

Bell Helicopter - Miami, Inc
 2011 South Perimeter Road, Suite L
 Fort Lauderdale, Florida 33309

Phone: [REDACTED]



HYPERION AIR
INVOICE: C014572FL

Thursday, August 17, 2017
 Make / Model: Bell 430
 Serial Number: 59078
 Registration: N331JE
 Total Time: 1865.7

Item	Description	Parts	Labor
PERFORM AIRCRAFT HISTORICAL RESEARCH			
1	Performed aircraft historical records research. Checked for FAA AD compliance through FAA AD Bi-weekly Issue 2017-17	\$0.00	included #7
COMPLY WITH 12 MONTH TEST REQUIREMENTS ON ELT SYSTEM			
2	Complied with 12 month test requirements in accordance with 14 CFR Part 91 §91.207(d) on ELT system. Battery due replacement; June 2021.	\$0.00	included #7
PERFORM ENGINE POWER ASSURANCE CHECK ON BOTH ENGINES			
3	Performed engine power assurance check in accordance with AFS-BH430-IBF-RMS on both engines. Engine No. 1 calculated at; + 2 Engine No. 2 calculated at; + 2	\$0.00	\$0.00
PERFORM 100 HOUR AND 150 HOUR LIMITATION INSPECTIONS ON MAIN ROTOR YOKE ASSEMBLY			
4	Performed 100 hour and 150 hour limitation inspections in accordance with BHT-430-MM-2 Table 4-2 on Main rotor yoke assembly.	\$0.00	included #7
PERFORM 50 HOUR / 12 MONTH INSPECTION			
5	Performed 50 hour / 12 month inspection in accordance with BHT-430-MM-2 Section 5-13.	\$0.00	included #7
PERFORM 150 HOUR / 12 MONTH INSPECTION			
6	Performed 150 hour / 12 month inspection in accordance with BHT-430-MM-2 Section 5-14. Battery check performed by AOG Aircraft Services under their work order reference; BAT-000619. Performed operational checks as applicable to disturbed components. Notes: Contents of the following performed during compliance of this action. 150 hour limitation inspection in accordance with BHT-430-MM-2 Table 4-2 on Main rotor yokes. FAA AD2008-19-02 and BHT ASB430-00-17 (Hydraulic actuator support)	\$185.00	included #7

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Item	Description	Parts	Labor
PERFORM 600 HOUR / ANNUAL INSPECTION			
7	Performed 600 hour / Annual inspection in accordance with BHT-430-MM-2 Section 5-15. Performed operational checks as applicable to disturbed components.	\$3,819.73	\$21,090.00
PERFORM EVERY 100 HOURS OF COMPONENT OPERATION INSPECTION ON SWASHPLATE CONTROLS AND MAIN ROTOR YOKES			
8	Performed every 100 hours of component operation inspection in accordance with BHT-430-MM-2 Section 5-28 on Swashplate controls and Main rotor yokes. Note: Content of 100 hour limitation inspection in accordance with BHT-430-MM-2 Table 4-2 on Main rotor yokes complied with during performance of this item	\$0.00	included #7
PERFORM EVERY 300 HOURS OF COMPONENT OPERATION INSPECTION ON SWASHPLATE			
9	Performed Every 300 hours of component operation inspection in accordance with BHT-430-MM-2 Section 5-29 on Swashplate controls. Performed operational checks as applicable to disturbed components. Swashplate friction measured at; 25-28 lbs.	\$0.00	included #7
PERFORM EVERY 600 HRS OR 6 MHS OF COMPONENT OPERATION INSPECTION ON ENGINE FIRE EXTINGUISHER AND CARTRIDGE			
10	Performed every 600 hours or 6 months of component operation inspection in accordance with BHT-430-MM-2 Section 5-30 on Engine fire extinguisher and cartridge. Performed operational checks as applicable to disturbed components.	\$0.00	included #7
PERFORM EVERY 600 HOURS OR 12 MONTHS OF COMPONENT OPERATION INSP AND 600 HOUR / 12 MONTH LUBRICATION			
11	Performed every 600 hours or 12 months of component operation inspection and 600 hour / 12 month lubrication requirements in accordance with BHT-430-MM-2 Section 5-31 and Figure 12-5 on Main driveshaft (Engine to transmission driveshaft) and Tail rotor driveshaft. Performed operational and leak checks as applicable to disturbed components and dynamic balance of main driveshafts.	\$0.00	included #7
PERFORM EVERY 6 MONTHS INSPECTION ON STANDBY ATTITUDE INDICATOR			
12	Performed Every 6 months inspection in accordance with BHT-430-MM-2 Section 5-35 on Standby Attitude Indicator. Test performed by AOG Aircraft Services under their work order reference; BAT-000618. Performed operational checks as applicable to disturbed components.	\$175.00	included #7

