CSM20155-5 IAW GAC Drawing # SE05000901-T. Checks good.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$98.26
Total USD:				\$558.26

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.47 #1	1159CSM20155-1	BUSHING	1	\$43.26	\$43.26
8.47 #2	1159CSM20155-5	BUSHING	1	\$55.00	\$55.00

Discrepancy: 8.48 L/H elevator trim tab hinge #2 lower hat bushings are corroded.

Resolution: Removed corroded L/H elevator trim tab hinge #2 lower hat bushings P/N's 1159CSM20155-1 and replaced with new hat bushings P/N's 1159CSM20155-1 IAW Drawing # SE05000901-T. Checks good.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$86.52
For detail description of the work complied with see the Work Order Detail report.				

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Total USD:

\$546.52

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.48 #1	1159CSM20155-1	BUSHING	2	\$43.26	\$86.52

Discrepancy: **8.49 L/H elevator trim tab hinge #2 slip bushing is out of limits. P/N SE05000901-911P1-5 (oversized).**

Resolution: **Removed the old hardware and cleaned the attach point fittings. Greased all of the new hardware and slip bushings provided in the 72month install Kit. Installed New slip bushing P/N SE05000901-5. Work performed IAW GIIB MM 27-04-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	OT	0.27 @ \$173.00	\$46.71
Parts USD:	Time and Material			\$388.64
Total USD:				\$435.35

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.49 #1	SE05000901-911P1-5	REPAIR BUSHING	1	\$388.64	\$388.64

Discrepancy: **8.50 L/H elevator trim tab hinge #2 hole 2-1 is No-Go. (Slip bushing hinge hole) Squawk 8.49 will resolve this squawk.**

Resolution: **Removed the old hardware and cleaned the attach point fittings. Greased all of the new hardware and slip bushings provided in the 72month install Kit. Installed New slip bushing P/N SE05000901-5. Work performed IAW GIIB MM 27-04-00. REF Item 8.49**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Total USD:				\$690.00

Discrepancy: **8.51 R/H elevator trim tab hinge #3 bushings are corroded.**

Resolution: **Removed corroded R/H elevator trim tab hinge #3 lower hat bushings P/N 1159CSM20155-1and P/N 1159CSM20155-5 and replaced with new hat bushings P/N 1159CSM20155-1 and P/N 1159CSM20155-5 IAW GAC Drawing # SE05000901-T.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$98.26

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Total USD:

\$558.26

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.51 #1	1159CSM20155-1	BUSHING	1	\$43.26	\$43.26
8.51 #2	1159CSM20155-5	BUSHING	1	\$55.00	\$55.00

Discrepancy: **8.52** L/H elevator trim tab hinge #3 hole 3-1 bushing is No-Go. Bushing being replaced under squawk 8.47.

Resolution: **Removed L/H elevator trim tab hinge #3 lower hat bushing 3-1 P/N 1159CSM20155-1 and replaced with new hat bushing P/N 1159CSM20155-1 IAW GAC Drawing # SE05000901-T. Checks good. See 8.47 for corrective action, bushing replacement.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Total USD:				\$690.00

Discrepancy: **8.53** Data tags on flight controls need paint removed to see part and serial numbers.

Resolution: **Removed paint on data tags (10 ea.) on flight controls to see part and serial numbers. All work performed IAW GAMPS 4000.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Total USD:				\$460.00

Discrepancy: **8.54** CMP GIIB 321810 BLEED L WHEEL BRAKES.

See 12.51 & 12.52

Resolution: **COMPLIED WITH CMP GIIB 321810 BLEED L WHEEL BRAKES. No defects noted. Work done IAW CMP 321810 & GIIB M/M 32-04-01. Complied with GIIB CMP 321815 Brake (Right) - Bleeding Procedure. No defects noted. Work done IAW CMP 321815 & GIIB M/M 32-04-01.**

Labor USD:	No Charge			N/C
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For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: 8.55 CMP GIB 321315 Antiskid Control System - Functional Test.

See 12.51 & 12.52

Resolution: COMPLIED WITH CMP GIB 321315 Antiskid Control System - Functional Test. No defects noted. Work done IAW CMP 321315 & GIB M/M 32-04-02.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$172.50
Total USD:				\$172.50

Discrepancy: 8.56 CMP GIB 321305 Wheel Brake (GIB) - Functional Test.

See 12.51 & 12.52

Resolution: COMPLIED WITH CMP GIB 321305 Wheel Brake (GIB) - Functional Test. No defects noted. Work done IAW CMP 321305 & GIB M/M 32-04-01.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$287.50
Total USD:				\$287.50

Discrepancy: 8.57 L/H wing upper sponson plate is corroded.

Resolution: Removed sealant from around L/H wing sponson plate. All work performed IAW GIB MM 51-10-00. Treated and primed all bare metal. Resealed sponson plate. Evaluated remaining thickness of sponson after corrosion removal and found to be within limits and acceptable for continued service IAW GAC drawing SE45802903 limitations as stated in GAC engineering disposition CS278586. Performed cadmium plating repair on L/H upper sponson plate. All work done IAW Standard Repair/Corrosion of Sponson Plates DWG No. SE45802903 REV.D

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST 20.60	@ \$115.00	\$2,369.00
Labor USD:	Time and Material	OT 2.77	@ \$173.00	\$479.21
Services USD:	Time and Material			\$400.00
Total USD:				\$3,248.21

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
8.57 #1	SERVICE	1.00	\$400.00	\$400.00

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

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- Step: 8.57.1** **Completed** Removed sealant from around L/H wing sponson plate. All work performed IAW GIIB MM 51-10-00.
- Step: 8.57.2** **Completed** Treated and primed all bare metal. Resealed sponson plate. Performed cadmium plating repair on L/H upper sponson plate. All work done IAW Standard Repair/Corrosion of Sponson Plates DWG No. SE45802903 REV.D

Discrepancy: **8.58** Several spots inboard of L/H wing leading edge need sealant.

Resolution: **Replaced missing sealant IAW GIIB SRM 51-22-00.**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	ST	3.42	@	\$115.00	<u>\$393.30</u>
Total USD:						<u>\$393.30</u>

Discrepancy: **8.59** While complying with CMP#575005 , Found Hydraulic line chaffing against structure at L/H wing trailing edge outboard of flap well.

Resolution: **Repositioned Hydraulic line and wrapped with protective wrap. All work done IAW GIIB AMM 20-00-00.Checks good. No further action required.**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate					<u>\$460.00</u>
Total USD:						<u>\$460.00</u>

Discrepancy: **8.60** While complying with CMP# 575005, Found a hydraulic leak has caused the under aircraft (Keel) sections to have Bare Metal.

Resolution: **Performed leak checks and found no leaks.Spoke with customer and was informed that this was an issue of the hydraulic system being over serviced. This issue was resolved prior to this visit. Paint was stripped, prepped, primed, and painted to match under item 9.1. No further action required on this item.**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	ST	0.25	@	\$115.00	<u>\$28.75</u>
Total USD:						<u>\$28.75</u>

Discrepancy: **8.61** All fuel tank access panels removed for X-Rays require cleaning because of excessive amounts of Fuel Lube applied to panels and wing mounting surfaces.

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **Cleaned and Installed all fuel panels removed for x-ray IAW GIB AMM 57-00-00. Performed pressure check with no defects noted.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$1,150.00
Total USD:				\$1,150.00

Discrepancy: **8.62 Right winglet attach pin holes have corrosion in attach bores.**

5 places on wing and 8 on winglet. No corrosion found on pins. Corroded holes marked. Ref 12.66 inspection.

Resolution: **Accomplished right winglet sleeve bore hole repairs to the vertical (wing side) attachments and the horizontal (winglet side) attachments. All work was performed by reference of the Gulfstream Aerospace repair drawings No. SE35809903 rev. "C" entitled, Sleeve installation - winglet attachment, Cres.**

Map of vertical (wing side) repairs are as follows.

Fwd. Inbd. 2nd. oversize new set (.9370 - 9390) PN - SE35809903-19. SE35809903-39.

Center outbd. remains at original standard size.

Inbd. center & inbd. Aft , Fwd. outbd. & Aft. outbd. are at 1st. oversize new set (.9057 - .9077) PN - SE35809903-15 & SE35809903-35.

Map of horizontal (winglet side) repairs are as follows.

Top row from Fwd. to Aft. (5 total) are 1st. oversize set (.9057 - .9077) PN - SE35809903-17 & SE35809903-37.

Lower row Fwd. No. 1, 2 , 3 & 5 remain at original standard size.

No. 4 and 5 are 1st oversize set (.9057 - .9077) PN - SE35809903- 17& SE35809903-37.

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST 36.63	@ \$115.00	\$4,212.45
Labor USD:	Time and Material	OT 2.74	@ \$173.00	\$474.02
Parts USD:	Time and Material			\$6,493.49
Total USD:				\$11,179.96

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.62 #1	SE35809903-15	SLEEVE	1	\$70.00	\$70.00
8.62 #2	SE35809903-19	SLEEVE	1	\$123.60	\$123.60
8.62 #3	SE35809903-35	SLEEVE	1	\$226.00	\$226.00
8.62 #4	SE35809903-39	SLEEVE	1	\$72.10	\$72.10
8.62 #5	SE35809903-15	SLEEVE	8	\$70.00	\$560.00
8.62 #6	SE35809903-35	SLEEVE	8	\$226.00	\$1,808.00
8.62 #7	SE35809903-19	SLEEVE	1	\$123.60	\$123.60
8.62 #8	SE35809903-39	SLEEVE	1	\$72.10	\$72.10
8.62 #9	NAS1165-43L	BOLT	8	\$131.43	\$1,051.44
8.62 #10	SE35809903-17	SLEEVE	12	\$30.21	\$362.52
8.62 #11	SE35809903-37	SLEEVE	12	\$129.42	\$1,553.04
8.62 #12	NAS1165-42L	BOLT	5	\$39.00	\$195.00
8.62 #14	NAS1165E43B	BOLT	8	\$5.46	\$43.68
8.62 #15	NAS1165E40B	BOLT	3	\$18.24	\$54.72

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.62 #16	GAS551B3C9	SCREW	100	\$1.21	\$121.00
8.62 #18	PS870-B-2	SEALANT	1	\$56.69	\$56.69

Discrepancy: **8.63** While complying with CMP#575091, (Ailerons leading edge and ballast weight corrosion inspection) Found protective paint finish on both ailerons leading edge ballast's cracking/missing.

Resolution: **Removed light corrosion on ballast weight as required. Treated and primed IAW GIIB SRM 51-13-20.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Total USD:				\$460.00

Discrepancy: **8.64** While complying with CMP#575086 (Ailerons leading edge and ballast weight corrosion inspection) Found corrosion on sections 4&5 (sections noted on aileron) of LH aileron's ballast weight leading edges and attachment fasteners.

Resolution: **Removed light corrosion on ballast weight as required. Treated and primed IAW GIIB SRM 51-13-20.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Total USD:				\$460.00

Discrepancy: **8.65** While complying with CMP#575086 (Ailerons leading edge and ballast weight corrosion (LEFT) inspection) Found middle static wick on aileron cracked.

Resolution: **Installed new static wick P/N 2-16SC-1 , S/N N/A IAW GIIB AMM 23-60-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Total USD:				\$460.00

Discrepancy: **8.66** Found on the Cabin threshold a potted insert needs to be replace.
For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **Installed cabin threshold potted insert P/N TYE2001-04-08. All work performed IAW GIIB MM 25-20-00. No defects noted.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST 3.05	@ \$115.00	\$350.75
Total USD:				\$350.75

Discrepancy: **8.67 L/H flap, outbd yoke upper bushing phenolic is debonded from the bushing. Bushing needs to be replaced.**

Resolution: **Removed damaged flap "C" track yoke assembly (PN - 1159CM20621-3) and discarded. Installed a new replacement yoke assembly (PN - 1159CM 20621-3). Also removed and replaced the flap "C" track Support Assembly PN - 1159CSM20836-1) by reference of the Gulfstream IIB M/M chapter 27-06-00 entitled, Flap Actuator Yoke - Inspection. During this install, no defects were noted at this time.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$3,450.00
Parts USD:	Time and Material			\$3,147.28
Total USD:				\$6,597.28

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.67 #4	1159CSM20836-1	SUPPORT ASSY,TRACK C-ACT YOKE	1	\$628.30	\$628.30
8.67 #5	NAS1921B04-02	BLIND RIVET	300	\$1.22	\$366.00
8.67 #6	NAS1921B04-03	RIVET	200	\$1.37	\$274.00
8.67 #8	NAS1921M04-02	RIVET	300	\$3.68	\$1,104.00
8.67 #9	NAS1921M04-03	RIVET	200	\$2.85	\$570.00
8.67 #10	NAS6605-58	BOLT, HEX HEAD, CLOSE TOL, UNDRILLED	2	\$11.93	\$23.86
8.67 #11	1159CS20005-35	WASHER	2	\$89.32	\$178.64
8.67 #12	MS21042L5	NUT, SELF-LOCKING RING BASE	2	\$1.24	\$2.48

Discrepancy: **8.68 L/H elevator hinge fitting A-7 bushing ID is No-Go. New bushing (P/N 1159CSM20094-25) is included in "install kit".**

Resolution: **Removed old bushing and installed new bushing P/N 1159CSM20094 as required IAW 27-00-00. Parts included in 72 Month kit.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Total USD:				\$690.00

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **8.69** R/H aileron #1 hinge fitting inbd hole 1-3 is corroded. (Clean, Treat, and remeasure).

Resolution: **Removed corrosion, treated and remeasured the R/H aileron #1 hinge fitting inbd hole 1-3. Hole is within limits. Work performed IAW GIIB MM 27-02-00 and SRM 51-13-10.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Total USD:				\$690.00

Discrepancy: **8.70** R/H aileron #1 hinge fitting aileron to wing, hole 1-2 is corroded. (Clean, Treat, and remeasure).

Resolution: **Removed corrosion, treated and remeasured the R/H aileron #1 hinge fitting hole 1-2. Hole measurement is within limits. Work performed IAW GIIB MM 27-02-00 and SRM 51-13-10.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Total USD:				\$690.00

Discrepancy: **8.71** L/H aileron #1 hinge fitting center attach hole 1-2 is corroded. (Clean, Treat, and remeasure).

Resolution: **Removed corrosion, treated and remeasured the L/H aileron #1 hinge fitting hole 1-2. Hole measurement is within limits. Work performed IAW GIIB MM 27-02-00 and SRM 51-13-10.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Total USD:				\$690.00

Discrepancy: **8.72** L/H aileron #3 wing attach fitting bolt hole, upper and lower bushings are corroded. P/N GAB10C5-12. 2ea

Resolution: **Removed corroded L/H aileron #3 wing attach fitting upper and lower hat bushings P/N's GAB10C5-12 and replaced with new hat bushings P/N's GAB10C5-12 IAW GAC Drawing # SE05000901-T. Checks good.**

Billing Method	Hours	Rate	Total
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For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

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Labor USD:	Flat Rate	\$690.00
Parts USD:	Time and Material	\$39.34
Total USD:		\$729.34

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.72 #1	GAB10C5-12	BUSHING	2	\$19.67	\$39.34

Discrepancy: 8.73 Found small amounts of corrosion in 10 bores of the L/H winglet attach hardware holes.

Resolution: Accomplished left winglet sleeve bore hole repairs to the vertical (wing side) attachments and the horizontal (winglet side) attachments. All work was performed by reference of the Gulfstream Aerospace repair drawings No. SE35809903 rev. "C" entitled, Sleeve installation - winglet attachment, Cres.

Map of vertical (wing side) repairs are as follows.

Fwd. Inbd. & Aft inbd. 1st oversize new set (.9057 - .9077) PN - SE35809903-15, SE35809903-35.

Center inbd. remains at original standard size.

Fwd. outbd. center & Aft , are at 1st. oversize new set (.9057 - .9077) PN - SE35809903-15 & SE35809903-35.

Map of horizontal (winglet side) repairs are as follows.

Top row from Fwd. No. 1, 2, 4, and 5 are at 1st. new oversize set (.9057 - .9077) PN - SE35809903-17 & SE35809903-37.

Top row No. 3 remains at original standart size.

Lower row No. 2,3,4,&5 remain at original standard size.

Lower No. 1 is at 1st oversize new set (.9057 - .9077) PN - SE35809903-17 & SE35809903-37.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$2,300.00
Parts USD:	No Charge			N/ C
Total USD:				\$2,300.00

Discrepancy: 8.74 Left inbd flight spoiler bellcrank assy to be sent to Savannah for insp.

Resolution: Replaced the L/H inboard flight spoiler bellcrank with a "repaired" bellcrank P/N 1159SB30144-11. Refer to item 12.150 for Removal and installation of bellcrank.

	Billing Method	Hours	Rate	Total
Services USD:	Time and Material			\$1,800.86
Total USD:				\$1,800.86

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
8.74 #1	Vendor Supplied Labor	1.00	\$1,800.86	\$1,800.86

Part Ref: 1159SB30144-11 CRANK ASSY

Discrepancy: 8.75 All three A/C oxygen bottles need new "O" rings at air line nut because they are worn.

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **Removed "O" rings on all 3 oxygen cylinder air lines and installed new o-rings P/N MS22012-6 IAW G-IIB AMM 35-00-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST	6.81 @ \$115.00	\$783.15
Parts USD:	Time and Material			\$48.04
Total USD:				\$831.19

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.75 #1	MS22012-6	PACKING RING	4	\$8.70	\$34.80
8.75 #3	MS28775-006	PACKING-PORT FITTING,3/16 LINE	4	\$0.26	\$1.04
8.75 #4	MS28778-5	O-RING, TUBE FITTING PREFORMED	4	\$3.05	\$12.20

Discrepancy: **8.76 Rudder hinge #1 bearing is rough and corroded.**

Resolution: **Removed old bearing P/N MS28913-5C and replaced with a new bearing P/N MS28913-5C on rudder hinge #1 IAW GAC engineering drawing # SEO5000901 -T. Checks good.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$113.82
Total USD:				\$573.82

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.76 #1	MS28913-5C	BEARING	1	\$113.82	\$113.82

Step: 8.76.1 Completed Removed bearing retainer and bearing on rudder hinge #1 IAW GAC engineering drawing # SEO5000901 -T.

Discrepancy: **8.77 Rudder hinge #2 bearing is rough and corroded.**

Resolution: **Removed old bearing P/N MS28913-5C and replaced with new bearing P/N MS28913-5C on rudder hinge #2 IAW GAC drawing # SEO5000901-T. Checks good.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$133.07
Total USD:				\$593.07

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.77 #1	MS28913-5C	BEARING	1	\$113.82	\$113.82
8.77 #2	MS20615-5M16	RIVET, MONEL UNIVERSAL HEAD	25	\$0.77	\$19.25

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Step: 8.77.1 **Completed** Removed bearing retainer and bearing from rudder hinge #2.
Step: 8.77.2 **Completed** Removed bearing retainers and bearing from rudder hinge #3. (written in Error refer to squawk 8.78.1)

Discrepancy: **8.78** **Rudder hinge #3 bearing is rough and corroded.**

Resolution: **Removed old bearing P/N MS28913-5C and replaced with a new bearing P/N MS28913-5C on rudder hinge #3 IAW GAC drawing SEO500090-T. Checks good.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$113.82
Total USD:				\$573.82

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
8.78 #1	MS28913-5C	BEARING	1	\$113.82	\$113.82

Step: 8.78.1 **Completed** Removed bearing retainers and bearing from rudder hinge #3 IAW GAC drawing # SEO500090-T.