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PERFORM 6 MONTH INSPECTION ON CLOCK BATTERY POWER SUPPLY			
13	Performed 6 month inspection in accordance with BHT-430-MM-10 Paragraph 95-200 on Clock battery power supply. Test performed by AOG Aircraft Services under their work order reference; BAT-000620. Performed operational checks as applicable to disturbed components	\$175.00	included #7
PERFORM EVERY 5 CALENDER YRS INSP AND 6 CALENDAR YEARS REPLACEMENT OF FIRE EXTINGUISHER CARTRIDGES			
14	Performed Every 5 calendar years inspection in accordance with BHT-430-MM-2 Section 5-38 on Fire extinguisher reservoir. Hydrostatic testing of fire extinguisher reservoirs performed by HFD Aero Systems under their work order reference 182418 and 182419. Replaced fire extinguisher cartridges p/n; 13083-5 (Qty 4) with new items p/n; AE13083-5 DOM; 06/16.	\$4,124.26	\$912.00
PERFORM AFTER EVERY MAIN ROTOR BLADE REMOVAL SPECIAL INSPECTION ON MAIN ROTOR YOKES			
15	Performed After every main rotor blade removal special inspection in accordance with BHT-430-MM-2 Section 5-40 on Main rotor yokes.	\$0.00	included #7
PERFORM CORROSION CONTROL INSPECTION			
16	Performed corrosion control inspection in accordance with CSSD-PSE-87-001 Section 30-7 and 30-8.	\$0.00	included #7
PERFORM 150 HOUR / 6 MONTH, 300 HOUR, 600 HOUR / 12 MONTH AND MISCELLANEOUS SERVICING REQUIREMENTS			
17	Performed 150 hour / 6 month, 300 hour, 600 hour / 12 month, and Miscellaneous servicing requirements in accordance with BHT-430-MM-2 Figure 12-1 on Tail rotor gearbox, Nose landing gear centering actuator, Environmental control unit cooling turbine, Transmission, Hydraulic reservoir, Battery, and Electrical power receptacle. Serviced Tail rotor gearbox and Transmission oil systems with oil; Royco 555. Replaced transmission oil filter with new item p/n; 206-040-128-003. Serviced hydraulic oil system with oil; MIL-H-5606. Performed operational and leak checks as applicable to disturbed components.	\$0.00	included #7
PERFORM 25 HOUR, 100 HOUR, 300 HOURS AND 300 HOUR / 12 MONTH LUBRICATION REQUIREMENTS			
18	Performed 25 hour, 100 hours, 300 hour and 300 hour / 12 month lubrication requirements in accordance with BHT-430-MM-2 Figure 12-5 on Pitch link universal, Rotating ring assembly, Cyclic lever, Tail rotor input lever bearing, Idler link, Collective lever, Tail rotor driveshaft hanger, Crosshead bearing housing, Counterweight bellcrank housing, Flapping bearing housing and Main and nose landing gear wheels. Performed operational checks as applicable to disturbed components.	\$0.00	included #7