Discrepancy: **8.79** **Rudder hinge #1 attach bolt, airframe side, hole is oversized.**

Resolution: **After re-evaluation and further inspection of Rudder # 1 airframe hinge attach point found hinge point fitting to be within limits IAW GIIB MM 27-02-00 requirements. No further action required.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$690.00
Total USD:				\$690.00

Discrepancy: **8.80** **Rudder hinge #2 attach bolt, airframe side, hole is oversized.**

Resolution: **After re-evaluation and further inspection of Rudder # 2 airframe hinge attach point found hinge point fitting to be within limits IAW GIIB MM 27-03-00 requirements. No further action required.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$690.00
Total USD:				\$690.00

Discrepancy: **8.81** **Rudder hinge #2 rudder fitting upper hole 2-2 is No-Go.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **Removed upper hat bushing 2-2 on rudder hinge fitting #2 PN. 1159SB30210-43 and installed a new bushing PN.1159SB30210-43 IAW GAC drawing # SE05000901-T. Checks good.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Parts USD:	Time and Material			\$12.88
Total USD:				\$702.88

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.81 #1	1159SB30210-43	BUSHING	1	\$12.88	\$12.88

Discrepancy: **8.82 Rudder hinge #3 attach bolt hole 3-4 is No-Go.**

Resolution: **Accomplished rudder No. 3 hinge attachment bore hole repair by reference of the Gulfstream Structural engineering drawing No. SE05000901 rev. "N" entitled, STD - repair - control surface hinge fittings. Match cut fitting bore hole to accept a new replacement 1st oversize attachment bolt PN - NAS6604D138X (0.2647).**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$924.37
Total USD:				\$1,384.37

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.82 #1	SEO5000901-913D1-26	BUSHING	1	\$666.87	\$666.87
8.82 #2	NAS6604D138X	BOLT	1	\$257.50	\$257.50

Discrepancy: **8.83 Rudder hinge #4 attach bolt ears, hat bushings 4-2 are No-Go.**

Resolution: **Removed hat bushings on rudder hinge #4 P/N GAB10E7-6 and installed new bushings PN. GAB10E7-6 as required IAW GAC drawing # SE05000901-T. Checks good.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Parts USD:	Time and Material			\$59.74
Total USD:				\$749.74

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.83 #1	GAB10E7-6	BUSHING	2	\$29.87	\$59.74

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: 8.84 Rudder torque tube sleeve 4-3 are No-Go and corroded.

Resolution: Corrosion was found on the surface of the inner torque tube sleeve was cleaned and treated by reference of the Gulfstream Aerospace engineering drawing SE05417902 rev. "D" entitled, Material Removal Limits - Vertical Stabilizer Torque Tube. The torque tube bushing (sleeve) was replaced with new. Reference 72 mo. inspection for replacement of bushing.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Parts USD:	Time and Material			\$1,227.33
Total USD:				\$1,917.33

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.84 #1	1159CSM20091-3	SLEEVE	1	\$128.96	\$128.96
8.84 #2	1159BM20959-11	PIN	1	\$143.92	\$143.92
8.84 #3	NAS6604H4	BOLT HEX HD-CL TOL,DRILLED HD	8	\$2.99	\$23.92
8.84 #4	NAS334CP23	BOLT	5	\$3.09	\$15.45
8.84 #5	NAS334CP22	BOLT	3	\$13.39	\$40.17
8.84 #8	NAS3004-14	BOLT	2	\$330.88	\$661.76
8.84 #9	HL218-10-17	HILOK	3	\$44.45	\$133.35
8.84 #10	HL218-8-12	HILOK	3	\$26.60	\$79.80

Discrepancy: 8.85 Rh Flight spoiler actuator fwd bearing pin is burred/defective.

Resolution: Complied with by replacing the bad RH flight spoiler actuator and installing a Modified RH flight spoiler actuator P/N 1159SCH250-30, S/N 1731. Work performed IAW GIIB MM 27-02-00. REF Item 12.151

	Billing Method	Hours	Rate	Total
Labor USD:	No Charge			N/ C
Parts USD:	Time and Material			\$16,506.90
Total USD:				\$16,506.90

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.85 #2	1159SCH250-30	ACTUATOR, HYDRAULIC FLIGHT SPOILER (R/H)	1	\$16,500.00	\$16,500.00
8.85 #3	NAS1612-6A	O-RING	8	\$0.48	\$3.84
8.85 #4	NAS1612-8A	SEAL	6	\$0.51	\$3.06

Discrepancy: 8.86 Rh flight spoiler actuator rear ball bearing is seized.

Resolution: Complied with by replacing the bad RH flight spoiler actuator and installing a Modified RH flight spoiler actuator P/N 1159SCH250-30, S/N 1731. Work performed IAW GIIB MM 27-02-00. REF Item 12.151

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Labor USD: **No Charge**

N/ C

Discrepancy: **8.87 Lh Flight spoiler actuator fwd bearing pin is burred/defective.**

Resolution: **Complied with by replacing the bad LH flight spoiler actuator and installing a Modified LH flight spoiler actuator P/N 1159SCH250-29, S/N 2002. Work performed IAW GIIB MM 27-02-00. REF Item 12.148**

Labor USD: **No Charge**

N/ C

Discrepancy: **8.88 Left "C " Flap track lower inbd and outbd failed measure check after corrosion removal (Ref. to item 8.7)**

Resolution: **Reworked and refinished left "C" flap track IAW Gulfstream drawing SE05811901. Flap track is acceptable for continued normal operation per eng. despo CS278586. No further action required on this item.**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	ST	1.50	@	\$115.00	\$172.50
Total USD:						\$172.50

Discrepancy: **8.89 Rh outboard flap actuator cannon plug has 2 bare wires.**

Resolution: **Inspected the R/H outboard flap actuator cannon plug wires and found no broken wire strands. Installed heat shrink sleeve over the wire on RH outboard flap actuator cannon plug. All work was done IAW Gulfstream GIIB AMM 20-10-01.**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	ST	1.00	@	\$115.00	\$115.00
Total USD:						\$115.00

Discrepancy: **8.90 While inspecting the rudder trim actuator found it to be sticking in the full left and full right positions.**

Resolution:

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **8.91 Found the elevator Auto Pilot servo FWD cable to be damaged and its bellcrank cable guide to be deeply grooved.**

Resolution: **Removed tension and disconnected elevator autopilot servo cable and removed cable guide and inspected Cable and bell crank found no damage on cable assy. or bellcrank. Blended out groove on cable guide. Installed cable and components. All work performed IAW GIIB MM 22-00-00.**

	Billing Method		Hours		Rate	Total
Labor USD:	Time and Material	ST	3.12	@	\$115.00	\$358.80
Parts USD:	No Charge					N/ C
Total USD:						\$358.80

Step: 8.91.1 Completed Removed tension and disconnected elevator autopilot servo cable and removed cable guide and inspected Cable and bell crank found no damaged on cable assy. or bellcrank. (Cable stop is on order to be replaced.) All work performed IAW GIIB MM 22-00-00.

Discrepancy: **8.92 Found the Left AFT elevator trim cable (1159C20080-27) to be kinked when tension was released from it.**

Resolution: **Visually Inspected cable and found to have a slight bend with no sharp bend (no kinks). Inspected cable with a 10x magnifier and found no damage to any strands at the bend area. No other defects noted. All work performed IAW GIIB MM 27-00-00. No further action required pertaining to this item.**

	Billing Method		Hours		Rate	Total
Labor USD:	Time and Material	ST	1.80	@	\$115.00	\$207.00
Parts USD:	Time and Material					\$63.74
Total USD:						\$270.74

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.92 #1	1159C20080-27	CABLE ASSEMBLY (SWAGED)	1	\$63.74	\$63.74

Discrepancy: **8.93 Right "C" Flap track lower inbd and outbd failed measure check after corrosion removal (Ref. to item 8.8)**

Resolution: **Reworked and refinished right "C" flap track IAW Gulfstream drawing SE05811901. Flap track is acceptable for continued normal operation per eng. despo CS278586. No further action required on this item.**

	Billing Method		Hours		Rate	Total
Labor USD:	Time and Material	ST	1.00	@	\$115.00	\$115.00
Total USD:						\$115.00

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **8.94 Left aileron hinge #2 inner and outer slip bushings were frozen and need to be replaced.**

Resolution: **Installed two new slip bushings P/N GAB10G6-0500 on inner and outer left aileron hinge. Parts supplied with 72 mnt kit. All work done IAW GIIB AMM 27-10-00.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$460.00
Total USD:				\$460.00

Discrepancy: **8.95 Right aileron hinge #1 hole 1.1 is to small. Go gauge and bolt will not fit. Needs to be replaced.**

Resolution: **Found small burr in hole, deburred as required. Assyembled with new hardware from 72 Mnt kit. All work done IAW GIIB AMM 20-00-00 and 27-00-00.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	OT 0.27	@ \$173.00	\$46.71
Total USD:				\$46.71

Discrepancy: **8.96 Left outbd horizontal to elevator gap seal retainer, all screws are corroded.**

Resolution: **Replaced all left elevator outboard gap seal screws on horizontal rear beam P/n MS27039-1-09 al work in accordance with GIIB amm chapter 27-01-00**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	ST 2.50	@ \$115.00	\$287.50
Parts USD:	Time and Material			\$4.16
Total USD:				\$291.66

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
8.96 #1	MS27039-1-09	SCREW, MACHINE PAN HEAD STRUCTURAL	16	\$0.26	\$4.16

Discrepancy: **8.97 Left center horizontal to elevator gap seal retainer is corroded and all screws are corroded.**

Resolution: **Replaced all left elevator center gap seal screws on horizontal rear beam P/n MS27039-1-09 al work in accordance with GIIB amm chapter 27-01-00**

<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
For detail description of the work complied with see the Work Order Detail report.			

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

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Labor USD:	Time and Material	ST	2.00	@	\$115.00	\$230.00
Parts USD:	Time and Material					\$4.16
Total USD:						\$234.16

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.97 #1	MS27039-1-09	SCREW, MACHINE PAN HEAD STRUCTURAL	16	\$0.26	\$4.16

Discrepancy: 8.98 Left elevator hinge fitting bearing A-10 is grooved and needs to be replaced.

Resolution: Removed grooved Left elevator hinge fitting bearing A-10 P/N MS28913-10C and installed new Left elevator hinge fitting bearing A-10 P/N MS28913-10C IAW GAC drawing # GAB14E. Checks good.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$504.91
Total USD:				\$734.91

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.98 #1	MS28913-10C	BEARING	1	\$504.91	\$504.91

Discrepancy: 8.99 Right elevator hinge fitting bearing A-10 is grooved and needs to be replaced.

Resolution: Removed grooved Right elevator hinge fitting bearing A-10 P/N MS28913-10C and installed new Right elevator hinge fitting bearing A-10 P/N MS28913-10C IAW GAC drawing # GAB14E. Checks good.

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$504.91
Total USD:				\$734.91

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.99 #1	MS28913-10C	BEARING	1	\$504.91	\$504.91

Discrepancy: 8.100 Left inbd horizontal to elevator gap seal retainer is corroded and screws are corroded.

Resolution: Replaced all left elevator inboard gap seal screws on horizontal rear beam P/n MS27039-1-09 al work in accordance with GIIB amm chapter 27-01-00

Billing Method	Hours	Rate	Total
For detail description of the work complied with see the Work Order Detail report.			

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Bill To: HYPERION AIR, INC.

WO #: SC10420

Invoice Number: 7112000954

Labor USD:	Flat Rate	\$230.00
Parts USD:	Time and Material	\$4.42
Total USD:		\$234.42

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.100 #1	MS27039-1-09	SCREW, MACHINE PAN HEAD STRUCTURAL	17	\$0.26	\$4.42

Discrepancy: **8.101** Right outbd horizontal to elevator gap seal, all screws are corroded.

Resolution: **Replaced all right elevator outboard gap seal screws on horizontal rear beam P/n MS27039-1-09 al work in accordance with GIB amm chapter 27-01-00**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$4.16
Total USD:				\$234.16

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.101 #1	MS27039-1-09	SCREW, MACHINE PAN HEAD STRUCTURAL	16	\$0.26	\$4.16

Discrepancy: **8.102** Right center horizontal to elevator gap seal, all screws are corroded.

Resolution: **Replaced all right elevator center gap seal screws on horizontal rear beam P/n MS27039-1-09 al work in accordance with GIB amm chapter 27-01-00**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$230.00
Total USD:				\$230.00

Discrepancy: **8.103** Right inbd horizontal to elevator gap seal, all screws are corroded.

Resolution: **Replaced all right elevator inboard gap seal screws on horizontal rear beam P/n MS27039-1-09 al work in accordance with GIB amm chapter 27-01-00**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$4.42

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Total USD:

\$234.42

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.103 #1	MS27039-1-09	SCREW, MACHINE PAN HEAD STRUCTURAL	17	\$0.26	\$4.42

Discrepancy: **8.104 Right elevator hinge fitting hole A-9 inner and outer hat bushings are No-Go and need to be replaced.**

Resolution: **Removed bad hat bushings, cleaned and installed New hat bushings P/N 1159CSM20095-5 at the A9 location IAW GAC Drawing # SE05000901-T and GIIB MM 27-10-00. Ref contingency sheet.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Parts USD:	Time and Material			\$370.80
Total USD:				\$1,060.80

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.104 #1	1159CSM20095-5	BUSHING	1	\$185.40	\$185.40
8.104 #2	1159CSM20095-5	BUSHING	1	\$185.40	\$185.40

Discrepancy: **8.105 Right elevator hinge fitting bushing A-7 is corroded. New part in install kit.**

Resolution: **Replaced the right elevator hinge fitting bushing A-7 (corroded bushing) with a new bushing p/n 1159CSM20094-25. All work was done IAW G-IIB m/m 27-01-00. (Installation to accomplished under item # 12.122 CMP 272315. Ref contingency sheet.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$690.00
Parts USD:	Time and Material			\$54.59
Total USD:				\$744.59

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.105 #1	1159CSM20094-25	SLEEVE	1	\$54.59	\$54.59

Discrepancy: **8.106 LH IB Flight Spoiler Bell Crank Rod End has siezed bearing.**

Resolution: **Removed the L/H inboard flight spoiler bellcrank pushrod P/N 1159C20262-3 and installed a "New" pushrod P/N 1159C20262-3 with rod end bearings inclusive. Refer to item 12.150 for Removal and installation of push rod.**

	Billing Method	Hours	Rate	Total
For detail description of the work complied with see the Work Order Detail report.				

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

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WO #: SC10420

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Labor USD:	Flat Rate	\$115.00
Parts USD:	Time and Material	\$529.40
Total USD:		\$644.40

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
8.106 #1	1159C20262-3	PUSH ROD	1	\$529.40	\$529.40

Discrepancy: 8.107 LH OB Flight Spoiler Bell Crank Rod End has seized bearing.

Resolution: Replaced the L/H outboard flight spoiler bellcrank pushrod with "New" rod P/N 1159C20262-3. Refer to item 12.149 for Removal and installation of push rod.

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$115.00
Parts USD:	Time and Material			\$529.40
Total USD:				\$644.40

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
8.107 #1	1159C20262-3	PUSH ROD	1	\$529.40	\$529.40

Discrepancy: 8.108 RH IB Flight Spoiler Bell Crank Rod End has siezed bearing.

Resolution: Replaced the R/H inboard flight spoiler bellcrank pushrod with "New" rod P/N 1159C20262-3. Refer to item 12.152 for Removal and installation of push rod.

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$115.00
Parts USD:	Time and Material			\$529.40
Total USD:				\$644.40

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
8.108 #1	1159C20262-3	PUSH ROD	1	\$529.40	\$529.40

Discrepancy: 8.109 R/H elevator trim tab hinge #2 lower hat bushings are corroded.

Resolution: Removed corroded R/H elevator trim tab hinge #2 lower hat bushings (2 each) P/N's 1159CSM20155-1 and replaced with new hat bushings (2 each) P/N's 1159CSM20155-1 IAW GAC Drawing # SE05000901-T. Checks good.

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$460.00
For detail description of the work complied with see the Work Order Detail report.				

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Bill To: HYPERION AIR, INC.

WO #: SC10420

Invoice Number: 7112000954

Parts USD: Time and Material

\$86.52

Total USD:

\$546.52

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.109 #1	1159CSM20155-1	BUSHING	2	\$43.26	\$86.52

Discrepancy: 8.110 L/H I/B flight spoiler is missing bushing 4-3.

P/N 1159SB30210-49. Part included in "install kit".

Resolution: Removed old hardware, cleaned attach point holes and greased all New hardware and slip bushings. Installed a New bushing P/N 1159SB30210-49 provided in the 72month install kit. Work performed IAW GIIB MM 27-02-00. Ref contingency sheet.

Parts USD: No Charge

N/ C

Discrepancy: 8.111 R/H O/B flight spoiler O/B bearing is halfway out of spoiler attach point.

Resolution: Reinstalled R/H O/B flight spoiler O/B bearing back into place and secured as required IAW GIIB AMM 20-00-00. Checks good.

	Billing Method		Hours	Rate	Total
Labor USD:	Time and Material	ST	0.85	@ \$115.00	\$97.75
Total USD:					\$97.75

Discrepancy: 8.112 Right middle flap drive has grooves in it and requires changing.

Resolution: Duplicate item See item 8.117 for details.

Discrepancy: 8.113 Left O/B flight spoilers have rust on steel plate on top surfaces.

Resolution: REMOVED , light corrossin ,and rust on top surfaceO/B flihg spoiler , treated and primed top coat as required IAW GIV gamps 4000.

Billing Method	Hours	Rate	Total
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For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Bill To: **HYPERION AIR, INC.**

WO #: **SC10420**

Invoice Number: **7112000954**

Labor USD:	Time and Material	ST	7.35	@	\$115.00	\$845.25
Labor USD:	Time and Material	OT	1.50	@	\$173.00	\$259.50
Total USD:						\$1,104.75

Step: 8.113.1 **Completed** **R/H elevator , outbd top skin , has spot of corrosion, removed light corrosion and sanded as required. Area ready for primer. All work IAW Gamps 4000.**

Discrepancy: **8.114** **Right O/B flight spoilers have rust on steel plate on top surfaces.**

Resolution: **REMOVED light corrosion on spoiler top surface, treated and primed as required IAW GIV GAMPS 4000**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	ST	6.25	@	\$115.00	\$718.75
Labor USD:	Time and Material	OT	1.44	@	\$173.00	\$249.12
Total USD:						\$967.87

Discrepancy: **8.115** **L/H ground spoiler has chipping paint and corrosion in numerous locations.**

Resolution: **Sanded L/H ground spoiler [bottom surface around hinges and top inboard edge.] where paint was chipping and removed light corrosion, treated, primed, painted ,clear coat as required All work performed IAW GAC. GAMPS 4000.**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	ST	5.33	@	\$115.00	\$612.95
Labor USD:	Time and Material	OT	0.50	@	\$173.00	\$86.50
Total USD:						\$699.45

Step: 8.115.1 **Completed** **REMOVED light corrosion and paint chipping in numerous areas on L/H ground spoiler IAW GIB 51-10-00.**

Discrepancy: **8.116** **L/H flight spoiler actuator rear bearing is seized.**

Resolution: **Complied with by replacing the bad LH flight spoiler actuator and installing a Modified LH flight spoiler actuator P/N 1159SCH250-29, S/N 2002. Work performed IAW GIB MM 27-02-00. REF Item 12.148**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	No Charge					N/ C
Parts USD:	Time and Material					\$19,912.00
Total USD:						\$19,912.00

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.116 #1	1159SCH250-29	ACTUATOR, HYDRAULIC FLIGHT SPOILER (L/H)	1	\$19,912.00	\$19,912.00

Discrepancy: **8.117 R/H wing, second from outbd flap drive rod is scored.**

Resolution: **Removed R/H wing outbd flap drive shaft P/N 1159SCC213-15 , S/N unreadable. Installed new flap shaft P/N 1159SCC213-15 , S/N 2591 IAW GIB AMM 27-06-00. See item 8.123 for rigging of flaps.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$4,984.00
Total USD:				\$5,214.00

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.117 #1	1159SCC213-15	SHAFT	1	\$4,984.00	\$4,984.00

Discrepancy: **8.118 R/H O/B flight spoiler O/B hat bushings bushing 6-2 on wing attach fitting failed no go gauge (Ref 12.23) .**

Resolution: **Removed bad bushings. Prepped and installed new bushings P/N 1159SB30210-29 IAW Gulfstream Drawing SE05000901 rev.T.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$460.00
Parts USD:	Time and Material			\$64.48
Total USD:				\$524.48

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.118 #1	1159SB30210-29	BUSHING	2	\$32.24	\$64.48

Discrepancy: **8.119 R/H ground spoiler has paint missing and light corrosion in numerous locations**

Resolution: **Sanded and primed R/H ground spoiler that had paint missing, removed light corrosion ,treated and primed and painted as required IAW GAC Gamps 4000.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST 4.75 @	\$115.00	\$546.25
Total USD:				\$546.25

For detail description of the work complied with see the Work Order Detail report.