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Item	Description	Parts	Labor
COMPLY WITH FAA AD2008-19-02 AND BHT ASB430-00-17			
19	Complied with FAA AD2008-19-02 Para (a) Effective October 16th, 2008 (Hydraulic actuator support) and BHT ASB430-00-17 Dated May 19th, 2000 (Hydraulic actuator support p/n 222-040-125-001, Inspection of) in accordance with AD and ASB Instructions.	\$0.00	included #7
PERFORM 150 HOUR AND 300 HOUR / 12 MONTH ENGINE INSPECTIONS AND SERVICING REQUIREMENTS ON NO. 1 ENGINE			
20	Performed 150 hour and 300 hour / 12 month engine inspections and servicing requirements in accordance with Rolls Royce 250-C40B Operation and Maintenance Manual Section 72-00-00 Table 606 on No. 1 engine. Drained and serviced engine oil system with oil; Royco 560. Replaced CEFA oil filter with new item p/n 23066697. Performed operational and leak checks as applicable to disturbed components and compressor wash and rinse.	\$0.00	included #7
PERFORM 150 HOUR AND 300 HOUR / 12 MONTH ENGINE INSPECTIONS AND SERVICING REQUIREMENTS ON NO. 2 ENGINE			
21	Performed 150 hour and 300 hour / 12 month engine inspections and servicing requirements in accordance with Rolls Royce 250-C40B Operation and Maintenance Manual Section 72-00-00 Tables 603 and 604 on No. 2 engine. Drained and serviced engine oil system with oil; Royco 560. Replaced CEFA oil filter with new item p/n 23066697. Performed operational and leak checks as applicable to disturbed components and compressor wash and rinse.	\$0.00	included #7
PERFORM 600 HOUR / 12 MONTH SERVICING REQUIREMENTS ON ENGINE OIL AND FILTER SYSTEMS ON BOTH ENGINES			
22	Performed 600 hour / 12 month servicing requirements in accordance with Rolls Royce 250-C40B Operation and Maintenance Manual Section 72-00-00 Para 8.D. - Engine servicing on Engine oil and filter systems on both engines. Serviced engine oil systems with oil; Royco 560. Replaced CEFA oil filters with new items p/n 23066697.	\$0.00	included #7
PERFORM 100 HOUR, ANNUAL AND 300 HOUR / 1 YEAR INSPECTIONS ON INLET BARRIER FILTER INSTALLATION			
23	Performed 100 hour, Annual and 300 hour / 1 year inspections in accordance with Aerospace Filtration Systems Instructions for Continued Airworthiness AFS-BH430-IBF-ICA on Inlet barrier filter installation. Performed operational checks as applicable to disturbed components.	\$0.00	\$684.00
COMPLY WITH 5000 HOUR / 5 YEAR OVERHAUL REQUIREMENTS ON MAIN DRIVESHAFT			
24	Complied with 5000 hour / 5 year overhaul requirements in accordance with BHT-430-MM Table 5-2 on Main driveshaft p/n; 222-044-685-105 s/n; A-1109 and s/n; A-1087. NDT performed by Engineering and Inspections Unlimited under their work order reference PO019570FL.	\$646.94	\$3,420.00

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Item	Description	Parts	Labor
NO 2 COMM UNUSABLE (CONSTANT)			
25	Investigated discrepancy of No. 2 Comm unusable and determined circuit breaker to be defective. Replaced circuit breaker with new item p/n; MS22073-7.5. Work performed in accordance with BHT-430-MM Chapter 97. Performed operational checks as applicable to disturbed components.	\$121.72	\$199.50
PUSH TO TALK ON PILOT CYCLIC IS STICKING AND DIFFICULT TO TRANSMIT			
26	Investigated discrepancy of push to talk switch on pilot cyclic sticking and difficult to transmit. Adjusted spring tension for PTT switch. Switch operates without sticking. Work performed in accordance with BHT-430-MM Chapter 97 and BHT-ELEC-SPM.	\$0.00	\$342.00
PERFORM COMPRESSOR WASH AND RINSE ON BOTH ENGINES			
27	Performed compressor wash and rinse in accordance with Rolls Royce 250-C40B Operation and Maintenance Manual Section 72-30-00 Para 4 on both engines. Performed operational checks as applicable to disturbed components.	\$0.00	\$342.00
COMPLY WITH FAA AD2016-21-01			
28	Complied with FAA AD2016-21-01 Effective November 18, 2016 (Main rotor head attachment bolt) in accordance with AD instructions.	\$0.00	\$114.00
SERVICE LANDING GEAR STRUTS			
29	Serviced landing gear struts with nitrogen in accordance with BHT-430-MM Chapter 12.	\$0.00	\$114.00
UPDATE GNS500 NAVIGATION DATABASE			
30	Updated GNS530 navigation database	\$218.75	\$0.00