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Discrepancy: **8.120** One oxygen bottle requires new packing at attach nut.

Resolution: **Removed and replaced "o" ring on oxygen bottle attach nut. Take note that new "o" ring is made of teflon and difficult to install.P/N MS22012-6 new "o" ring was installed IAW G-II B AMM 35-00-00. Checks good.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST	0.25 @ \$115.00	\$28.75
Parts USD:	Time and Material			\$52.20
Total USD:				\$80.95

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.120 #1	MS22012-6	PACKING RING	6	\$8.70	\$52.20

Discrepancy: **8.121** L/H aileron inboard hinge has bad bearing.Bearing is falling apart.

Resolution: **Bearing was already removed for NDT.See item 12.11 for details. Removed old bearing PN. MS28913-6C and installed new bearing P/N MS28913-6C in L/H aileron inboard hinge as required IAW GAC drawing # GAB14E.**

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$230.00
Parts USD:	Time and Material			\$321.32
Total USD:				\$551.32

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.121 #2	MS20427M5-11	MONEL RIVET	20	\$1.93	\$38.60
8.121 #3	MS20427M5-18	RIVET, MONEL 100 DEGREE COUNTERSUNK	25	\$2.15	\$53.75
8.121 #4	MS28913-6C	BEARING	1	\$228.97	\$228.97

Discrepancy: **8.122** During NDT Inspections Multiple Spots Of Corrosion Were Found In L/H Center Chanel Skegg and FWD R/H Breastplate Inboard Hi-Lok Fastners.

Resolution: **Removed Hi-Lok Fastners From L/H Center Chanel Skegg and FWD R/H Breastplate Inboard, Cleaned And Removed Surface Corrosion. Installed Hi-Lok Fastners Wet With PRC 1422 B2. All Maintenance Performed In Reference To GIIB AMM 57-00-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST	12.76 @ \$115.00	\$1,467.40
Parts USD:	Time and Material			\$17.72
Total USD:				\$1,485.12

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.122 #1	HL19PB8-12	HILOK	1	\$8.24	\$8.24
8.122 #2	GAB510B4-10	BOLT	3	\$3.16	\$9.48

Discrepancy: **8.123 Flaps require rigging, due to actuator removal.**

Resolution: **Complied with flap rigging. Work done I/A/W GIBB MM chapter 27-06-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST	2.00 @ \$115.00	\$230.00
Total USD:				\$230.00

Step: 8.123.1 Completed Installed both flaps and soft rigged. No hyd. or power was available and flaps require final rigging.

Step: 8.123.2 Completed Complied with flap rigging. Work done I/A/W GIBB MM chapter 27-06-00.

Discrepancy: **8.124 NDT of all hardware over 3/8"**
 2 each P/Ns.
 NAS6610D56
 1159CM20223-1
 1159CM20264-1

Resolution: **Complied with NDT of 2 each PN. NAS6610D56, 2 each PN. 1159CM20223-1, and 2 each PN. 1159CM20264-1 bolts as required. NDT Performed by E&I under PO12-00574 / Work Order EIA12-6039. No defects noted..**

Discrepancy: **8.125 R/H flight spoiler attach hardware missing.**

Resolution: **Installed new hardware P/Ns 1159CM20265-1, AN4C11, NAS1149C0463R and MS17826-4. All work done IAW Gulfstream AMM 27-00-00.**

Parts USD: **No Charge** N/ C

Parts:

Ref. No.	Part number:	Description	Quantity
8.125 #1	1159CM20265-1	DOWEL	1
8.125 #2	AN4C11	BOLT MACHINE-CRES,DRILLED SHK	1
8.125 #3	NAS1149C0463R	WASHER-FLAT CRES	1
8.125 #4	MS17826-4	NUT	1

Discrepancy: **8.126 Corrosion found on surface of left and right wing wedges .**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **Removed surface corrosion on both left and right side surface of wing wedge areas IAW G-IIB SRM 51-13-10. Treated and primed areas IAW G-IIB SRM 51-07-10. Checks good.**

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	ST	3.50	@	\$115.00	\$402.50
Total USD:						\$402.50

Discrepancy: **8.127 Left aileron outboard attach point bushings need reamed prior to bolt installation.**

Resolution: **After further investigation found that reaming is not required. Cleaned and lubricated the attach fitting hole and bolt. Installed New bolt P/N NAS6605D50 as required. Work performed IAW GIIB MM 27-02-00. Ref contingency sheet.**

Discrepancy: **8.128 Left elev. trim tab cracked on inbd trailing edge .**

Resolution: **Removed cracked left elevator trim tab P/N 1159CSH20500-1 S/N 147, with 10,804.3 TSN, 5,663 CSN. Installed "serviceable" as removed from G-II S/N 055 trim tab P/N 1159CSH20500-1 S/N 57, with 18,947.2 TSN 7,651 CSN. All work performed IAW GIIB AMM 27-04-00. Ref squawk 12.124 for CMP 272510.**

Parts USD:	No Charge					N/ C
Parts:						
<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>			
8.128 #1	1159CSH20500-1	TAB ASSY, TRIM, ELEVATOR	1			
8.128 #2	2-16SC-1	DISCHARGER, STATIC WICK	1			

Step: 8.128.1 Completed Inspected LH Elevator trim tab hinge fittings on trim tab P/N 1159CSH20500-1 S/N T57 IAW GIIB AMM 27-04-00. Found bushings 3-1 and 3-2 to be out of tolerance. Removed bad bushings P/N 1159CSM20155-1 and P/N1159CSM20155-5 and installed new hat bushings P/N 1159CSM20155-1 and P/N 1159CSM20155-5 removed from damaged LH elevator trim tab P/N 1159CSH20500-1 S/N 147 hinge fittings 3-1 and 3-2. All work was performed IAW GAC drawing # SE05000901-T.

Discrepancy: **8.129 Perform Ops check of all flight controls as required.**

Resolution:

	<u>Billing Method</u>		<u>Hours</u>		<u>Rate</u>	<u>Total</u>
Labor USD:	Time and Material	ST	3.50	@	\$115.00	\$402.50
Total USD:						\$402.50

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **8.130** The RH elevator trim tab actuator's outboard bellows is torn

Resolution: **Removed torn boot from the Right Elevator trim tab actuator P/N: 1201-10. Installed a new actuator boot P/N: 1201-10. Work was done IAW GIB AMM Ch.27-04-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST 5.00	@ \$115.00	\$575.00
Parts USD:	Time and Material			\$796.60
Total USD:				\$1,371.60

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.130 #1	1201-10	BELLOW	1	\$796.60	\$796.60

Discrepancy: **8.131** Free air temp gauge pegs out all the time (located on center overhead panel)

Resolution: **Removed OAT sensor P/N-MS28038-1 S/N-12329, from Fwd RH nose area . Installed NEW OAT sensor P/N-MS28038-1 S/N-00138. Performed ops check IAW GIB AMM 34-02-00. No defects noted. Updated CMP 342515.**

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST 9.00	@ \$115.00	\$1,035.00
Parts USD:	Time and Material			\$1,115.60
Total USD:				\$2,438.10

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.131 #3	102-00035-1	OAT TEMP BULB	1	\$1,115.60	\$1,115.60

Step: 8.131.1 Completed Verified discrepancy removed OAT sensor connector , reading remains the same. Telling us the sensor is in a open-defective state. Removed OAT sensor from Fwd RHS nose area P/N-MS28038-1 S/N-12329 and checked on bench I.A.W. G IIB M.M. 34-02-00 Performed OAT system - Inspection Check / Bulb resistance vs Temperature Chart , and temp bulb fails. Placed OAT Temp bulb P/N-MS28038-1 S/N-12329 on order and reinstalled OAT Temp Bulb back in acft and strermered part in acft, awaiting parts delivery ETA early Am. All work performed I.A.W. GIB M.M. 34-02-00.

Discrepancy: **8.132** APU door motor intermittant.

Resolution: **Removed APU door actuator P/N GYLC9482-1 S/N 33 and installed overhauled APU door actuator P/N GYLC9482-2 S/N 592 IAW GIB AMM 49-00-00. Ops check good. See steps for details.**

Labor USD:	No Charge	N/ C	
Parts USD:	No Charge	N/ C	
Parts:			
Ref. No.	Part number:	Description	Quantity
8.132 #1	GYLC9482-2	ACTUATOR	1

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

- Step: 8.132.1** **Completed** Inspected APU door motor, wiring and linkage NO DEFECTS NOTED. Operated APU door several times with no faults noted. Could not duplicate. Ops check good. All work done IAW GIIB MM 49-10-00
- Step: 8.132.2** **Completed** Removed APU door actuator P/N GYLC9482-1 S/N 33 and installed overhauled APU door actuator P/N GYLC9482-2 S/N 592 IAW GIIB AMM 49-00-00. Actuator stops need safetied.
- Step: 8.132.3** **Completed** Safetied actuator stops on APU door actuator. Performed Ops check on APU door actuator IAW GIIB AMM 49-00-00. Ops check good.

Discrepancy: **8.133** L/H elevator trim tab actuator outboard boot is torn in several places.

Resolution:

	Billing Method	Hours	Rate	Total
Labor USD:	Time and Material	ST	0.50 @ \$115.00	\$57.50
Total USD:				\$57.50

- Step: 8.133.1** **Completed** Removed left elevator trim tab actuator mounts for access to change the bellows. Removed outboard bellow, new bellow arriving at noon on 3/22/2011. Ref GIIB MM 27-04-00.

Discrepancy: **8.134** Left main gear side brace actuator up line saturated and leaking.

Resolution:

	Billing Method	Hours	Rate	Total
Parts USD:	Time and Material			\$398.77
Total USD:				\$398.77

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
8.134 #1	1159H20200-83	HOSE ASSEMBLY	1	\$398.77	\$398.77

Item: **9** **Paint**

Team Manager: **Sean Brown**
 Customer Coordinator: **Ray Kraber**

Part/Model #:
 AC Serial No./Tail No.:
 TSN/TSO:
 CSN/CSO:
 Aircraft Time:

Discrepancy: **9.1** **Paint as required, bare metal and after inspection.**

Resolution:

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$2,500.00
Parts USD:	No Charge			N/ C

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Total USD:

\$2,500.00

Parts:

Ref. No.	Part number:	Description	Quantity
9.1 #1	P1251 HN	IMRON 5000 PHONTON GREY EFFECT	1
9.1 #2	193S	ACTIVATOR	1

Step: 9.1.1 **Waiting Insp** **SAND AREAS treated and primed as required gamps 4000**

Item: **10** **Outstanding Parts, Materials, and / or Services**

Team Manager: **Sean Brown**

Customer Coordinator: **Ray Kraber**

Part/Model #:

AC Serial No./Tail No.:

TSN/TSO:

CSN/CSO:

Aircraft Time:

Discrepancy: **10.1** **72 month kit - order in advance.**

Resolution: **Complied with**

Discrepancy: **10.2** **Outstanding services waiting.**

Resolution:

	Billing Method	Hours	Rate	Total
Services USD:	Time and Material			\$5,792.58
Total USD:				\$5,792.58

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
10.2 #4	Vendor Supplied Labor	1.00	\$2,226.72	\$2,226.72
	Part Ref: 1159CM20209-6 CRANK ASSEMBLY			
10.2 #1	SERVICE	1.00	\$3,215.86	\$3,215.86
10.2 #2	SERVICE	1.00	\$0.00	\$0.00
10.2 #3	SERVICE	1.00	\$350.00	\$350.00

Discrepancy: **10.3** **Service for man extra manpower.**

Resolution:

Services USD: **No Charge** **N/ C**

Services:

Ref. No.	Description	Quantity
10.3 #1	SERVICE	1.00

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **10.4** **Parts pulled not turned green,(check with weekend crew)**

Resolution:

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Parts USD:	Time and Material			\$609.39
Total USD:				\$609.39

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
10.4 #5	2-16SC-1	DISCHARGER, STATIC WICK	1	\$75.80	\$75.80
10.4 #3	1159CM20228-17	BUSHING	1	\$263.21	\$263.21
10.4 #2	1159CM20228-17	BUSHING	1	\$263.21	\$263.21
10.4 #1	MS29513-214	PACKING PREFORMED-O RING	1	\$0.41	\$0.41
10.4 #4	MS27039-1-09	SCREW, MACHINE PAN HEAD STRUCTURAL	16	\$0.26	\$4.16
10.4 #7	MS25083-1BB4	JUMPER, ELEC BONDING - 4 INCH	1	\$2.60	\$2.60

Discrepancy: **10.5** **Customer approved overtime.**

Resolution:

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	Flat Rate			\$12,900.13
Total USD:				\$12,900.13

Item: **12** **Customer Supplied CMP Due List**

Team Manager: **Sean Brown**

Customer Coordinator: **Ray Kraber**

Part/Model #:
AC Serial No./Tail No.:
TSN/TSO:
CSN/CSO:
Aircraft Time:

Discrepancy: **12.1** **Comply with Customer Supplied CMP Due List.**

Labor - Primary Codes \$53,550.00
Labor - Mandatory Related Codes \$32,110.00
Total Labor \$85,660.00

Resolution: **Complied with Customer Supplied CMP Due List.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

	Billing Method	Hours	Rate	Total
Labor USD:	Flat Rate			\$85,660.00
Parts USD:	Time and Material			\$14,503.49
Total USD:				\$100,163.49

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
12.1 #1	G3 72MONTH KIT 2	361,385,387,388,390 & SUB / 0386-0499	1	\$9,893.53	\$9,893.53
12.1 #3	1159CSM20091-3	SLEEVE	1	\$96.72	\$96.72
12.1 #4	PR1422-B-2	SEALANT, PRC, CARTRIDGE	4	\$38.47	\$153.88
12.1 #5	HL19PB8-8	BOLT	30	\$5.15	\$154.50
12.1 #6	HL70-8	COLLAR	50	\$0.90	\$45.00
12.1 #7	NAS1303-29D	BOLT	2	\$4.35	\$8.70
12.1 #9	GAB10G5-C0500	BUSHING	3	\$74.74	\$224.22
12.1 #10	1159CSM20094-17	SLEEVE	2	\$282.13	\$564.26
12.1 #11	MS17825-3	NUT, STEEL CASTELLATED SELF-LOCKING	2	\$2.08	\$4.16
12.1 #12	NAS1305-29H	BOLT	1	\$6.08	\$6.08
12.1 #13	NAS1149F0532P	WASHER-FLAT STEEL	1	\$0.26	\$0.26
12.1 #14	MS21042L5	NUT, SELF-LOCKING RING BASE	1	\$1.24	\$1.24
12.1 #15	NAS6604D31	BOLT HEX HD-CL TOL,DRILLED SHK	4	\$1.87	\$7.48
12.1 #16	1159CSM20118-19	PAD ASSY,ELEV HINGE-FLT CONT	1	\$170.28	\$170.28
12.1 #17	MS21151-8C	ROD END WITH KEY-WAY	1	\$76.22	\$76.22
12.1 #18	GAN512H4	NUT-COLLAR,SHEAR SELF LOCKING	50	\$0.84	\$42.00
12.1 #19	PR1440-B-2	SEALANT, PRC, CARTRIDGE	2	\$46.48	\$92.96
12.1 #20	AN525-10R26	SCREW	1	\$2.58	\$2.58
12.1 #21	NAS1149D1690H	WASHER-FLAT ALUM	1	\$0.74	\$0.74
12.1 #22	NAS6603D29	BOLT	2	\$11.60	\$23.20
12.1 #23	NAS1305-29H	BOLT	1	\$6.08	\$6.08
12.1 #24	PR1422-B-2	SEALANT, PRC, CARTRIDGE	2	\$38.47	\$76.94
12.1 #25	PR1440-B-1/2	SEALANT, PRC,CARTRIDGE	2	\$38.28	\$76.56
12.1 #26	NAS1303-27H	BOLT	2	\$12.47	\$24.94
12.1 #27	AV30-00372	CORROSION PREVENTIVE, 13.5 OUNCE AEROSOL (12/CS)	1	\$69.53	\$69.53
12.1 #28	LUBRI-BOND-220	LUBRICANT, 12 OUNCE AEROSOL SPRAY	1	\$47.87	\$47.87
12.1 #29	PR1422-B-2	SEALANT, PRC, CARTRIDGE	4	\$38.47	\$153.88
12.1 #30	PS870-B-2	SEALANT	2	\$56.69	\$113.38
12.1 #31	PR1440-B-2	SEALANT, PRC, CARTRIDGE	2	\$46.48	\$92.96
12.1 #32	PR1422-B-2	SEALANT, PRC, CARTRIDGE	2	\$38.47	\$76.94
12.1 #33	AN735D48	CLAMP-ELEC BONDING,3.00 IN	1	\$0.96	\$0.96
12.1 #34	PR1440-B-1/2	SEALANT, PRC,CARTRIDGE	1	\$38.28	\$38.28
12.1 #35	TRIBOLUBE-2N	LUBRICANT	1	\$271.92	\$271.92
12.1 #36	1201-10	BELLOW	1	\$796.60	\$796.60
12.1 #37	1201-10	BELLOW	1	\$796.60	\$796.60
12.1 #38	PR1440-B-2	SEALANT, PRC, CARTRIDGE	6	\$26.82	\$160.92
12.1 #39	NAS6605D24	BOLT, HEX HD-CLOSE TOL, DR SHANK	2	\$11.53	\$23.06

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
12.1 #40	PR1440-B-2	SEALANT, PRC, CARTRIDGE	1	\$26.82	\$26.82
12.1 #41	PR1422-B-1/2	SEALANT,PRC,CARTRIDGE	2	\$40.62	\$81.24

Discrepancy: **12.2 CMP GIIB 246020 C Inverter - Frequency / Voltage Check**

Resolution: **COMPLIED WITH CMP GIIB 246020 C Inverter - Frequency / Voltage Check with no discrepancies noted at this time. All work performed IAW G-IIB maintenance manual 24-03-00.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.3 CMP GIIB 247020 L Alternator Spline - Lubrication**

Resolution: **COMPLIED WITH CMP GIIB 247020 L Alternator Spline - Lubrication IAW GIIB MM 24-02-04. No defects noted at this time.**

	Billing Method	Hours	Rate	Total
Parts USD:	Time and Material			\$32.38
Total USD:				\$32.38

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
12.3 #1	NAS1593-214	O-RING	2	\$0.74	\$1.48
12.3 #2	KB21307	SEAL	2	\$15.45	\$30.90

Discrepancy: **12.4 CMP GIIB 247025 R Alternator Spline - Lubrication**

Resolution: **COMPLIED WITH CMP GIIB 247025 R Alternator Spline - Lubrication IAW GIIB MM 24-02-04. No defects noted at this time.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.5 CMP GIIB 252020 Life Vest(s) - Inspect / Re-Certify**

Resolution: **COMPLIED WITH CMP GIIB 252020 Life Vest(s) - Inspect / Re-Certify IAW GIIB AMM 25-00-00. See squawks 12.94 thru 12.110 for life vests status.**

Billing Method	Hours	Rate	Total
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For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Bill To: **HYPERION AIR, INC.**

WO #: **SC10420**

Invoice Number: **7112000954**

Labor USD: **No Charge**

N/ C

Services USD: **Time and Material**

\$487.69

Total USD:

\$487.69

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
<u>12.5 #1</u>	<u>Vendor Supplied Labor</u>	<u>1.00</u>	<u>\$487.69</u>	<u>\$487.69</u>

Discrepancy: **12.6** **CMP GIB 257530 Lavatory Compartment (Above Floor) - Inspection (AD 74-08-09)**

Resolution: **COMPLIED WITH CMP GIB 257530 Lavatory Compartment (Above Floor) - Inspection (AD 74-08-09). All work done per AD74-08-09-R2. No defects noted.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.7** **CMP GIB 261760 Portable Halon Fire Extinguisher (Cockpit No. 1) - Inspection**

Resolution: **COMPLIED WITH CMP GIB 261760 Portable Halon Fire Extinguisher (Cockpit No. 1) - Inspection. All work done per G-IIB amm ch 26-03-00. No defects noted**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.8** **CMP GIB 261761 Portable Halon Fire Extinguisher (Forward Cabin) - Inspection**

Resolution: **COMPLIED WITH CMP GIB 261761 Portable Halon Fire Extinguisher (Forward Cabin) - Inspection. All work done per G-IIB amm ch 26-03-00. No defects noted**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.9** **CMP GIB 261763 Portable Halon Fire Extinguisher (Baggage Compartment) - Inspection**

Resolution: **COMPLIED WITH CMP GIB 261763 Portable Halon Fire Extinguisher (Baggage Compartment) - Inspection. All work done per G-IIB amm ch 26-03-00. No defects noted.**

Labor USD: **No Charge**

N/ C

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **12.10** CMP GIIB 270145 Takeoff Alarm System - Operational Test

Resolution: **COMPLIED WITH CMP GIIB 270145 Takeoff Alarm System - Operational Test. No defects noted. All work done IAW GIIB MM Chapt 27-02-00.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.11** CMP GIIB 270210 Aileron / Hinge Fitting (Left) - Inspection

Resolution: **COMPLIED WITH CMP GIIB 270210 Aileron / Hinge Fitting (Left) - Inspection as required IAW GIIB AMM 27-02-00. Refer to squawks ,8.71,8.72 8.94, for discrepancies noted. NDT performed by Structural testing systems Cert # ES1R124K. No defects noted..**

Labor USD: **No Charge**

N/ C

- Step: 12.11.1** **Completed** Removed all three hinges from L/H aileron and cleaned.
 * Note* all hinges are in the perspective positions and clearly marked as to the inboard/out board facing side..
- Step: 12.11.2** **Completed** Complied with step (3) of part B, the removal of the bearing retainers and bearings from the three hinges on the left aileron IAW G-II B AMM 27-02-00. Ready for NDT.

Discrepancy: **12.12** CMP GIIB 270215 Aileron / Hinge Fitting (Right) - Inspection

Resolution: **COMPLIED WITH CMP GIIB 270215 Aileron / Hinge Fitting (Right) - Inspection as required IAW GIIB AMM 27-02-00. Refer to squawks 8.2,8.69,8.70,8.95 for discrepancies noted. NDT performed by Structural testing systems Cert. # ES1R124K. No defects noted..**

Labor USD: **No Charge**

N/ C

- Step: 12.12.1** **Completed** Removed all three hinges from R/H aileron and cleaned.
 * Note* all hinges are in the perspective positions and clearly marked as to the inboard facing side..
- Step: 12.12.2** **Completed** Complied with step (3) of part B, the removal of the bearing retainers and bearings from the three hinges on the right aileron IAW G-II B AMM 27-02-00. Ready for NDT.

Discrepancy: **12.13** CMP GIIB 270220 Aileron Trim Tab Hinge Fitting - Inspection

Resolution: **COMPLIED WITH CMP GIIB 270220 Aileron Trim Tab Hinge Fitting - Inspection IAW GIIB 27-04-00. No defects noted. Refer to squawk 8.20 for discrepancy noted while complying with this CMP.**

Discrepancy: **12.14** CMP GIIB 270225 Elevator Hinge Fitting (Left) - Inspection

Resolution: **COMPLIED WITH CMP GIIB 270225 Elevator Hinge Fitting (Left) - Inspection IAW GIIB 27-01-00 . Refer to squawk 8.98 for discrepancy noted.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Labor USD: No Charge

N/ C

Step: 12.14.1 **Completed** Complied with CMP GIB 270225 Elevator Hinge Fitting (Left) - Inspection stepB(3) thru B(6) IAW GIB AMM 27-01-00.

Discrepancy: **12.15** CMP GIB 270230 Elevator Hinge Fitting (Right) - Inspection

Resolution: **COMPLIED WITH CMP GIB 270230 Elevator Hinge Fitting (Right) - Inspection IAW GIB 27-01-00. No defects noted.**

Labor USD: No Charge

N/ C

Step: 12.15.1 **Completed** Complied with CMP GIB 270230 Elevator Hinge Fitting (RT - Inspection stepB(3) thru B(6) IAW GIB AMM 27-01-00.

Discrepancy: **12.16** CMP GIB 270235 Elevator Trim Tab / Hinge Fittings (Left) - Inspection

Resolution: **COMPLIED WITH CMP GIB 270235 Elevator Trim Tab / Hinge Fittings (Left) - Inspection. Discrepancies entered under item 8. All work done per G-IIB maint manual 27-04-00.**

Discrepancy: **12.17** CMP GIB 270240 Elevator Trim Tab / Hinge Fittings (Right) - Inspection

Resolution: **COMPLIED WITH CMP GIB 270240 Elevator Trim Tab / Hinge Fittings (Right) - Inspection. Discrepancies entered under item 8. All work done per G-IIB maint manual 27-04-00.**

Discrepancy: **12.18** CMP GIB 270245 Rudder / Hinge - Inspection

Resolution: **COMPLIED WITH CMP GIB 270245 Rudder / Hinge - Inspection IAW GIB AMM 27-20-00. No defects noted.**

Labor USD: No Charge

N/ C

Parts USD: No Charge

N/ C

Step: 12.18.1 **Completed** Gained access to hinge points, removed rudder hinge fittings from rudder. Performed dimensional checks of rudder and hinge fittings. Performed IAW GIB 27-03-00.

Step: 12.18.2 **Completed** Installed lower rudder hinge fitting P/N 1159CSM20025-5 as required and safetied IAW GIB AMM 27-03-00.

Discrepancy: **12.19** CMP GIB 270247 Rudder Torque Tube - Inspection

Resolution: **COMPLIED WITH CMP GIB 270247 Rudder Torque Tube - Inspection. All Maintenance Performed In Reference To GIB AMM 27-03-00.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Labor USD: No Charge

N/ C

Step: 12.19.1 Completed Complied with CMP GIB 270247 Rudder Torque Tube - Inspection, step A. through B. (10). All maintenance performed IAW GIB MM 27-03-00.

Discrepancy: **12.20** CMP GIB 270250 Inboard Flight Spoiler Hinge (Left) - Inspection

Resolution: **COMPLIED WITH CMP GIB 270250 Inboard Flight Spoiler Hinge (Left) - Inspection. Used go no go gauges IAW GIB AMM 27-02-00. Ref squawk 8.110 for discrepancies.**

Labor USD: No Charge

N/ C

Discrepancy: **12.21** CMP GIB 270255 Outboard Flight Spoiler Hinge (Left) - Inspection

Resolution: **COMPLIED WITH CMP GIB 270255 Outboard Flight Spoiler Hinge (Left) - Inspection. Used go no go gauges. No defects found IAW GIB AMM 27-02-00.**

Discrepancy: **12.22** CMP GIB 270260 Inboard Flight Spoiler Hinge (Right) - Inspection

Resolution: **COMPLIED WITH CMP GIB 270260 Inboard Flight Spoiler Hinge (Right) - Inspection. Used go no go gauges. No defects found IAW GIB AMM 27-02-00.**

Labor USD: No Charge

N/ C

Discrepancy: **12.23** CMP GIB 270265 Outboard Flight Spoiler Hinge (Right) - Inspection

Resolution: **COMPLIED WITH CMP GIB 270265 Outboard Flight Spoiler Hinge (Right) - Inspection. Use go no go gauges. IAW GIB AMM 27-02-00. Ref squawk 8.111 for discrepancies.**

Discrepancy: **12.24** CMP GIB 270271 Left Ground Spoiler Hinge - Inspection (Aircraft Having 6 Panel Speed Brakes)

Resolution: **COMPLIED WITH CMP GIB 270271 Left Ground Spoiler Hinge - Inspection (Aircraft Having 6 Panel Speed Brakes) Used go no go gauges. No defects found IAW GIB AMM 27-02-01.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **12.25** CMP GIB 270276 Right Ground Spoiler Hinge - Inspection (Aircraft Having 6 Panel Speed Brakes)

Resolution: **COMPLIED WITH CMP GIB 270276 Right Ground Spoiler Hinge - Inspection (Aircraft Having 6 Panel Speed Brakes)**
Used go no go gauges. No defects found IAW GIB AMM 27-02-01.

Labor USD: **No Charge**

N/ C

Discrepancy: **12.26** CMP GIB 270610 Rudder Pedal Controls - Lubrication

Resolution: **COMPLIED WITH CMP GIB 270610 Rudder Pedal Controls - Lubrication IAW GIB MM 27-03-00. No defects noted at this time.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.27** CMP GIB 270618 Flap Actuator Bushing (Left Inboard) - Lubrication

Resolution: **COMPLIED WITH CMP GIB 270618 Flap Actuator Bushing (Left Inboard) - Lubrication. Lubrication. Work IAW GIB MM 27-06-00**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.28** CMP GIB 270619 Flap Actuator Bushing (Left Outboard) - Lubrication

Resolution: **COMPLIED WITH CMP GIB 270619 Flap Actuator Bushing (Left Outboard) - Lubrication. Work IAW GIB mm 27-06-00**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.29** CMP GIB 270622 Flap Actuator Bushing (Right Inboard) - Lubrication

Resolution: **COMPLIED WITH CMP GIB 270622 Flap Actuator Bushing (Right Inboard) - Lubrication IAW GIB MM 27-06-00. No defects noted at this time.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Step: 12.29.1 **Completed** **Complied with CMP GIB 270622 Flap Actuator Bushing (Right Inboard) - Lubrication, step A.through B.(2). All maintenance performed IAW GIB MM 27-06-00.**

Step: 12.29.2 **Completed** **Complied with CMP GIB 270622 Flap Actuator Bushing (Right Inboard) - IAW GIB MM 27-06-00. No defects noted at this time.**

Discrepancy: 12.30 **CMP GIB 270623 Flap Actuator Bushing (Right Outboard) - Lubrication**

Resolution: COMPLIED WITH CMP GIB 270623 Flap Actuator Bushing (Right Outboard) - Lubrication IAW GIB MM 27-06-00. No defects noted at this time.

Step: 12.30.1 **Completed** **Complied with CMP GIB 270623 Flap Actuator Bushing (Right Outboard) - Lubrication. Step A. through B.(2). All maintenance performed IAW GIB MM 27-06-00.**

Step: 12.30.2 **Completed** **Complied with CMP GIB 270623 Flap Actuator Bushing (Right Outboard) - IAW GIB MM 27-06-00. No defects noted at this time.**

Discrepancy: 12.31 **CMP GIB 270635 Control Surface Hinge Joints - Lubrication**

Resolution: COMPLIED WITH CMP GIB 270635 Control Surface Hinge Joints - Lubrication. All Maintenance Performed In Reference To GIB AMM 27-00-00.

Discrepancy: 12.32 **CMP GIB 270720 Elevator Trim Tab Actuator (Left) - Lubrication**

Resolution: COMPLIED WITH CMP GIB 270720 Elevator Trim Tab Actuator (Left) - Lubrication. All Maintenance Performed In Reference To GIB AMM 27-04-00.

Discrepancy: 12.33 **CMP GIB 270725 Elevator Trim Tab Actuator (Right) - Lubrication**

Resolution: COMPLIED WITH CMP GIB 270725 Elevator Trim Tab Actuator (Right) - Lubrication. Work done IAW CMP 270725 & GIB M/M 27-04-00.

Discrepancy: 12.34 **CMP GIB 271005 Aileron Boost Bell Crank (Left) - Inspection**

Resolution: COMPLIED WITH CMP GIB 271005 Aileron Boost Bell Crank (Left) - Inspection as required IAW GIB MM 27-02-00.No defects noted. Ref contingency sheet.

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Labor USD: No Charge

N/ C

Step: 12.34.1 **Completed** Complied with section (B) 1-4 in procedure; the measurements are .625 on upper ID of shaft hole, lower bushing ID is .625 and OD is .748 and lower hole size ID is .625 all dimensions are within limits IAW GIB AMM.

Discrepancy: **12.35** **CMP GIB 271012 Aileron Actuator (Left) - Inspection**

Resolution: **COMPLIED WITH CMP GIB 271012 Aileron Actuator (Left) - Inspection and wing attach fittings measured within limits of .250-.252 IAW G-IIB AMM 27-02-00. No defects found.**

Discrepancy: **12.36** **CMP GIB 271014 Aileron Boost Bell Crank (Right) - Inspection**

Resolution: **Complied with CMP GIB 271014 Aileron Boost Bell Crank (Right) - Inspection as required IAW GIB MM 27-02-00. Refer to squawk 8.27 for discrepancy noted.**

Labor USD: No Charge

N/ C

Step: 12.36.1 **Completed** Complete steps A-B(4) IAW GIB AMM 27-02-00.

Step: 12.36.2 **Completed** Complete steps A through B(4) IAW GIB AMM 27-02-00

Discrepancy: **12.37** **CMP GIB 271017 Aileron Actuator (Right) - Inspection**

Resolution: **COMPLIED WITH CMP GIB 271017 Aileron Actuator (Right) - Inspection IAW GIB AMM 27-02-00. Ref to squawk 8.32 for discrepancies.**

Labor USD: No Charge

N/ C

Step: 12.37.1 **Completed** Complete steps A through B-(3) IAW GIB AMM 27-02-00

Discrepancy: **12.38** **CMP GIB 271021 Elevator Actuator - Inspection**

Resolution: **COMPLIED WITH CMP GIB 271021 Elevator Actuator - Inspection IAW GIB MM 27-01-00. No defects noted at this time.**

Labor USD: No Charge

N/ C

Discrepancy: **12.39** **CMP GIB 271311 Rudder Actuator and Forward Attach Fitting - Inspection**

Resolution: **COMPLIED WITH CMP GIB 271311 Rudder Actuator and Forward Attach Fitting - Inspection as required. No discrepancies found at this time. All work performed I.A.W. GIB M.M. 27-03-00**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **12.40** CMP GIB 271314 Rudder Trim Tab Actuator - Inspection

Resolution: **COMPLIED WITH CMP GIB 271314 Rudder Trim Tab Actuator - Inspection as required. No discrepancies found at this time. All work performed I.A.W. GIB M.M. 27-04-00.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.41** CMP GIB 271511 Aileron Trim Tab Actuator - Inspection

Resolution: **COMPLIED WITH CMP GIB 271511 Aileron Trim Tab Actuator - Inspection No defects found IAW GIB AMM 27-04-00.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.42** CMP GIB 274011 Flap Actuator Yoke (Left Outboard) - Inspection

Resolution: **COMPLIED WITH CMP GIB 274011 Flap Actuator Yoke (Left Outboard) - Inspection. Work IAW GIB MM 27-40-00**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.43** CMP GIB 274012 Flap Actuator Yoke (Left Inboard) - Inspection

Resolution: **COMPLIED WITH CMP GIB 274012 Flap Actuator Yoke (Left Inboard) - Inspection. Work IAW GIB MM 27-40-00.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.44** CMP GIB 274311 Flap Actuator Yoke (Right Outboard) - Inspection

Resolution: **COMPLIED WITH CMP GIB 274311 Flap Actuator Yoke (Right Outboard) - Inspection IAW GIB AMM 27-06-00. See item 8.16 for discrepancy noted.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Labor USD: No Charge

N/ C

Step: 12.44.1 Completed Complied with CMP GIB 274311 Flap Actuator Yoke (Right Outboard) - Inspection. Step A. through B.(2). All maintenance performed IAW GIB MM 27-06-00.

Discrepancy: **12.45** CMP GIB 274312 Flap Actuator Yoke (Right Inboard) - Inspection

Resolution: **COMPLIED WITH CMP GIB 274312 Flap Actuator Yoke (Right Inboard) - Inspection IAW GIB AMM 27-06-00.No defects noted.**

Labor USD: No Charge

N/ C

Step: 12.45.1 Completed Complied with CMP GIB 274312 Flap Actuator Yoke (Right Inboard) - Inspection. Step A through B.(4). All maintenance performed IAW GIB MM 27-06-00.

Discrepancy: **12.46** CMP GIB 274813 Ground Spoiler Actuator (Left) - Inspection (Aircraft Having 6 Panel Speed Brakes)

Resolution: **COMPLIED WITH CMP GIB 274813 Ground Spoiler Actuator (Left) - Inspection (Aircraft Having 6 Panel Speed Brakes) No defects found IAW GIB AMM 27-01-00.**

Labor USD: No Charge

N/ C

Discrepancy: **12.47** CMP GIB 274818 Ground Spoiler Actuator (Right) - Inspection (Aircraft Having 6 Panel Speed Brakes)

Resolution: **COMPLIED WITH CMP GIB 274818 Ground Spoiler Actuator (Right) - Inspection (Aircraft Having 6 Panel Speed Brakes) No defects found IAW GIB AMM 27-02-00**

Labor USD: No Charge

N/ C

Discrepancy: **12.48** CMP GIB 275312 Flight Spoiler Actuator (Left) - Inspection

Resolution: **COMPLIED WITH CMP GIB 275312 Flight Spoiler Actuator (Left) - Inspection IAW GIB MM 27-02-00. See squawk item 8.87 for discrepancy corrective action.**

Labor USD: No Charge

N/ C

Discrepancy: **12.49** CMP GIB 275317 Flight Spoiler Actuator (Right) - Inspection

Resolution: **COMPLIED WITH CMP GIB 275317 Flight Spoiler Actuator (Right) - Inspection IAW GIB MM 27-02-00. See squawk items 8.85 / 8.86 for discrepancy corrective actions.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Labor USD: **No Charge**

N/ C

Discrepancy: **12.50 CMP GIIB 280510 Fuel Control System - Functional Test**

Resolution: **COMPLIED WITH CMP GIIB 280510 Fuel Control System - Functional Test IAW 28-05-00. No defects noted.**

Discrepancy: **12.51 CMP GIIB 321210 Left Anti-Skid Valve Filter - Inspection**

Resolution: **COMPLIED WITH CMP GIIB 321210 Left Anti-Skid Valve Filter - Inspection IAW G2-B AMM 32-04-02 with no defects noted.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	No Charge			N/ C
Parts USD:	Time and Material			\$0.91
Total USD:				\$0.91

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
12.51 #1	NAS1612-4A	SEAL, 0.250 INCH OD TUBING	1	\$0.43	\$0.43
12.51 #2	NAS1612-6A	O-RING	1	\$0.48	\$0.48

Discrepancy: **12.52 CMP GIIB 321215 Right Anti-Skid Valve Filter - Inspection**

Resolution: **COMPLIED WITH CMP GIIB 321215 Right Anti-Skid Valve Filter - Inspection IAW G2-B AMM 32-04-02 with no defects noted.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	No Charge			N/ C
Parts USD:	Time and Material			\$0.91
Total USD:				\$0.91

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
12.52 #1	NAS1612-4A	SEAL, 0.250 INCH OD TUBING	1	\$0.43	\$0.43
12.52 #2	NAS1612-6A	O-RING	1	\$0.48	\$0.48

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: 12.53 CMP GIB 330200 Emergency Exit Light Battery - Scrap

Resolution: COMPLIED WITH CMP GIB 330200 Emergency Exit Light Battery - Scrap IAW GIB AMM 33-02-00.