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Item	Description	Parts	Labor
REMOVE LEFT AND RIGHT HAND CABIN DOOR WINDOW SHADES AND SEND TO VENDOR FOR CLEANING			
31	Removed left and right hand cabin window shades p/n; 430-5630-100-003 s/n; G0210116 and p/n; 430-5630-100-004 s/n; E0210085 in accordance with Aero Solutions Instructions for Continued Airworthiness M430-5630-101. Equipment originally installed under FAA Form 337 Dated March 3, 2004. Updated aircraft weight and balance data and equipment listing. FAA Form 337 completed.	\$0.00	\$798.00
LEFT HAND INLET COWLING HAS ONE EJECTOR BLADE BROKEN AND 12 GROMMETS NOT INSTALLED			
32	Replaced ejector blade with new item p/n; 50-010-5 and installed grommets p/n; 50-009-5 (Qty 2) to left hand inlet cowling in accordance with BHT-430-MM Chapter 71.	\$6.50	\$114.00
NO. 2 ENGINE DRAIN VALVE AT RESERVOIR BAS BROKEN HANDLE			
33	Replaced No. 2 engine drain valve with new items p/n; MS29530-8 in accordance with BHT-430-MM Chapter 79. Performed leak checks as applicable to disturbed components.	\$871.00	\$114.00
EVIDENCE OF ROTOR BRAKE LEAK ABOVE PILOT SEAT. INVESTIGATE FOR CAUSE			
34	Investigated discrepancy of evidence of oil leak above pilot seat. Noted residual oil on ECS drain line. No damage noted to drain line. Cleaned oil and sealed drain line at junction to cabin deck head. Work performed in accordance with BHT-ALL-SPM.	\$0.00	\$114.00
GREEN MAIN ROTOR PITCH LINK LOWER BOLT HAS BENT SPRING RETAINER RING. REPLACE WITH NEW			
35	Replaced green main rotor pitch link lower bolt with new item p/n; 50-047C6-32 in accordance with BHT-430-MM Chapter 62.	\$308.00	\$0.00
CABIN FIRE EXTINGUISHER CHARGE LOW. REPLACE WITH NEW			
36	Replaced cabin fire extinguisher with new item p/n A344T in accordance with BHT-430-MM Chapter 26.	\$161.00	\$0.00

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Item	Description	Parts	Labor
CHIP DETECTOR CANNON PLUG (LEFT QUILL) BROKEN. REPLACE WITH NEW			
37	Replaced transmission left input quill chip detector cannon plug with new item p/n; 450FA005B08 in accordance with BHT-ELEC-SPM. Performed operational checks as applicable to disturbed components.	\$195.00	\$171.00
SEVERAL SPOTS OF CORROSION ON BOTTOM OF GREEN MAIN ROTOR BLADE. REMOVE, CLEAN AND REFINISH			
38	Mechanically removed surface corrosion from lower surface of green main rotor blade in accordance with BHT-430-MM Chapter 62. Protective finish re-applied.	\$0.00	\$342.00
PITCH, ROLL AND COLLECTIVE ACTUATORS HAVE BROKEN GUIDE ROLLERS. REPLACE WITH NEW			
39	Replaced servo actuator roller assemblies with new items p/n; 141243-103 (Qty 3) in accordance with BHT-430-MM Chapter 29.	\$3,597.00	\$570.00
CO-PILOT DOOR SEAL HAS TEAR AT FORWARD PART OF DOOR			
40	Replaced torn section of co-pilots door seal in accordance with BHT-430-MM Chapter 52. Performed leak checks as applicable to disturbed components.	\$0.00	\$0.00
EVIDENCE OF OIL LEAK FROM TAIL ROTOR GEARBOX INPUT AND OUTPUT SEALS			
41	Removed tail rotor hub and blade assembly, tail rotor pitch change spider and aft tail rotor driveshaft for access. Replaced tail rotor gearbox input seal and wear sleeve with new items p/n; 222-342-401-103 and 536836 respectively. Replaced tail rotor gearbox output seal and wear sleeve with new items p/n; 222-342-402-101 and 222-042-443-103 respectively. Reinstalled aft tail rotor driveshaft, tail rotor pitch change spider and tail rotor hub and blade assembly. Work performed in accordance with BHT-430-MM Chapter 64 and 65, and BHT-430-CR& O Chapter 65. Performed operational and leak checks as applicable to disturbed components.	\$13,428.15	\$2,736.00
BROKEN RING TERMINAL (2) ON RIGHT HAND FIRE BOTTLE CONNECTORS			
42	Replaced ring terminals to right hand fire extinguisher reservoir connectors in accordance with BHT-ELEC-SPM.	\$0.00	\$0.00