Discrepancy: 12.54 CMP GIB 331510 Emergency Exit Light Battery - Replacement

Resolution: COMPLIED WITH CMP GIB 331510 Emergency Exit Light Battery - Replacement IAW GIB AMM 33-02-00. See item 6.1 for discrepancy noted.

Labor USD: No Charge

N/ C

Discrepancy: 12.55 CMP GIB 341524 Attitude Indicator Battery No. 1 - Service (PS 823)

Resolution: COMPLIED WITH CMP GIB 341524 Attitude Indicator Battery No. 1 - Service (PS 823) IAW GIB AMM 34-04-00. No defects noted. AVIALL WO43520715. Removed battery P/N 501-1075-02 S/N 2060. Installed TESTED unit P/N 501-1075-02 S/N 2060. Updated CMP's 341530 and 341524.

	Billing Method	Hours	Rate	Total
Labor USD:	No Charge			N/ C
Services USD:	Time and Material			\$1,432.86
Total USD:				\$1,432.86

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
12.55 #1	SERVICE	1.00	\$1,432.86	\$1,432.86

Part Ref: 501-1075-02 POWER SUPPLY

Step: 12.55.1 Completed Removed Attitude Indicator battery P/N 501-1075-02 S/N 2060 for service. All work accomplished IAW GIB AMM 34-04-00.

Discrepancy: 12.56 CMP GIB 351615 Oxygen Cylinder (No. 2) - Hydrostatic Test

Resolution: COMPLIED WITH CMP GIB 351615 Oxygen Cylinder (No. 2) - Hydrostatic Test oxygen cylinder. Hydro complied with by Ameron Global CRS RU2R148L on 2/10/12. Next Hydro due 2/2015. See item 12.154 for P/N and S/N information.

	Billing Method	Hours	Rate	Total
Labor USD:	No Charge			N/ C
Services USD:	Time and Material			\$250.00
Total USD:				\$250.00

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
12.56 #1	Vendor Supplied Labor	1.00	\$250.00	\$250.00

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
Part Ref: 176521-64	CYLINDER 64 CUBIC FT.			

Discrepancy: **12.57** CMP GIB 351620 Oxygen Cylinder (No. 3) - Hydrostatic Test

Resolution: **COMPLIED WITH CMP GIB 351620 Oxygen Cylinder (No. 3) - Hydrostatic Test oxygen cylinder. Hydro complied with by Ameron Global CRS RU2R148L on 2/10/12. Next Hydro due 2/2015. See item 12.158 for P/N and S/N information.**

	Billing Method	Hours	Rate	Total
Labor USD:	No Charge			N/ C
Services USD:	Time and Material			\$250.00
Total USD:				\$250.00

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
12.57 #1	Vendor Supplied Labor	1.00	\$250.00	\$250.00
Part Ref: 176521-49	BOTTLE, OXYGEN			

Discrepancy: **12.58** CMP GIB 351623 Portable Oxygen Bottle (No. 1) - Service / Hydrostatic Test

Resolution: **COMPLIED WITH CMP GIB 351623 Portable Oxygen Bottle (No. 1) - Service / Hydrostatic Test all work done by Ameron Global Product Support , tracking reference # 255045. Next due date 02/2017.**

	Billing Method	Hours	Rate	Total
Services USD:	Time and Material			\$250.00
Total USD:				\$250.00

Services:

Ref. No.	Description	Quantity	Sell Price USD	Extended USD
12.58 #1	Vendor Supplied Labor	1.00	\$250.00	\$250.00
Part Ref: 900-002-035-01	portable O2 bottle			

Discrepancy: **12.59** CMP GIB 380101 ELEMENT,FRESH WATER SYS FILTER

Resolution: **COMPLIED WITH CMP GIB 380101 ELEMENT,FRESH WATER SYS FILTER. Removed filter P/N AL-11-OG NSN and installed new filter AL-11-OG NSN. Work IAW GIB 38-00-00**

	Billing Method	Hours	Rate	Total
Labor USD:	No Charge			N/ C
Parts USD:	Time and Material			\$59.51
Total USD:				\$59.51

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
12.59 #1	AL-11-0G	FILTER	1	\$59.51	\$59.51

Discrepancy: **12.60 CMP GIB 531535 Shear Clips Aft Fuselage - Inspection**

Resolution: **COMPLIED WITH CMP GIB 531535 Shear Clips Aft Fuselage - Inspection. No defects noted at this time. Performed IAW GIB MM 53-00-00.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.61 CMP GIB 531545 Static Ports (RVSM) - Inspection**

Resolution: **COMPLIED WITH CMP GIB 531545 Static Ports (RVSM) - Inspection IAW GIB AMM 53-00-00. No defects noted.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.62 CMP GIB 534015 Main Power Junction Box Interior - Inspection**

Resolution: **COMPLIED WITH CMP GIB 534015 Main Power Junction Box Interior - Inspection. All work was done IAW Gulfstream GIB AMM 53-40-15. No defects noted.**

Discrepancy: **12.63 CMP GIB 539010 Main Landing Gear Installation (Left) - Inspection**

Resolution: **COMPLIED WITH CMP GIB 539010 Main Landing Gear Installation (Left) - Inspection, IAW GIB M/M Chapter 53-00-00. No defects noted at this time.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.64 CMP GIB 539020 Main Landing Gear Installation (Right) - Inspection**

Resolution: **COMPLIED WITH CMP GIB 539020 Main Landing Gear Installation (Right) - Inspection IAW GIB MM 53-00-00. No defects noted at this time.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Labor USD: **No Charge**

N/ C

Discrepancy: **12.65 CMP GIIB 553035 Rudder Leading Edge and Ballast Weight Corrosion - Inspection**

Resolution: **COMPLIED WITH CMP GIIB 553035 Rudder Leading Edge and Ballast Weight Corrosion - Inspection IAW GII-B AMM 55-00-00. No defects noted.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	No Charge			N/ C
Parts USD:	Flat Rate			\$2,500.00
Total USD:				\$2,500.00

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>
12.65 #1	NAS1919M05-04	RIVET	200
12.65 #2	NAS1919M05-05	RIVET	300
12.65 #3	NAS1919M05-06	RIVET	200
12.65 #4	NAS1919M04-03	RIVET	100
12.65 #5	NAS1919M05-03	RIVET	38
12.65 #6	NAS1919M05-03	RIVET	62
12.65 #7	PR1440-B-1/2	SEALANT, PRC,CARTRIDGE	1

Step: 12.65.1 Completed Complied with rudder ballast weight inspection section (B) 1-5, IAW G-II B AMM 55-00-00. Found no corrosion on ballast weights or cracks on the structure for the ballast weights.

Discrepancy: **12.66 CMP GIIB 575002 Winglet Attach Hardware (Right) - Inspection**

Resolution: **COMPLIED WITH CMP GIIB 575002 Winglet Attach Hardware (Right) - Inspection. IAW GIIB AMM 57-00-00. Ref squawk 8.62 for discrepancies.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	No Charge			N/ C
Parts USD:	Time and Material			\$127.00
Total USD:				\$127.00

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
12.66 #1	6856K	GREASE, MASTINOX	1	\$127.00	\$127.00

Discrepancy: **12.67 CMP GIIB 575003 Winglet Attach Hardware (Left) - Inspection**

Resolution: **COMPLIED WITH CMP GIIB 575003 Winglet Attach Hardware (Left) - Inspection IAW G-II AMM 57-00-00. Found 10 bores with corrosion in them see squawk 8.73 for repair information.**

Labor USD: **No Charge**

N/ C

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Step: 12.67.1 **Completed** Complied with section (B) step1-3 . Found 10 winglet bores with some corrosion see squawk 8.73 .

Discrepancy: **12.68** **CMP GIB 575005 Wing Exterior (Left) - Inspection (GIB)**

Resolution: **COMPLIED WITH CMP GIB 575005 Wing Exterior (Left) - Inspection (GIB) IAW GIB MM 57-00-00. Discrepancies were entered under item 8.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.69** **CMP GIB 575010 Wing Exterior (Right) - Inspection (GIB)**

Resolution: **COMPLIED WITH CMP GIB 575010 Wing Exterior (Right) - Inspection (GIB) IAW GIB MM 57-00-00. Refer to Squawks 8.14 for discrepancy noted .**

Labor USD: **No Charge**

N/ C

Step: 12.69.1 **Completed** Complied with CMP# 575010 sections 1-5 IAW GIB MM 57-00-00. Sections 6-10 still remain to be completed.

Step: 12.69.2 **Completed** Complied with CMP# 575010 R/H wing exterior inspection IAW GIB MM 57-00-00. Go to Squawks 8.14 for discrepancy corrective action.

Discrepancy: **12.70** **CMP GIB 575025 Wing to Fuselage Interior (Left) - Inspection (GIB)**

Resolution: **COMPLIED WITH CMP GIB 575025 Wing to Fuselage Interior (Left) - Inspection (GIB) IAW GIB MM 57-00-00. Go to Squawk 8.2 for discrepancy corrective action.**

Labor USD: **No Charge**

N/ C

Step: 12.70.1 **Completed** Completed CMP# 575025 wing to fuselage interior (Left) inspection Sections A 1&2 , and section B subsection 1 a through i IAW GIB MM 57-00-00. Go to Squawk 8.2 for discrepancy corrective action.

Discrepancy: **12.71** **CMP GIB 575030 Wing to Fuselage Interior (Right) - Inspection (GIB)**

Resolution: **COMPLIED WITH CMP GIB 575030 Wing to Fuselage Interior (Right) - Inspection (GIB) IAW GIB MM 57-00-00. Go to Squawk 8.2 for discrepancy corrective action.**

Labor USD: **No Charge**

N/ C

Step: 12.71.1 **Completed** Began CMP# 575030 wing to fuselage interior inspection IAW GIB MM 57-00-00. Note: Screws of access panels are rusted/corroded.

Step: 12.71.2 **Completed** Completed CMP# 575030 wing to fuselage interior (Right) inspection Sections A 1&2 , and section B subsection 1 a through i IAW GIB MM 57-00-00. Go to Squawk 8.2 for discrepancy corrective action.

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **12.72** CMP GIB 575049 Wing Plank Fastener / Radius Block Removal (Left) - Corrosion Inspection (GIB)

Resolution: **COMPLIED WITH CMP GIB 575049 Wing Plank Fastener / Radius Block Removal (Left) - Corrosion Inspection (GIB) IAW GIB 57-00-00. Work performed by Structural testing systems Cert. # ES1R124K.refer to squawk 8.122 for discrepancy noted.**

Labor USD: **No Charge** N/ C
 Parts USD: **No Charge** N/ C

- Step: 12.72.1** **Completed** Removed Fuel panels 21-FT LWR-1, 21-FT LWR-2, 21-FT LWR-3, 21-FT LWR-4, 22-FT LWR-1, 22FT LWR-2. 22-FT LWR-3, 22-FT LWR-4, 22-FT UPR-12, 22-FT UPR-13, 22-FT UPR-15, 22 FT UPR-17. Done I.A.W. GIB 57-00-00.
- Step: 12.72.2** **Completed** Began removing Hi-loc fasteners IAW GIB AMM 57-00-00.
- Step: 12.72.3** **Completed** Finished removing Hi-Loc fasteners and complied with CMP GIB 575049 Wing Plank Fastener / Radius Block Removal (Left) - Corrosion Inspection (GIB) Step 2(a).
- Step: 12.72.4** **Completed** Completed Inspection By Structural Testing Systems (Cert. # ES1R124K). Refer To Squawk 8.122 For Discrepancy Noted. Completed Re-installation Of Hi-Loc Fasteners And Complied With (CMP GIB 575049) Wing Plank Fastener / Radius Block Removal (Left) - Corrosion Inspection. All Maintenance Performed In Reference To GIB AMM 57-00-00.

Discrepancy: **12.73** CMP GIB 575054 Wing Plank Fastener / Radius Block Removal (Right) - Corrosion Inspection (GIB)

Resolution: **COMPLIED WITH CMP GIB 575054 Wing Plank Fastener / Radius Block Removal (Right) - Corrosion Inspection (GIB) IAW GIB 57-00-00. Work performed by Structural testing systems Cert. # ES1R124K. No defects noted..**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	No Charge			N/ C
Parts USD:	Time and Material			<u>\$533.69</u>
Total USD:				<u>\$533.69</u>

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
12.73 #1	HL70-8	COLLAR	220	\$0.90	\$198.00
12.73 #2	GAB510B4-13	BOLT	3	\$2.83	\$8.49
12.73 #3	GAB510B4-7	BOLT	10	\$1.90	\$19.00
12.73 #4	GAB510B4-8	BOLT	10	\$22.57	\$225.70
12.73 #5	GAB510B4-9	BOLT	10	\$2.07	\$20.70
12.73 #6	GAB510B4-6	BOLT	10	\$6.18	\$61.80

- Step: 12.73.1** **Completed** Removed Fuel panels 21-FT LWR-1, 21-FT LWR-2, 21-FT LWR-3, 21-FT LWR-4, 22-FT LWR-1, 22FT LWR-2. 22-FT LWR-3, 22-FT LWR-4, 22-FT UPR-12, 22-FT UPR-13, 22-FT UPR-15, 22 FT UPR-17. Done I.A.W. GIB 57-00-00.
- Step: 12.73.2** **Completed** Began removing Hi-loc fasteners IAW GIB AMM 57-00-00.
- Step: 12.73.3** **Completed** Finished removing Hi-Loc fasteners and complied with CMP GIB 575054 Wing Plank Fastener / Radius Block Removal (Left) - Corrosion Inspection (GIB) Step 2(a).
- Step: 12.73.4** **Completed** Completed Inspection By Structural Testing Systems (Cert. # ES1R124K). Refer To Squawk 8.122 For Discrepancy Noted. Completed Re-installation Of Hi-Loc Fasteners And Complied With (CMP GIB 575049) Wing Plank Fastener / Radius Block Removal (Left) - Corrosion Inspection. All Maintenance Performed In Reference To GIB AMM 57-00-00.

Discrepancy: **12.74** CMP GIB 575065 Wing Rear Beam (Left) - Inspection (GIB)

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **COMPLIED WITH CMP GIB 575065 Wing Rear Beam (Left) - Inspection (GIB). Work IAW GIB MM57-00-00**

Discrepancy: **12.75 CMP GIB 575070 Wing Rear Beam (Right) - Inspection (GIB)**

Resolution: **COMPLIED WITH CMP GIB 575070 Wing Rear Beam (Right) - Inspection (GIB).Work IAW GIB MM57-00-00**

Discrepancy: **12.76 CMP GIB 575086 Aileron Leading Edge and Ballast Weight Corrosion (Left) - Inspection (GIB)**

Resolution: **COMPLIED WITH CMP GIB 575086 Aileron Leading Edge and Ballast Weight Corrosion (Left) - Inspection (GIB) IAW GIB MM 57-00-00. See Squawk items 8.63/8.64 and 8.65 for discrepancy corrective actions.**

Labor USD: **No Charge** N/ C

- Step: 12.76.1** **Completed** Began CMP#575086 Aileron leading edge and ballast weight corrosion (LEFT) inspection Sections A #'s 1-2, and section B #'s 1-6 IAW GIB MM 57-00-00. See Squawk items 8.63/8.64 and 8.65 for discrepancy corrective actions.
- Step: 12.76.2** **Completed** Complied with CMP#575086 Aileron leading edge and ballast weight corrosion (LEFT) inspection IAW GIB MM 57-00-00. See Squawk items 8.63/8.64 and 8.65 for discrepancy corrective actions.

Discrepancy: **12.77 CMP GIB 575091 Aileron Leading Edge and Ballast Weight Corrosion (Right) - Inspection (GIB)**

Resolution: **COMPLIED WITH CMP GIB 575091 Aileron Leading Edge and Ballast Weight Corrosion (Right) - Inspection (GIB) IAW GIB MM 57-00-00. See squawk item 8.63 for discrepancy corrective actions.**

Labor USD: **No Charge** N/ C

Discrepancy: **12.78 CMP GIB 575105 Wing Wedge (Left) - Inspection (GIB)**

Resolution: **COMPLIED WITH CMP GIB 575105 Wing Wedge (Left) - Inspection (GIB) IAW GIB 57-00-00. Work performed by Structural testing systems Cert. # ES1R124K. See item 8.126 for discrepancy noted..**

Labor USD: **No Charge** N/ C

- Step: 12.78.1** **Completed** Complied with section (B) 1, removed L/H wing wedge IAW G-IIB AMM 57-00-00.

Discrepancy: **12.79 CMP GIB 575110 Wing Wedge (Right) - Inspection (GIB)**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **COMPLIED WITH CMP GIB 575110 Wing Wedge (Right) - Inspection (GIB) IAW GIB 57-00-00. Work performed by Structural testing systems Cert. # ES1R124K. See item 8.126 for discrepancy noted.**

Labor USD: **No Charge**

N/ C

Step: 12.79.1 Completed Completed steps A,B and C removing right wing wedge panel as required IAW GIB AMM 57-00-00. REF Squawk 8.6 for discrepancy noted.

Discrepancy: **12.80 CMP GIB 575115 Wing Sponson Rib (Left) - Inspection (GIB)**

Resolution: **COMPLIED WITH CMP GIB 575115 Wing Sponson Rib (Left) - Inspection (GIB). Work IAW GIB MM 57-51-00**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.81 CMP GIB 575120 Wing Sponson Rib (Right) - Inspection (GIB)**

Resolution: **COMPLIED WITH CMP GIB 575120 Wing Sponson Rib (Right) - Inspection (GIB) All work done per G-IIB amm ch 05-01-00. No defects noted.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.82 CMP GIB 731510 TEST ENG TEMP CONT SYS**

Resolution: **COMPLIED WITH CMP GIB 731510 TEST ENG TEMP CONT SYS with no faults noted. All work done IAW RR Spay 511-8 EMM 5-10.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.83 CMP GIB 732010 Engine Temperature Control System - Functional Test**

Resolution: **COMPLIED WITH CMP GIB 732010 Engine Temperature Control System - Functional Test with no faults noted. All work done IAW RR Spay 511-8 EMM 5-10.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.84 CMP GIB 733015 ENG LP GOV RESET CK**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **Performed CMP code 733015 as per GIBB Rolls Royce EMM 5-10. No defects were noted.**

Discrepancy: **12.85 CMP GIBB 733515 ENG LP GOV RESET CK**

Resolution: **COMPLIED WITH CMP GIBB 733515 ENG LP GOV RESET CK as per Rolls Royce EMM 5-10. No defects were noted.**

Discrepancy: **12.86 CMP GIBB 771510 ENG TGT CIRCUIT RES CHECK**

Resolution: **COMPLIED WITH CMP GIBB 771510 ENG TGT CIRCUIT RES CHECK. No defects noted. Work performed IAW GAC GIBB MM 77-04-00.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.87 CMP GIBB 771515 ENG TGT CIRCUIT RES CHECK**

Resolution: **COMPLIED WITH CMP GIBB 771515 ENG TGT CIRCUIT RES CHECK. No defects noted. Work performed IAW GAC GIBB MM 77-04-00.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.88 CMP GIBB 801520 Engine Starter (Left) - Oil Change**

Resolution: **COMPLIED WITH CMP GIBB 801520 Engine Starter (Left) - Oil Change with no defects noted IAW G2-B AMM 80-01-00.**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	No Charge			N/ C
Parts USD:	Time and Material			\$115.28
Total USD:				\$115.28

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
12.88 #1	S9809-6	GASKET "C"	1	\$3.71	\$3.71
12.88 #2	S9809-3	GASKET S8995-3	2	\$8.84	\$17.68

For detail description of the work complied with see the Work Order Detail report.

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Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
12.88 #5	BE20057	SEAL	2	\$39.94	\$79.88
12.88 #4	KB21614	SEAL	1	\$14.01	\$14.01

Discrepancy: **12.89 CMP GIB 801535 Engine Starter (Right) - Oil Change**

Resolution: **COMPLIED WITH CMP GIB 801535 Engine Starter (Right) - Oil Change IAW G2-B AMM 80-01-00 with no defects noted.**

	Billing Method	Hours	Rate	Total
Labor USD:	No Charge			N/ C
Parts USD:	Time and Material			\$115.28
Total USD:				\$115.28

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
12.89 #1	S9809-6	GASKET "C"	1	\$3.71	\$3.71
12.89 #2	S9809-3	GASKET S8995-3	2	\$8.84	\$17.68
12.89 #5	BE20057	SEAL	2	\$39.94	\$79.88
12.89 #4	KB21614	SEAL	1	\$14.01	\$14.01

Step: 12.89.1 Completed COMPLIED WITH CMP GIB 801535 Engine Starter (Right) - Oil Change steps A. to B(7). Awaiting for filler plug gasket.

Discrepancy: **12.90 CMP GIB 997407 AD74-08-09 R2 MOD LAVATORIES**

Resolution: **COMPLIED WITH CMP GIB 997407 AD74-08-09 R2, MOD LAVATORIES, IAW AD 74-08-09 R2. No defects noted. Next due 11804.3 Airframe hours.**

Discrepancy: **12.91 RELATED CMP CODES**

Resolution: **Complied with related codes.**

Discrepancy: **12.92 CMP GIB 243510 Alternator, Left Engine**

Resolution: **COMPLIED WITH CMP GIB 243510 L Engine Alternator - Removal / Installation P/N 28B135-126A, S/N 119, IAW GIB MM 24-02-04. No defects noted at this time.**

Labor USD: **No Charge** N/ C

For detail description of the work complied with see the Work Order Detail report.

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Step: 12.92.1 **Completed** Removed Left engine alternator (P/N: 28B135-126A ; S/N: 119) IAW GIB MM 24-02-04. No defects noted at this time.

Step: 12.92.2 **Completed** Installed Left engine alternator (P/N: 28B135-126A ; S/N: 119) IAW GIB MM 24-02-04. No defects noted at this time.

Discrepancy: **12.93** **CMP GIB 244010 Alternator, Right Engine**

Resolution: **COMPLIED WITH CMP GIB 244010 R Engine Alternator - Removal / Installation P/N 28B135-126A, S/N 369, IAW GIB MM 24-02-04. No defects noted at this time.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.94** **CMP GIB 252031 VEST, LIFE NO 1**

Resolution: **COMPLIED WITH CMP GIB 252031 VEST, LIFE NO 1. Removed life vest #1 P/N PO1074-101 S/N 263978 and installed #1 life vest P/N P01074-101 S/N263978, IAW GIB MM chapter 25-00-00. Next inspection due 9/2013.**

Step: 12.94.1 **Completed** Complied with section (A) removal and inspection of #1 life vest IAW G-IIB AMM 25-20-31. Next inspection due date is SEPT. 2013.

Discrepancy: **12.95** **CMP GIB 252032 VEST, LIFE NO 2**

Resolution: **COMPLIED WITH CMP GIB 252031 VEST, LIFE NO 2. Removed life vest #2 P/N PO1074-101 S/N 264254 and installed overhauled life vest P/N S-21850-6300 S/N 82901-1141, IAW GIB MM chapter 25-00-00. Next inspection due 2/2015.**

Discrepancy: **12.96** **CMP GIB 252033 VEST, LIFE NO 3**

Resolution: **COMPLIED WITH CMP GIB 252033 VEST, LIFE NO 3. Removed life vest #3 P/N PO1074-101 S/N 263936 and installed overhauled life vest P/N 51850-6300 S/N 21987-1660, IAW GIB MM chapter 25-00-00. Next inspection due 2/2015.**

Discrepancy: **12.97** **CMP GIB 252034 VEST, LIFE NO 4**

Resolution: **COMPLIED WITH CMP GIB 252034 VEST, LIFE NO 4. Removed life vest #4 P/N PO1074-101 S/N 263871 and installed overhauled life vest P/N S-51850-6300 S/N 20071-1660, IAW GIB MM chapter 25-00-00. Next inspection due 2/2015.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **12.98** CMP GIB 252035 VEST, LIFE NO 5

Resolution: **COMPLIED WITH CMP GIB 252035 VEST, LIFE NO 5. Removed life vest #5 P/N PO1074-101 S/N 264143 and installed overhauled life vest P/N S-51850-6300 S/N 21948-1660, IAW GIB MM chapter 25-00-00. Next inspection due 2/2015.**

Discrepancy: **12.99** CMP GIB 252036 VEST, LIFE NO 6

Resolution: **COMPLIED WITH CMP GIB 252036 VEST, LIFE NO 6. Removed life vest #6 P/N PO1074-101 S/N 264059 and installed #6 life vest P/N P01074-101 S/N 264059, IAW GIB MM chapter 25-00-00. Next inspection due 9/2013.**

Step: 12.99.1 **Completed** Complied with section (A) removal and inspection of #6 life vest IAW G-IIB AMM 25-20-31. Next inspection due date is SEPT. 2013.

Discrepancy: **12.100** CMP GIB 252037 VEST, LIFE NO 7

Resolution: **COMPLIED WITH CMP GIB 252037 VEST, LIFE NO 7. Removed life vest #7 P/N PO723E105P S/N 57151 and installed overhauled life vest P/N P01074-101 S/N 264256., IAW GIB MM chapter 25-00-00. Next inspection due 2/2017.**

Discrepancy: **12.101** CMP GIB 252038 VEST, LIFE NO 8

Resolution: **COMPLIED WITH CMP GIB 252038 VEST, LIFE NO 8. Removed life vest #8 P/N PO1074-101 S/N 263943 and installed #8 life vest P/N P01074-101 S/N 263943, IAW GIB MM chapter 25-00-00. Next inspection due 9/2013.**

Step: 12.101.1 **Completed** Complied with section (A) removal and inspection of #8 life vest P/N P01074-101 S/N 263943 IAW G-IIB AMM 25-20-31. Next inspection due date is SEPT. 2013.

Discrepancy: **12.102** CMP GIB 252039 VEST, LIFE NO 9

Resolution: **COMPLIED WITH CMP GIB 252039 VEST, LIFE NO 9. Removed life vest #9 P/N PO1074-101 S/N 264108 and installed #9 life vest P/N P01074-101 S/N 264108, IAW GIB MM chapter 25-00-00. Next inspection due 9/2013.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Step: 12.102.1 **Completed** Complied with section (A) removal and inspection of #9 life vest P/N P01074-101 S/N 264108 IAW G-IIB AMM 25-20-31. Next inspection due date is SEPT. 2013.

Discrepancy: **12.103** CMP GIB 252040 VEST, LIFE NO 10

Resolution: **COMPLIED WITH CMP GIB 252040 VEST, LIFE NO 10. Removed life vest #10 P/N PO1074-101 S/N 264206 and installed overhauled life vest P/N P01074-101 S/N 264206, IAW GIB MM chapter 25-00-00.. Next inspection due 2/2017.**

Step: 12.103.1 **Completed** Complied with section (A) removal and inspection of #10 life vest P/N P01074-101 S/N 264206 IAW G-IIB AMM 25-20-31. Next inspection due date is SEPT. 2013.

Discrepancy: **12.104** CMP GIB 252041 VEST, LIFE NO 11

Resolution: **COMPLIED WITH CMP GIB 252041 VEST, LIFE NO 11. Removed life vest #11 P/N PO1074-101 S/N 264204 and installed #11 life vest P/N P01074-101 S/N 264204, IAW GIB MM chapter 25-00-00. Next inspection due 9/2013.**

Step: 12.104.1 **Completed** Complied with section (A) removal and inspection of #11 life vest P/N P01074-101 S/N 264204 IAW G-IIB AMM 25-20-31. Next inspection due date is SEPT. 2013.

Discrepancy: **12.105** CMP GIB 252042 VEST, LIFE NO 12

Resolution: **COMPLIED WITH CMP GIB 252042 VEST, LIFE NO 12. Removed life vest #12 P/N PO1074-101 S/N 263941 and installed #12 life vest P/N P01074-101 S/N 263941, IAW GIB MM chapter 25-00-00. Next inspection due 9/2013.**

Step: 12.105.1 **Completed** Complied with section (A) removal and inspection of #12 life vest P/N P01074-101 S/N 263941 IAW G-IIB AMM 25-20-31. Next inspection due date is SEPT. 2013.

Discrepancy: **12.106** CMP GIB 252043 VEST, LIFE NO 13

Resolution: **COMPLIED WITH CMP GIB 252043 VEST, LIFE NO 13. Removed life vest #13 P/N 3505-101 S/N Q0503655 and installed overhauled life vest P/N P01074-101 S/N 689973, IAW GIB MM chapter 25-00-00. Next inspection due 2/2017.**

Discrepancy: **12.107** CMP GIB 252044 VEST, LIFE NO 14

Resolution: **COMPLIED WITH CMP GIB 252044 VEST, LIFE NO 14. Removed life vest #14 P/N PO1074-101 S/N 264247 and installed overhauled life vest P/N P01074-101 S/N 264247, IAW GIB MM chapter 25-00-00. Next inspection due 2/2017.**

Step: 12.107.1 **Completed** Complied with section (A) removal and inspection of #14 life vest P/N P01074-101 S/N 264247 IAW G-IIB AMM 25-20-31. Next inspection due date is SEPT. 2013.

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **12.108 CMP GIB 252056 LIFE VEST, JUMPSEAT**Resolution: **COMPLIED WITH CMP GIB 252056 LIFE VEST, JUMPSEAT. Removed jump seat life vest P/N 3505-101 S/N Q0704477 and installed overhauled vest P/N 3505-101 S/N Q0704477, IAW GIB MM chapter 25-00-00. Next inspection due 2/2014.**

Step: 12.108.1	Completed	Complied with section (A) removal and inspection of Jumpseat life vest P/N 3505-101 S/N Q0704477 IAW G-IIB AMM 25-20-31. Next inspection due date is
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Discrepancy: **12.109 CMP GIB 252057 LIFE VEST, PILOT**Resolution: **COMPLIED WITH CMP GIB 252057 LIFE VEST, PILOT. Removed pilot's life vest P/N S51850-6300 S/N 21965-1660 and installed overhauled life vest P/N S51850-6300 S/N 21965-1660, IAW GIB MM chapter 25-00-00. Next inspection due 2/2015.**

Step: 12.109.1	Completed	Complied with section (A) removal and inspection of pilot's life vest P/N S51850-6300 S/N 21965-1660 IAW G-IIB AMM 25-20-31. Next inspection due date is
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Discrepancy: **12.110 CMP GIB 252058 LIFE VEST, CO-PILOT**Resolution: **COMPLIED WITH CMP GIB 252058 LIFE VEST, CO-PILOT. Removed Co-pilots life vest P/N PO723W105P S/N 74634 and installed Co-pilots life vest P/N P0723E105P S/N 74634, IAW GIB MM chapter 25-00-00. Next inspection due 9/2013.**

Step: 12.110.1	Completed	Complied with section (A) removal and inspection of Co-Pilot's life vest P/N P0723W105P S/N74634 IAW G-IIB AMM 25-20-31. Next inspection due date is
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Discrepancy: **12.111 CMP GIB 266626 HALON FIRE EXT, COCKPIT**Resolution: **COMPLIED WITH CMP GIB 266626 HALON FIRE EXT, COCKPIT. Removed Fire Extinguisher P/N 6120-0000, S/N W575565 and installed Fire Extinguisher P/N 6120-0000, S/N W575565. All work done per G-IIB amm ch 26-03-00. No defects noted.**Labor USD: **No Charge**

N/ C

Discrepancy: **12.112 CMP GIB 266627 HALON FIRE EXT, FWD CABIN**Resolution: **COMPLIED WITH CMP GIB 266627 HALON FIRE EXT, FWD CABIN. Removed Fire Extinguisher P/N C352, S/N X-983309 and installed Fire Extinguisher P/N C352, S/N X-983309. All work done per G-IIB amm ch 26-03-00. No defects noted**Labor USD: **No Charge**

N/ C

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **12.113 CMP GIB 266629 BAGG COMP HALON FIRE BTL**

Resolution: **COMPLIED WITH CMP GIB 266629 BAGG COMP HALON FIRE BTL. Removed Fire extinguisher P/N 898052, S/N 575549 and installed Fire extinguisher P/N 898052, S/N 575549. All work done per G-IIB amm ch 26-03-00. No defects noted.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.114 CMP GIB 270115 Lateral Control System (Power) - Operational Test**

Resolution: **COMPLIED WITH CMP GIB 270115 Lateral Control System (Power) - Operational Test IAW GIB AMM 27-02-00. No defects noted.**

Discrepancy: **12.115 CMP GIB 272010 Aileron (Left)**

Resolution: **COMPLIED WITH CMP GIB 272010 Aileron (Left). Removed L/H Aileron P/N 1159CS20006-5, S/N 151. Installed the LH aileron P/N 1159CS20006-5, S/N 151 IAW GIB MM 27-02-00.**

	Billing Method	Hours	Rate	Total
Labor USD:	No Charge			N/ C
Parts USD:	Time and Material			\$6.19
Total USD:				\$6.19

Parts:

Ref. No.	Part number:	Description	Quantity	Sell price USD	Extended USD
12.115 #1	NAS6605D50	BOLT HEX HD-CLOSE TOL,DR SHANK	1	\$6.19	\$6.19

Step: 12.115.1 Completed Complied with CMP GIB 272010 L/H Aileron Removal & Installation Step C to D. Removed L/H Aileron Pn#1159CS20006-5 Sn#151 All work done IAW GIB MM 27-2-00.

Step: 12.115.2 Completed Installed the LH aileron P/N 1159CS20006-5, S/N 151 IAW GIB MM 27-02-00.

Discrepancy: **12.116 CMP GIB 272015 Aileron (Right)**

Resolution: **COMPLIED WITH CMP GIB 272015 Aileron (Right). Removed the RH aileron P/N 1159CS20006-6, S/N 151. Installed the same RH aileron P/N 1159CS20006-6, S/N 151 IAW GIB MM 27-02-00. Ref contingency sheet.**

Labor USD: **No Charge**

N/ C

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Step: 12.116.1 **Completed** Removed upper and lower right wing aileron removal access panels IAW GIBB MM 05-04-01. No defects noted at this time.
Step: 12.116.2 **Completed** Complied with CMP 272015 steps C. [1] thru [9]. Removed RH aileron P/N 1159CS20006-6 S/N 151. All work done IAW GIBB MM 27-02-00
Step: 12.116.3 **Completed** Installed the RH aileron P/N 1159CS20006-6, S/N 151 IAW GIBB MM 27-02-00. Ref contingency sheet.

Discrepancy: **12.117 CMP GIBB 272020 Aileron Trim Tab**

Resolution: **Complied With CMP GIBB 272020 Aileron Trim Tab - Removal / Installation. All Maintenance Performed In Reference To GIBB AMM 27-04-00. No Defects Noted At This Time.**

Step: 12.117.1 **Completed** Removed aileron trim tab.

Discrepancy: **12.118 CMP GIBB 270410 ELEV FREE PLAY INSP**

Resolution: **COMPLIED WITH CMP GIBB 270410 ELEV FREE PLAY INSP. Found to be within limits and no other defects noted. All work done IAW GIBB MM 27-01-00.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.119 CMP GIBB 270510 Elevator Control System - Operational Test**

Resolution: **COMPLIED WITH CMP GIBB 270510 Elevator Control System - Operational Test. All Maintenance Performed In Reference To GIBB AMM 27-01-00.**

Discrepancy: **12.120 CMP GIBB 272310 Elevator (Left)**

Resolution: **COMPLIED WITH CMP GIBB 272310 Elevator (Left) IAW GIBB AMM 27-01-00. No defect noted.**

Labor USD: **No Charge**

N/ C

Services USD: **No Charge**

N/ C

Services:

Ref. No.	Description	Quantity
12.120 #1	Vendor Supplied Labor	1.00

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Step: 12.120.1	Completed	COMPLIED WITH Elevator (Right) removal preparation IAW CMP GIB 272310 steps A. to B.(g). No defects noted.
Step: 12.120.2	Completed	COMPLIED WITH Elevator (LEFT) removal preparation by removing several access panels IAW CMP GIB 272310 No defects noted.
Step: 12.120.3	Completed	Complied with CMP GIB 272310 Elevator (Left), step C. (1) through (18). All maintenance performed IAW GIB MM 27-01-00.
Step: 12.120.4	Completed	Complied with CMP GIB 272310 Elevator (Left), step D. (1) through (k) and D (m) to (o). All maintenance performed IAW GIB MM 27-01-00.
Step: 12.120.5	Completed	weekend nights

Discrepancy: **12.121 CMP GIB 271810 ACTUATOR, L ELEVATOR TRIM TAB**

Resolution: **COMPLIED WITH CMP GIB 271810 ACTUATOR, L ELEVATOR TRIM TAB** Removed trim tab actuator P/N 159SCC100-21 S/N 56. Installed trim tab acutor P/N 159SCC100-21 S/N 56 IAW GIB AMM 27-04-00. Ops checked good.

Discrepancy: **12.122 CMP GIB 272315 Elevator (Right)**

Resolution: **COMPLIED WITH CMP GIB 272315 Elevator (Right) IAW GIB AMM 27-01-00. Ops checked good.**

Labor USD:	No Charge	N/ C
Parts USD:	No Charge	N/ C
Services USD:	No Charge	N/ C

Services:

<u>Ref. No.</u>	<u>Description</u>	<u>Quantity</u>
12.122 #1	Vendor Supplied Labor	1.00

Step: 12.122.1	Completed	COMPLIED WITH Elevator (Right) removal preparation IAW CMP GIB 272315 steps A. to B.(g). No defects noted.
Step: 12.122.2	Completed	Complied with CMP GIB 272315 Elevator (Right), step C. (1) through (18). All maintenance performed IAW GIB MM 27-01-00.
Step: 12.122.3	Completed	Complied with CMP GIB 272315 Elevator (Right), step D. (1) through (k) and D (m) to (o). All maintenance performed IAW GIB MM 27-01-00.
Step: 12.122.4	Completed	Complied with steps D.(3),(4),(5) and (6) of CMP 272315 by reconnecting and safetying the previously removed Right Elevator Trim Tab Actuator. All work was done IAW GIB AMM Ch.27-01-00.