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Item	Description	Parts	Labor
TAIL ROTOR CONTROL TUBE AFT NYLATRON SLEEVE IS DEBONDED AT EDGE			
43	Re-bonded aft nylatron sleeve for tail rotor control tube in accordance with BHT-430-CR&O Chapter 67.	\$0.00	\$114.00
COMPLY WITH BHT ASB430-16-57 AND ROLLS ROYCE 250-C40B CEB A-72-5070			
44	Complied with BHT ASB430-16-57 Revision C Dated March 3, 2017 (Engine, Combustion assembly - Inspection of outer case p/n; M250-10404) and Rolls Royce 250-C40B CEB A-72-5070 Revision 3 Dated February 2, 2017 (Engine, Combustion assembly - Inspection of outer combustion case p/n; M250-10404) in accordance with ASB and CEB instructions. Not applicable due to part number of outer combustion case installed.	\$0.00	\$0.00
COMPLY WITH BHT ASB430-14-53 AND ROLLS ROYCE 250-C40B CEB A72-5066			
45	Complied with BHT ASB430-14-53 Dated September 3, 2014 (Engine, Compressor assembly - Inspection of front support p/n; M250-10051, M250-10052, M250-10059, and M250-10060) and Rolls Royce 250-C40B CEB A72-5066 Revision 1 Dated August 22, 2014 (Engine, Compressor assembly - Inspection of front support p/n; M250-10051, M250-10052, M250-10059, and M250-10060) in accordance with ASB and CEB instructions. Not applicable due to part number of front support installed.	\$0.00	\$0.00
COMPLY WITH BHT ASB430-14-52 AND ROLLS ROYCE CEB A72-5064			
46	Complied with BHT ASB430-14-52 Dated August 18, 2014 (Engine - Spur adapter gearshaft, Retaining ring) and Rolls Royce CEB A-72-5064 Dated May 14, 2014 (Engine - Spur adapter gearshaft, Retaining ring) in accordance with ASB and CEB instructions. Not applicable due to serial number of engine or compressor installed and adapter coupling not replaced in two years prior to bulletin.	\$0.00	\$0.00
COMPLY WITH BHT ASB430-13-50 AND ROLLS ROYCE 250-C40B CEB-72-5062			
47	Complied with BHT ASB430-13-50 Dated August 7, 2013 (Engine, assembly - Inspection of mag plugs to detect metal generation from the No. 2 bearing) and Rolls Royce 250-C40B CEB-72-5062 Revision 1 Dated June 10, 2013 (Engine, assembly - Inspection of mag plugs to detect metal generation from the No. 2 bearing) in accordance with ASB and CEB instructions. Not applicable due to serial number of engine or compressor installed.	\$0.00	\$0.00
FORWARD TRANSMISSION COWLING HAS DZUS FASTENER DETACHING			
48	Re-secured dzus fastener to forward transmission cowling in accordance with BHT-430-MM Chapter 71 and BHT-ALL-SRM.	\$0.00	\$0.00

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Item	Description	Parts	Labor
49	Aircraft washed and waxed by Mike Pro Detailing.	\$750.00	
50	Consumables / Hazardous Waste	\$1,290.00	
51	Freight Charges	\$490.00	
SUB-TOTAL		\$30,563.05	\$32,290.50
TOTAL		\$62,853.55	