Discrepancy: **12.123 CMP GIB 271815 ACTUATOR, R ELEVATOR TRIM TAB**

Resolution: **COMPLIED WITH CMP GIB 271815 ACTUATOR, R ELEVATOR TRIM TAB- Removal / Installation. All Maintenance Performed In Reference To GIB AMM 27-04-00. Ops checked good.**

Step: 12.123.1	Completed	Complied with Installation step D.(5) of CMP 271815 by reconnecting the previously disconnected Elevator Trim Tab Actuator. All work was done IAW GIB AMM Ch. 27-04-00.
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Discrepancy: **12.124 CMP GIB 272510 Elevator Trim Tab (Left)**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **COMPLIED WITH CMP GIB 272510 Elevator Trim Tab (Left)**

Removed trim tab P/N 1159CSH20500-1 S/N 147, with 10,804.3 TSN, 5,663 CSN. Installed "serviceable" as removed from G-II S/N 055 trim tab P/N 1159CSH20500-1 S/N 57, with 18,947.2 TSN 7,651 CSN. All work performed IAW GIB AMM 27-04-00. Ref squawk 8.128

Labor USD: **No Charge**

N/ C

Step: 12.124.1 Completed Complied with CMP GIB 272510 Elevator Trim Tab (Left) steps C (1) thru C(4) IAW GIB AMM 27-04-00.

Step: 12.124.2 Completed Continued installation of the L/H elevator trim tab. Complied with CMP 272510 C(5) through D(11). Work done I/A/W GIB MM chapter 27-04-00.

Discrepancy: **12.125 CMP GIB 272515 Elevator Trim Tab (Right)**

Resolution: COMPLIED WITH CMP GIB 272515 Elevator Trim Tab (Right)-Removal / Installation. All Maintenance Performed In Reference To GIB AMM 27-04-00.

Labor USD: **No Charge**

N/ C

Step: 12.125.1 Completed Complied with CMP GIB 272515 Elevator Trim Tab (RT) steps C (1) thru C(4) IAW GIB AMM 27-04-00.

Step: 12.125.2 Completed Complied with CMP 272515 Elevator trim tab (Right) - removal / installation steps D (2) through (6) installed right elevator trim tab bolts, all work in accordance with GIB amm chapter 27-04-00.

Discrepancy: **12.126 CMP GIB 270310 Rudder Control System (SP-50G Aircraft) - Operational Test**

Resolution: COMPLIED WITH CMP GIB 270310 Rudder Control System (SP-50G Aircraft) - Operational Test. All Maintenance Performed In Reference To GIB AMM 27-03-00.

Discrepancy: **12.127 CMP GIB 272810 Rudder (Aluminum) - Removal / Installation**

Resolution: COMPLIED WITH CMP GIB 272810 Rudder (Aluminum) - Removal / Installation IAW GIB AMM 27-00-00. See steps for details.

Labor USD: **No Charge**

N/ C

For detail description of the work complied with see the Work Order Detail report.

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Step: 12.127.1	Completed	Complied with CMP 272810 Rudder Removal/ Installation. Step B. [2] [a] thru [d] and C. [1] Removed required access panels and bonding straps. Removed hinge bolt safties. Applied pretrant to all hinges. Rudder is ready to remove. All work done IAW GIBB MM 27-03-00
Step: 12.127.2	Completed	Complied with CMP GIBB 272810 Rudder (Aluminum) - Removal / Installation, step C. (2) through (5). All maintenance performed IAW GIBB MM 27-03-00. Right throw (22), Left throw (23).
Step: 12.127.3	Completed	Complied with CMP GIBB 272810 Rudder (Aluminum) - Removal / Installation, step D. (1) through (8) and D (11) to (13). All maintenance performed IAW GIBB MM 27-03-00.
Step: 12.127.4	Completed	Complied with step D (9),(10), (14) thru (16) of CMP 272810 Rudder (Aluminum - removal / installation.

Discrepancy: 12.128 CMP GIBB 270121 Speed Brake - Operational Test (A/C Having 6 Panel Speed Brakes)

Resolution: COMPLIED WITH CMP GIBB 270121 Speed Brake - Operational Test (A/C Having 6 Panel Speed Brakes)IAW GIBB AMM 27-02-00. No defects noted.

Discrepancy: 12.129 CMP GIBB 275515 Flight Spoiler, Inboard (Left)

Resolution: COMPLIED WITH CMP GIBB 275515 Flight Spoiler, Inboard (Left). Removed the LH inboard flight spoiler P/N 1159SCH20700-5, S/N AV119MAR02, and Installed the same LH inboard flight spoiler P/N 1159SCH20700-5, S/N AV119MAR02. Work performed IAW GIBB MM 27-02-00. Ref contingency sheet.

Labor USD: No Charge

N/ C

Step: 12.129.1	Completed	Complied with CMP GIBB 275515 L/H Inb Flight Spoiler Removal & Installation Step C to D. Removed L/H Inb Flight Spoiler Pn# 1159SCH20700-5 Sn# AV119MAR02 All work done IAW GIBB MM 27--02-00
Step: 12.129.2	Completed	Cleaned and greased all hardware and attach fitting holes. Installed the LH inboard flight spoiler P/N 1159SCH20700-5, S/N AV119MAR02 using new hardware and slip bushings from 72 month install kit. Work performed IAW GIBB MM 27-02-00. Ref contingency sheet.

Discrepancy: 12.130 CMP GIBB 275510 Flight Spoiler, Outboard (Left)

Resolution: COMPLIED WITH CMP GIBB 275510 Flight Spoiler, Outboard (Left). Removed the LH outboard flight spoiler P/N 1159GSH20702-7, S/N AV-154 and Installed the same LH outboard flight spoiler P/N 1159GSH20702-7, S/N AV-154 Work performed IAW GIBB MM 27-02-00. Ref contingency sheet.

Labor USD: No Charge

N/ C

Step: 12.130.1	Completed	Complied with CMP GIBB 275510 L/H Out Flight Spoiler Removal & Installation Step C to D. Removed L/H Out Flight Spoiler Pn# 1159GSH20702-7 Sn# AV154 All work done IAW GIBB MM 27--02-00
Step: 12.130.2	Completed	Cleaned and greased all hardware and attach fitting holes. Installed the LH outboard flight spoiler P/N 1159GSH20702-7, S/N AV-154 using new hardware and slip bushings from 72 month install kit. Work performed IAW GIBB MM 27-02-00. Ref contingency sheet.

Discrepancy: 12.131 CMP GIBB 275525 Flight Spoiler, Inboard (Right)

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **COMPLIED WITH CMP GIB 275525 Flight Spoiler, Inboard (Right). Removed the RH inboard flight spoiler P/N 1159SCH20700-6, S/N AV-155 and Installed the same RH inboard flight spoiler P/N 1159SCH20700-6, S/N AV-155. Work performed IAW GIB MM 27-02-00. Ref contingency sheet.**

Labor USD: No Charge

N/ C

Step: 12.131.1 Completed Complied with CMP 275525 Steps B.[1] thru C.[5]. Removed RH Inboard Flight Spoiler P/N1159SCH20700-6 S/N AV-155. All work done IAW GIB MM 27-02-00.
Step: 12.131.2 Completed Cleaned and greased all hardware and attach fitting holes. Installed the RH inboard flight spoiler P/N 1159SCH20700-6, S/N AV-155 using new hardware and slip bushings from 72 month install kit. Work performed IAW GIB MM 27-02-00. Ref contingency sheet.

Discrepancy: **12.132 CMP GIB 275520 Flight Spoiler, Outboard (Right)**

Resolution: **COMPLIED WITH CMP GIB 275520 Flight Spoiler, Outboard (Right). Removed the RH outboard flight spoiler P/N 1159SCH20702-8, S/N AV-148, and installed the same RH outboard flight spoiler P/N 1159SCH20702-8, S/N AV-148 IAW GIB MM 27-02-00.**

Labor USD: No Charge

N/ C

Step: 12.132.1 Completed Complied with CMP 275520 Steps B.[1] thru C.[5]. Removed RH Outboard Flight Spoiler P/N1159SCH20702-8 S/N AV-148. All work done IAW GIB MM 27-02-00.
Step: 12.132.2 Completed Installed the RH outboard flight spoiler P/N 1159SCH20702-8, S/N AV-148 IAW GIB MM 27-02-00. Ref contingency sheet.

Discrepancy: **12.133 CMP GIB 270011 6 PNL SPEEDBRAKE RIG CK (GIB)**

Resolution: **COMPLIED WITH CMP GIB 270011 6 PNL SPEEDBRAKE RIG CK (GIB) IAW GIB AMM 27-02-00. No defects noted.**

Discrepancy: **12.134 CMP GIB 275011 SPLR,GRND L (6PNL) S/BRK(GIB)**

Resolution: **COMPLIED WITH CMP GIB 275011 SPLR,GRND L (6PNL) S/BRK(GIB). Removed the LH ground spoiler P/N 1159CSH30000-5, S/N AV-414, and installed the same LH ground spoiler P/N 1159CSH30000-5, S/N AV-414 IAW GIB MM 27-02-00.**

Labor USD: No Charge

N/ C

Step: 12.134.1 Completed Complied with CMP GIB 275011 L/H Ground Spoiler Removal & Installation Step C to D. Removed L/H Ground Spoiler Pn#1159CSH30000-5 Sn#AV414 All work done IAW GIB MM 27--02-00
Step: 12.134.2 Completed Installed the LH ground spoiler P/N 1159CSH30000-5, S/N AV-414 IAW GIB MM 27-02-00. Ops checks still required. Ref contingency sheet.

Discrepancy: **12.135 CMP GIB 270020 CK GRD SPOILER RIGGING**

For detail description of the work complied with see the Work Order Detail report.

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Resolution: **COMPLIED WITH CMP GIB 270020 CK GRD SPOILER RIGGING ,IAW GIB AMM 27-02-01. No defects noted.**

Discrepancy: **12.136 CMP GIB 275166 L.GRD SPOILER IND SW ADJ**

Resolution: **COMPLIED WITH CMP GIB 275166 L.GRD SPOILER IND SW ADJ IAW GIB AMM 27-02-01. No defects noted.**

Discrepancy: **12.137 CMP GIB 275016 SPLR,GRND R (6PNL) S/BRK(GIB)**

Resolution: **COMPLIED WITH CMP GIB 275016 SPLR,GRND R (6PNL) S/BRK(GIB). Removed RH ground spoiler P/N 1159CSH30000-5, S/N AV-411, and installed the same RH ground spoiler P/N 1159CSH30000-5, S/N AV-411 IAW GIB MM 27-02-00.**

Labor USD: **No Charge**

N/ C

Step: 12.137.1 Completed Complied with CMP 275016 Steps B.[1] thru C.[6]. Removed RH Ground Spoiler P/N1159SCH30000-5 S/N AV-411. All work done IAW GIB MM 27-02-00.

Step: 12.137.2 Completed Installed RH ground spoiler P/N 1159CSH30000-5, S/N AV-411 IAW GIB MM 27-02-00. Ops checks still required, ref item # 135 for ops checks. (Ref contingency sheet.)

Discrepancy: **12.138 CMP GIB 275171 R.GRD SPOILER IND SW ADJ**

Resolution: **COMPLIED WITH CMP GIB 275171 R.GRD SPOILER IND SW ADJ IAW GIB AMM 27-02-01. No defects noted.**

Discrepancy: **12.139 CMP GIB 274510 Flap (Left) - Removal / Installation**

Resolution: **Complied with CMP 274510 Left Flap Removal and Installation. P/n off 1159CS20005-1 S/n 151 L/H P/N on 1159CS20005-1 S/n on 151 L/H Work done I/A/W GIB MM chapter 27-06-00.**

Labor USD: **No Charge**

N/ C

For detail description of the work complied with see the Work Order Detail report.

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Step: 12.139.1	Completed	Complied with CMP 274510 steps A through C(1), IAW GIBB M/M Chapter 27-06-00.
Step: 12.139.2	Completed	Complied with CMP GIBB 274510 Flap (Left) - Removal / Installation, step C.(2) through C.(7). All maintenance performed IAW GIBB MM 27-06-00.
Step: 12.139.3	Completed	Complied with CMP 274510 steps D (1) thru (7) installed previously removed flap p/n 1159CS20005-1 s/n 151 and saftied. Work done IAW G-IIB m/m 27-06-00. (flap still requires symetry and ops checks). ref contingency sheet.
Step: 12.139.4	Completed	Complied with CMP 274510 items D(8) to the end. Work done I/A/W GIBB MM chapter 27-06-00.

Discrepancy: 12.140 CMP GIBB 274710 Flap (Right) - Removal / Installation

Resolution: Complied with CMP 274710 Right Flap Removal and Installation. P/n off 1159CS20005-2 S/n 151 R/H P/N on 1159CS20005-2 S/n on 151 R/H Work done I/A/W GIBB MM chapter 27-06-00.

Labor USD: No Charge N/ C

Step: 12.140.1	Completed	Complied with CMP 274710 steps A through C(1), IAW GIBB M/M Chapter 27-06-00.
Step: 12.140.2	Completed	Complied with CMP GIBB 274710 Flap (Right) - Removal / Installation, step C.(2) through C.(7). All maintenance performed IAW GIBB MM 27-06-00.
Step: 12.140.3	Completed	Complied with CMP 274510 steps D (1) thru (7) installed previously removed flap p/n 1159CS20005-2 s/n 151 and saftied. Work done IAW G-IIB m/m 27-06-00. (flap still requires symetry and ops checks). ref contingency sheet.
Step: 12.140.4	Completed	Complied with CMP 274710 items D(8) to the end. Work done I/A/W GIBB MM chapter 27-06-00.

Discrepancy: 12.141 CMP GIBB 271010 ACTUATOR, LEFTAILERON

Resolution: COMPLIED WITH CMP GIBB 271010 ACTUATOR, LEFTAILERON. Removed the LH aileron actuator P/N 1159H20029-11D3, S/N 468 and installed the LH aileron actuator P/N 1159H20029-11D3, S/N 468 IAW GIBB MM 27-02-00.

Labor USD: No Charge N/ C

Step: 12.141.1	Completed	Complied with L/H aileron actuator removal ,steps(C) 1-8 IAW G-IIB AMM27-02-00, P/N 1159H20029-11D3 S/N 468. Ready to be sent out for overhaul.
Step: 12.141.2	Completed	Installed LH aileron actuator P/N 1159H20029-11D3, S/N 468 IAW GIBB MM 27-02-00. Ref contingency sheet.

Discrepancy: 12.142 CMP GIBB 271015 ACTUATOR, RIGHTAILERON

Resolution: COMPLIED WITH CMP GIBB 271015 ACTUATOR, RIGHTAILERON. Removed and installed R/H aileron actuator PN. 1159SCH2522E SN. 224GT as required IAW GIBB MM 27-02-00.

Labor USD: No Charge N/ C

Step: 12.142.1	Completed	Complied with R/H aileron actuator removal ,steps(C) 6-8 IAW G-IIB AMM27-02-00, P/N 1159SCH2522E S/N 224GT.
Step: 12.142.2	Completed	Installed R/H aileron actuator PN. 1159SCH2522E SN. 224GT as required IAW GIBB CMP 271010 D4 -D 10 and GIBB MM 27-02-00. Requires operational and leak checks.

Discrepancy: 12.143 CMP GIBB 271315 ACTUATOR, RUDDER TRIM

Resolution: COMPLIED WITH CMP GIBB 271315 ACTUATOR, RUDDER TRIM IAW GIBB AMM 27-04-00. No defected noted.

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Labor USD: No Charge

N/ C

Step: 12.143.1 **Completed** Complied with CMP 271315 steps C4, D2 and C7, to comply with rudder trim tab actuator inspection (see item 12.40). Rudder trim tab actuator cables need to be tensioned. Cables are flagged. All work performed I.A.W. GIB M.M. 27-04-00. Ops checked NO GOOD. Sticks in full left and full right travels. See Squawk 8.90

Discrepancy: **12.144** CMP GIB 271510 ACTUATOR, AILERON TRIM TAB

Resolution: **COMPLIED WITH CMP GIB 271510 ACTUATOR, AILERON TRIM TAB. Removed the aileron trim tab actuator P/N 1159SCC201-3, S/N 279 and installed the same aileron trim tab actuator P/N 1159SCC201-3, S/N 279. Work performed IAW GIB MM 27-04-00.**

Discrepancy: **12.145** CMP GIB 270316 Wing Flap / Stabilizer - Operational Test (GIB)

Resolution: **Complied with CMP 270316 Wing Flap Stabilizer Operational Test. Work done I/A/W GIB MM chapter 27-06-01.**

Discrepancy: **12.146** CMP GIB 274811 ACTUATOR,L GRND SPOILER (GIB)

Resolution: **Complied with CMP GIB 274811 ACTUATOR,L GRND SPOILER (GIB) removal and installation as required IAW GIB MM 27-02-00.Removed PN. 1159SCH305-11 SN. 0105 and installed same unit PN. 1159SCH305-11 SN. 0105. Ref contingency sheet.**

Labor USD: No Charge

N/ C

Step: 12.146.1 **Completed** **COMPLIED WITH** Removal of ACTUATOR, L/H GRND SPOILER P/N 1159SCH305-11 S/N 0105 IAW CMP GIB 274816 steps A. to C.(10).
Step: 12.146.2 **Completed** Installed same ground spoiler actuator as removed previously PN. 1159SCH305-11 SN. 0105 IAW GIB CMP 274811 and MM 27-02-00. Ground spoiler rigging check shall be accomplished under squawk 12.135 of this Work Order CMP 270020. Servicing and bleed of hydraulic reservoir to be complied with under squawk 12.179 ,CMP 290110.

Discrepancy: **12.147** CMP GIB 274816 ACTUATOR,R GRND SPOILER (GIB)

Resolution: **Complied with CMP GIB 274816 ACTUATOR,R GRND SPOILER (GIB) IAW GIB MM 27-02-00. Ref contingency sheet.**

Labor USD: No Charge

N/ C

For detail description of the work complied with see the Work Order Detail report.

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Step: 12.147.1 **Completed** **COMPLIED WITH** Removal of ACTUATOR,R GRND SPOILER P/N 1159SCH305-11 S/N 0130GT IAW CMP GIB 274816 steps A. to C.(10).
Step: 12.147.2 **Completed** **Installed** R/H ground spoiler actuator PN. 1159SCH305-11 SN. 0130GT IAW GIB CMP 274811 and GIB MM 27-02-00. Ground spoiler rigging check will be C/W under squawk 12.135 and reservoir service under squawk 12.179. Ref contingency sheet.

Discrepancy: **12.148** CMP GIB 275310 ACTUATOR, L FLT SPOILER

Resolution: **COMPLIED WITH** CMP GIB 275310 ACTUATOR, L FLT SPOILER. Removed LH flight spoiler actuator P/N 1159SCH250-3 S/N 362GT. Installed a Modified LH flight spoiler actuator P/N 1159SCH250-29, S/N 2002. Work performed IAW GIB MM 27-02-00. Ref contingency sheet.

Labor USD: **No Charge**

N/ C

Step: 12.148.1 **Completed** **COMPLIED WITH** Removal of ACTUATOR, L FLT SPOILER P/N 1159SCH250-3 S/N 362GT IAW CMP GIB 275310 steps C.(1) to C.(5).
Step: 12.148.2 **Completed** **Installed** a Modified LH flight spoiler actuator P/N 1159SCH250-29, S/N 2002. Work performed IAW GIB MM 27-02-00. Ref contingency sheet.

Discrepancy: **12.149** CMP GIB 275320 Outboard Flight Spoiler Bellcrank Assembly (Left) - Removal / Installation

Resolution: **COMPLIED WITH** CMP GIB 275320 Outboard Flight Spoiler Bellcrank Assembly (Left) - Removal / Installation. Removed the LH outboard flight spoiler bellcrank assembly P/N 1159CM20209-5, NSN. Installed a Repaired LH outboard flight spoiler bellcrank assembly P/N 1159CM20209-5, NSN IAW GIB MM 27-02-00.

Labor USD: **No Charge**

N/ C

Step: 12.149.1 **Completed** **Complied with** CMP GIB 275320 Outboard Flight Spoiler Bellcrank Assembly (Left) - Removal / Installation- Steps C.- (1) thrgh C.- (6) - removed (Left) Outboard Flight Spoiler Bellcrank Assembly P/N:1159CM20209-5 (recorded from CMP) S/N: n/a. work done IAW GIB AMM 27-02-00.
Step: 12.149.2 **Completed** **Installed** a Repaired LH outboard flight spoiler bellcrank assembly P/N 1159CM20209-5, NSN IAW GIB MM 27-02-00.

Discrepancy: **12.150** CMP GIB 275321 Inboard Flight Spoiler Bellcrank Assembly (Left) - Removal / Installation

Resolution: **COMPLIED WITH** CMP GIB 275321 Inboard Flight Spoiler Bellcrank Assembly (Left) - Removal / Installation. Removed the LH inboard flight spoiler bellcrank assembly P/N 1159SB30144-11, NSN. Installed a Repaired LH inboard flight spoiler bellcrank assembly P/N 1159SB30144-11, NSN IAW GIB MM 27-02-00.

Labor USD: **No Charge**

N/ C

Step: 12.150.1 **Completed** **Complied with** CMP GIB 275321 Inboard Flight Spoiler Bellcrank Assembly (Left) - Removal / Installation- Steps C.- (1) thrgh C.- (7) - Removed (Left) Inboard Flight Spoiler Bellcrank Assembly P/N:1159SB30144-11 (recorded from CMP) S/N: n/a. work done IAW GIB AMM 27-02-00.
Step: 12.150.2 **Completed** **Installed** a Repaired LH inboard flight spoiler bellcrank assembly P/N 1159SB30144-11, NSN IAW GIB MM 27-02-00. Ref contingency sheet.

Discrepancy: **12.151** CMP GIB 275315 ACTUATOR, R FLT SPOILER

For detail description of the work complied with see the Work Order Detail report.

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Resolution: **COMPLIED WITH CMP GIB 275315 ACTUATOR, R FLT SPOILER. Removed R/H flight spoiler actuator P/N 1159SCH250-4D, S/N 294GT. Installed a Modified RH flight spoiler actuator P/N 1159SCH250-30, S/N 1731. Work performed IAW GIB MM 27-02-00.**

Labor USD: No Charge

N/ C

Step: 12.151.1 Completed Removed R/H flight spoiler actuator. Work done I/A/W GIB MM chapter 27-02-00. Complied with CMP 275315 C(1) through (5).

Step: 12.151.2 Completed Installed a Modified RH flight spoiler actuator P/N 1159SCH250-30, S/N 1731. Work performed IAW GIB MM 27-02-00. Ref contingency sheet.

Discrepancy: **12.152 CMP GIB 275322 Inboard Flight Spoiler Bellcrank Assembly (Right) - Removal / Installation**

Resolution: **Installed the new R/H inbd flight spoiler bellcrank pushrod P/N 1159C20262-3 to the R/H Inbd flight spoiler bellcrank ref. GIB CMP 275322 and the GIB AMM 27-02-00. Removed P/N 1159C20262-3 S/N NSN. Installed the repaired R/H inbd flight spoiler bellcrank P/N 1159CM20208-2 NSN ref. GIB CMP 275322 and the GIB AMM 27-02-00. Removed P/N 1159CM20208-2 S/N NSN. Refer to items 12.51, 12.114 and 12.128 for final attachments and checks.**

Labor USD: No Charge

N/ C

Step: 12.152.1 Completed Complied with CMP GIB 275322 Inboard Flight Spoiler Bellcrank Assembly (Right) - Removal / Installation- Steps C.- (1) thrgh C.- (7) - Removed (Right) Inboard Flight Spoiler Bellcrank Assembly P/N:1159CM20208-2 (recorded from CMP) S/N: n/a. work done IAW GIB AMM 27-02-00.

Discrepancy: **12.153 CMP GIB 275323 Outboard Flight Spoiler Bellcrank Assembly (Right) - Removal / Installation**

Resolution: **Installed the repaired R/H outbd flight spoiler bellcrank P/N 1159CM20209-6 NSN ref. GIB CMP 275323 and the GIB AMM 27-02-00. Removed P/N 1159CM20209-6 S/N NSN. Refer to items 12.132, 12.131, 12.114 and 12.128 for final attachment and checks.**

Labor USD: No Charge

N/ C

Step: 12.153.1 Completed Complied with CMP GIB 275323 Outboard Flight Spoiler Bellcrank Assembly (Right) - Removal / Installation- Steps C.- (1) thrgh C.- (6) - removed (Right) Outboard Flight Spoiler Bellcrank Assembly P/N:1159CM20209-6 (recorded from CMP) S/N: n/a. work done IAW GIB AMM 27-02-00.

Discrepancy: **12.154 CMP GIB 351511 Oxygen Cylinder (No.2)**

Resolution: **COMPLIED WITH CMP GIB 351511 Oxygen Cylinder (No. 2) - Removal / Installation. Removed #2 oxygen bottle P/N 176521-64, S/N 881993 and sent out for Hydro inspection. Installed inspected O2 bottle P/N 176521-64, S/N 881993 IAW G IIB AMM 35-00-00.**

Labor USD: No Charge

N/ C

For detail description of the work complied with see the Work Order Detail report.

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Bill To: HYPERION AIR, INC.

WO #: SC10420

Invoice Number: 7112000954

- Step: 12.154.1** **Completed** Complied with section (A), (B) removed #2 oxygen bottle from aircraft P/N 176521-64, S/N 881993 and sent to vendor for new hydro IAW G IIB AMM 35-00-00.
- Step: 12.154.2** **Completed** Complied with section (C) (1) through (D)(6) reinstalled #2 oxygen bottle from aircraft P/N 176521-64, S/N 881993 IAW G IIB AMM 35-00-00.

Discrepancy: 12.155 CMP GIB 351010 Oxygen System (Pilot) - Operational Test

Resolution: **COMPLIED WITH CMP GIB 351010 Oxygen System (Pilot) - Operational Test IAW GII-B AMM 35-00-00. No defects noted.**

Discrepancy: 12.156 CMP GIB 351015 Oxygen System (Copilot) - Operational Test

Resolution: **COMPLIED WITH CMP GIB 351015 Oxygen System (Copilot) - Operational Test IAW GII-B AMM 35-00-00. No defects noted.**

Discrepancy: 12.157 CMP GIB 351016 Passenger Oxygen System - Operational Test (Gulfstream Installed System)

Resolution:

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Parts USD:	Time and Material			\$87.00
Total USD:				\$87.00

Parts:

<u>Ref. No.</u>	<u>Part number:</u>	<u>Description</u>	<u>Quantity</u>	<u>Sell price USD</u>	<u>Extended USD</u>
12.157 #1	MS22012-6	PACKING RING	10	\$8.70	\$87.00

Discrepancy: 12.158 CMP GIB 351512 Oxygen Cylinder (No.3)

Resolution: **COMPLIED WITH CMP GIB 351512 Oxygen Cylinder (No. 3) - Removal / Installation. Removed #3 oxygen bottle from aircraft P/N 176521-49, S/N 642653 and sent out for Hydro inspection. Installed inspected O2 bottle P/N 176521-49, S/N 642653 IAW G IIB AMM 35-00-00.**

Labor USD: **No Charge** N/ C

- Step: 12.158.1** **Completed** Complied with section (A), (B) removed #3 oxygen bottle from aircraft P/N 176521-49, S/N 642653 and sent to vendor for new hydro IAW G IIB AMM 35-00-00.
- Step: 12.158.2** **Completed** Complied with section (C) (1) through (D)(6) reinstalled #3 oxygen bottle from aircraft P/N 176521-49, S/N 642653 IAW G IIB AMM 35-00-00.

Discrepancy: 12.159 CMP GIB 351205 BOTTLE, NO 1 PORTABLE OXYGEN

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Resolution: **Complied with section (A), removed portable oxygen bottle from aircraft P/N 900-002-035-01, S/N 33925R and sent to vendor for new hydro IAW G IIB AMM 35-00-00. Received portable oxygen bottle from vendor P/N 900-002-035-01, S/N 33925R see item 12.58 for details; and replaced portable oxygen bottle in aircraft IAW G-II B AMM 35-00-00.**

Step: 12.159.1 Completed Complied with section (A), removed portable oxygen bottle from aircraft P/N 900-002-035-01, S/N 33925R and sent to vendor for new hydro IAW G IIB AMM 35-00-00.

Discrepancy: **12.160 CMP GIIB 242520 Inverter, E Instrument**

Resolution: **COMPLIED WITH CMP GIIB 242520 Inverter, E Instrument. Re-moved Inverter E. P/N PC-17A1 S/N 2794. Re-installed Inverter E. P/N PC-17A1 S/N 2794. All work was done IAW Gulfstream GIIB AMM 24-25-20. No defects noted.**

Discrepancy: **12.161 CMP GIIB 242710 Inverter, B Standby**

Resolution: **COMPLIED WITH CMP GIIB 242710 Inverter, B Standby. Removed Inverter B. P/N 501-1128-01 S/N 1160. Re-installed Inverter B. P/N 501-1128-01 S/N 1160. All work was done IAW Gulfstream GIIB AMM 24-27-10. No defects noted.**

Discrepancy: **12.162 CMP GIIB 242720 Rectifier, Inverter B Transformer**

Resolution: **COMPLIED WITH CMP GIIB 242720 Rectifier, Inverter B Transformer. Removed Inverter Rectifier Transformer B. P/N 28VS200Y-17 S/N310. Re-installed Inverter Rectifier Transformer B. P/N 28VS200Y-17 S/N310. All work was done IAW Gulfstream GIIB AMM 24-27-20. No defects noted.**

Discrepancy: **12.163 CMP GIIB 246015 B Inverter - Frequency / Voltage Check**

Resolution: **COMPLIED WITH CMP GIIB 246015 B Inverter - Frequency / Voltage Check with no discrepancies noted at this time. All work performed IAW G-II B maintenance manual 24-03-00.**

Labor USD: **No Charge**

N/ C

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **12.164 CMP GIB 246025 E Inverter - Frequency / Voltage Check**

Resolution: **COMPLIED WITH CMP GIB 246025 E Inverter - Frequency / Voltage Check with no discrepancies noted at this time. All work performed IAW G-IIB maintenance manual 24-03-00.**

Labor USD: **No Charge**

N/C

Discrepancy: **12.165 CMP GIB 280110 AIRCRAFT DEFUELING**

Resolution: **Defueled 6400 lbs. Drained fuel from sumps. All work done IAW 28-00-00.**

Labor USD: **No Charge**

N/C

Discrepancy: **12.166 CMP GIB 280210 BLEED FUEL SYSTEM**

Resolution: **COMPLIED WITH CMP GIB 280210 BLEED FUEL SYSTEM IAW GIB AMM 28-04-00. No defects noted.**

Step: 12.166.1 Completed Refueled aircraft with 6400 lbs of fuel. Work done I/A/W GIB MM chapter 28-04-00.

Discrepancy: **12.167 CMP GIB 284511 No. 1 Left Fuel Quantity Probe - Removal / Installation**

Resolution: **N/A Removal was not required.**

Discrepancy: **12.168 CMP GIB 284636 No. 13 Left Fuel Quantity Probe - Removal / Installation**

Resolution: **N/A Removal was not required.**

Discrepancy: **12.169 CMP GIB 281111 L FUEL PROBE INSUL/CAP CK(GIB**

Resolution: **No Action Required At This Time Due To Maintenance Performed, Fuel Probes Were Never Disconnected.**

For detail description of the work complied with see the Work Order Detail report.

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Discrepancy: **12.170 CMP GIB 289080 SCREEN ASSY,LM FUEL P/U (GIB)**

Resolution: **COMPLIED WITH CMP GIB 289080 SCREEN ASSY,LM FUEL P/U (GIB) Removed fuel pickup P/N 1159P20142-1 NSN. Installed same fuel pickup P/N 1159P20142-1 NSN IAW GIB AMM 28-00-00**

Labor USD: **No Charge**

N/ C

Step: 12.170.1 Completed Complied with CMP GIB 289080 SCREEN ASSY,LM FUEL P/U (GIB) Steps A(1) thru A(3).

Discrepancy: **12.171 CMP GIB 289085 SCREEN ASSY,LA FUEL P/U (GIB)**

Resolution: **COMPLIED WITH CMP GIB 289085 SCREEN ASSY,LA FUEL P/U (GIB) Removed fuel pickup P/N 1159P20142-3 NSN. Installed same fuel pickup P/N 1159P20142-3 NSN IAW GIB AMM 28-00-00.**

Labor USD: **No Charge**

N/ C

Step: 12.171.1 Completed Complied with CMP GIB 289085 SCREEN ASSY,LA FUEL P/U (GIB) Steps A(1) thru A(3).

Discrepancy: **12.172 CMP GIB 575048 Lower Wing Planks BL 0 to BL 19 (Left) - X-ray / NDT Inspection**

Resolution: **COMPLIED WITH CMP GIB 575048 Lower Wing Planks BL 0 to BL 19 (Left) - X-ray / NDT Inspection IAW GIB 57-00-00. Work performed by Structural testing Systems Cert. # ES1R124K. See item 8.122 for discrepancy noted.**

Labor USD: **No Charge**

N/ C

Step: 12.172.1 Completed Removed Fuel panels 21-FT LWR-1, 21-FT LWR-2, 21-FT LWR-3, 21-FT LWR-4, 22-FT LWR-1, 22FT LWR-2, 22-FT LWR-3, 22-FT LWR-4, 22-FT UPR-12, 22-FT UPR-13, 22-FT UPR-15, 22 FT UPR-17. Done I.A.W. GIB 57-00-00.

Discrepancy: **12.173 CMP GIB 285011 No. 1 Right Fuel Quantity Probe - Removal / Installation**

Resolution: **N/A Removal was not required.**

Discrepancy: **12.174 CMP GIB 285136 No. 13 Right Fuel Quantity Probe - Removal / Installation**

Resolution: **N/A Removal was not required.**

For detail description of the work complied with see the Work Order Detail report.

Unless otherwise indicated, all financial values listed in this document are in US Dollars (USD).

Discrepancy: **12.175 CMP GIB 281116 R FUEL PROBE INSUL/CAP CK(GIIB)**

Resolution: **No Action Required At This Time Due To Maintenance Performed, Fuel Probes Were Never Disconnected.**

Discrepancy: **12.176 CMP GIB 289090 SCREEN ASSY, RM FUEL P/U (GIIB)**

Resolution: **Complied With CMP GIB 289090 Screen Assembly, Removal / Installation. No Defects Noted At This Time. All Maintenance Performed In Reference To GIB AMM 28-00-00.**

Step: 12.176.1 Completed Complied with CMP GIB 289090 SCREEN ASSY, RM FUEL P/U (GIIB) Steps A(1) thru A(3).

Discrepancy: **12.177 CMP GIB 289095 SCREEN ASSY, RA FUEL P/U (GIIB)**

Resolution: **Complied With CMP GIB 289095 Screen Assembly. Removal / Installation. No Defects Noted At This Time. All Maintenance Performed In Reference To GIB AMM 28-00-00.**

Discrepancy: **12.178 CMP GIB 575053 Lower Wing Planks BL 0 to BL 19 (Right) - X-ray / NDT Inspection**

Resolution: **COMPLIED WITH CMP GIB 575053 Lower Wing Planks BL 0 to BL 19 (Right) - X-ray / NDT Inspection IAW GIB 57-00-00. Work performed by Structural Testing Systems Cert. # ES1R124K. See item 8.122 for discrepancy noted..**

	<u>Billing Method</u>	<u>Hours</u>	<u>Rate</u>	<u>Total</u>
Labor USD:	No Charge			N/ C
Services USD:	Time and Material			<u>\$14,545.27</u>
Total USD:				<u>\$14,545.27</u>

Services:		<u>Quantity</u>	<u>Sell Price USD</u>	<u>Extended USD</u>
<u>Ref. No.</u>	<u>Description</u>			
12.178 #1	SERVICE	1.00	\$14,545.27	\$14,545.27

Step: 12.178.1 Completed Removed Fuel panels 21-FT LWR-1, 21-FT LWR-2, 21-FT LWR-3, 21-FT LWR-4, 22-FT LWR-1, 22FT LWR-2. 22-FT LWR-3, 22-FT LWR-4, 22-FT UPR-12, 22-FT UPR-13, 22-FT UPR-15, 22 FT UPR-17. Done I.A.W. GIB 57-00-00.

Discrepancy: **12.179 CMP GIB 290110 SRVC HYD SYS RESERVOIR**

For detail description of the work complied with see the Work Order Detail report.

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Resolution: **Complied with CMP 290110 Hydraulic System Reservoir Filling Procedure. Work done I/A/W GIB MM chapter 29-01-00.**

Discrepancy: **12.180 CMP GIB 803027 Engine Starter Splines (Right) - Lubrication / Inspection**

Resolution: **COMPLIED WITH CMP GIB 803027 Engine Starter Splines (Right) - Lubrication / Inspection with satisfactory results IAW G2-B AMM 80-01-00.**

Labor USD: **No Charge**

N/ C

Discrepancy: **12.181 CMP GIB 803028 Engine Starter Assembly (Right) - Removal / Installation**

Resolution: **COMPLIED WITH CMP GIB 803028 Engine Starter Assembly (Left) - Removal / Installation. Removed Right Engine Starter Assy P/N 383700-1-1 S/N P-364 for spline lubrication and inspection and re-installed StarterP/N 383700-1-1 S/N P-364 IAW G-2B AMM 80-01-00 with no defects noted.**

Labor USD: **No Charge**

N/ C

Step: 12.181.1 Completed COMPLIED WITH CMP GIB 803028 Engine Starter Assembly (Right) - Removal / Installation Steps A. to C(10). Awaiting for Starter duct corrjoint seal.

Discrepancy: **12.182 Nine rivnuts need replaced on R/H wing wedge panel. Required drilling of screws due to rivnut spinning.**

Resolution: **Entered in error.**

Discrepancy: **12.183 Upper and lower bearings on L/H aileron bellcrank are chiped.**

Resolution: **Written in error. Duplicate squawk ,refer to squawk 8.28 for corrective action.**

For detail description of the work complied with see the Work Order Detail report.

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Communications concerning disputed debts, including a check or other instrument
tendered as full satisfaction of a debt, must be forwarded to:

Gulfstream Aerospace
Attn: Credit Manager
M/S B-05
PO BOX 2206,
Savannah, GA 31402

For detail description of the work complied with see the Work Order Detail report.

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